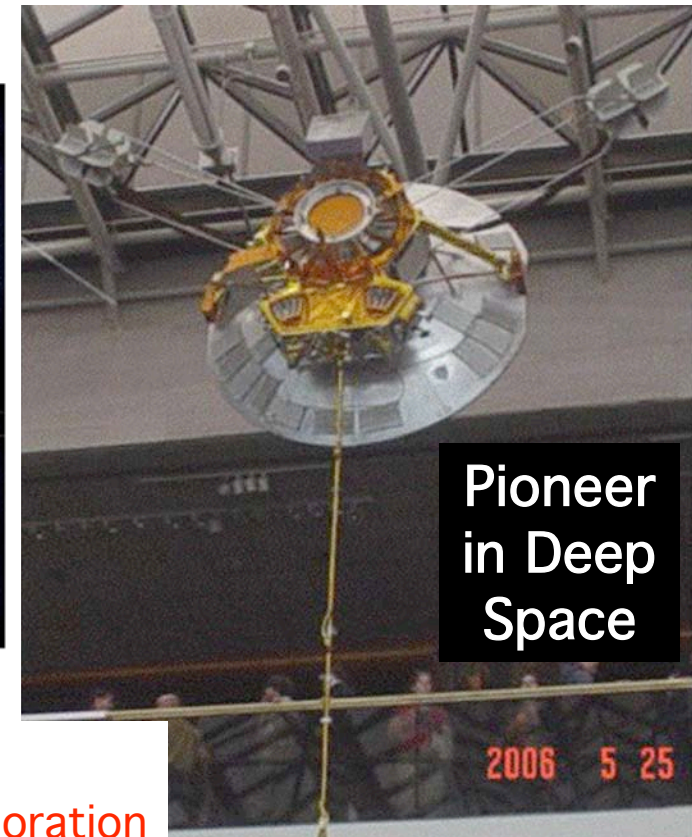
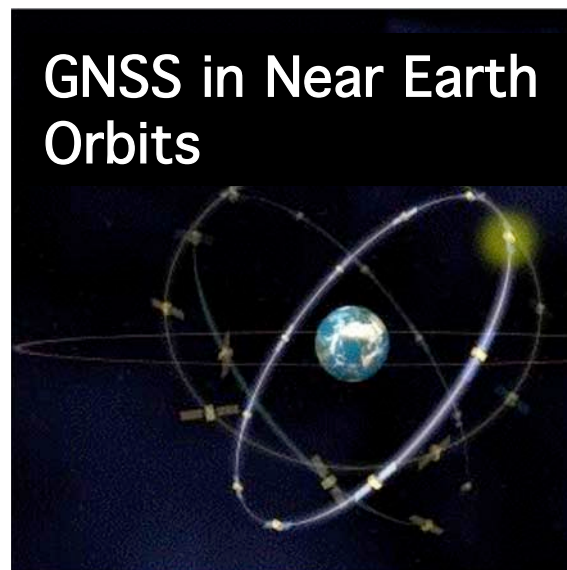


# Probing Gravity in the Solar System with ETRUSCO

Extra Terrestrial Ranging to Unified Satellite Constellations

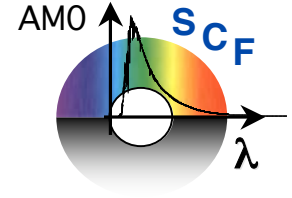
S. Dell'Agnello, INFN-LNF  
(Resp., 30%)  
G. Delle Monache, INFN-LNF  
(30%)  
C. Cantone, INFN-LNF  
(Bors. INFN, 30%)  
M. Garattini, INFN-LNF  
(Bors. INFN, 30%)  
G. Bellettini, Roma2  
(PO, 30%)  
TOTAL = 1.5 FTE



Slava Turyshev, NASA-JPL, Spokesman of the  
“Pioneer Explorer and Deep Space Gravity Probe” Collaboration

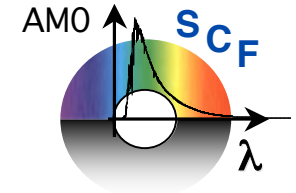
# Outline

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- Probing gravity in the Solar System
- ETRUSCO concept and goals, **in brief** (slides 5-9)
- “Unified” GNSS constellations
- The existing INFN-LNF Space Climatic Facility (SCF)
- Analysis of PIONEER data and the new proposed “Unified” satellite formation, DSGP
- Conclusions (slide 41)
- **Group and financial requests** (slides 42-45)

# Experimental tests of General Relativity



Space-time  
curvature in GR,  
Cassini probe:

$$\gamma = 1 \pm 2 \times 10^{-5}$$

$$\gamma(\text{GR}) = 1$$

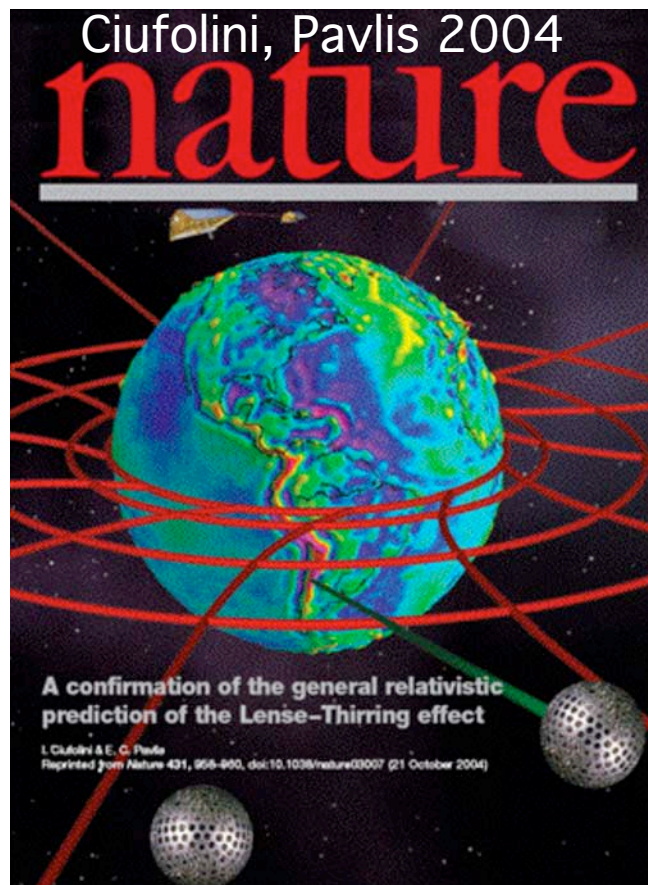
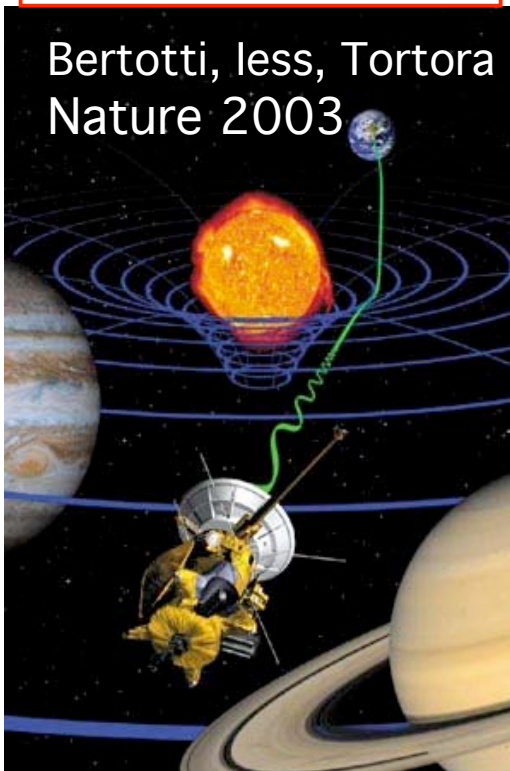
$$\gamma(\text{Newton}) = 1/2$$

Dragging of space-time in GR, LAGEOS satellites

$$\dot{\Omega}^{L-T} = 99\% \times \text{GR value} (\pm 5-10\% \text{ error})$$

LARES satellite (INFN-GR2) to get  $\leq 1\%$  error

Bertotti, Iess, Tortora  
Nature 2003



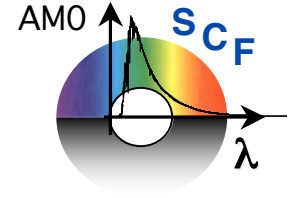
Gravitational waves  
of space-time:  
'the' most important  
dynamical test of GR

NAUTILUS @ LNF



# Laboratory physics of General Relativity

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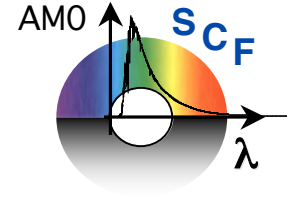


- The **Solar System is our laboratory**
- Near Earth Orbits (NEO)
  - Accurate measurement of space-time coordinates of “Global Positioning” satellite constellations
    - GPS-3 (USA), GALILEO (Europe)
- Deep Space
  - Study of the anomalous deceleration of the PIONEERS (72-02).  
**Test of  $1/r^2$  law in Deep Space**
    - Publish a technology paper on thermal re-analysis of the Pioneers
    - Publish a physics measurement on the  $1/r^2$  test
  - In parallel, by the end of 2006, propose the new mission:  
**Deep Space Gravity Probe**, for the **ESA Cosmic Vision Program**



# The ETRUSCO concept

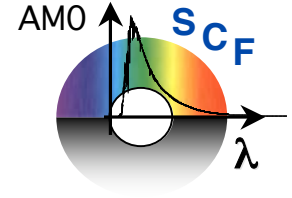
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- “**E**xtra **T**errestrial **R**anging”: measurement of space-time positions of active satellites or passive test-masses **in space** with electromagnetic waves (**ranging**)
- Use two “**U**nified **S**atellite **C**onstellations” to probe gravity in the solar system
  - GNSS (Global Navigation Satellite System) in NEO
    - GALILEO (30 active satellites, 29601 Km altitude)
    - GPS-3 (30 active satellites, 20195 Km altitude)
    - ITRF: International Terrestrial Reference Frame
  - Deep Space Gravity Probe (DSGP)
    - Satellite formation of 1 active spacecraft + a few passive test masses
    - Outbound orbit, spanning the whole solar system
    - ICRF: International Celestial Reference Frame
- “**U**nification”: **addition of LASER ranging to MICROWAVE ranging**

# Goals of ETRUSCO for GNSS

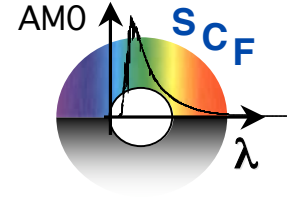
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- Physics goals
  - high-accuracy experimental map of space-time with a high-multiplicity satellite constellation
  - Excellent data set for rigorous and non trivial definition of the most-general non-inertial reference frames in GR
- Technological goals
  - GALILEO is “unified” (laser ranging included in the design)
    - So far, traditional retro-reflectors are foreseen : solid, fused-silica CCRs
  - GPS-3 is to be “unified” (ie to convince US agencies to add laser ranging)
  - Upgrade the LNF Space Climatic Facility (SCF) to perform a dedicated climatic and optical characterization of retro-reflectors arrays for GNSS conditions (20000-29000 Km altitude)
  - Develop new, state-of-art retroreflectors
    - Hollow, metallic CCRs (Be or Al). Lighter and smaller than solid CCRs

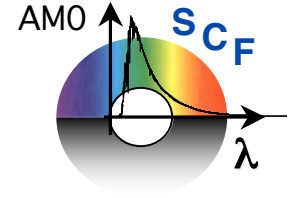
# Goals of ETRUSCO for DSGP

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- Physics goals
  - Test of  $1/r^2$  law in deep space. Unique to Pioneers/DSGP
- Technological goals:
  - Use our expertise for a critical thermal analysis of Pioneer data to find out if anomalous deceleration is due to an instrumental effect. Never done before. Radioactivity on board is the suspect
  - Test small samples and components of the Pioneers at the LNF SCF
  - Design laser-ranged spherical test masses for DSGP, the first-ever unified satellite formation in deep space. Here our expertise with the design and test of LARES is crucial
    - LARES designed to be the most accurate laser-ranged test mass ever
    - LARES typical thermal accelerations in NEO are 1/10th of Pioneer anomalous decelerations: so we are already dealing with fractions of the effect of interest
    - If LARES, as designed for NEO, were to be sent to Saturn and beyond, thermal accelerations on it would be 1/100th than in NEO, because the solar constant  $\propto 1/r^2 \Rightarrow$  so known thermal effects on a LARES in deep space would be 1/1000th of the Pioneer anomalous deceleration

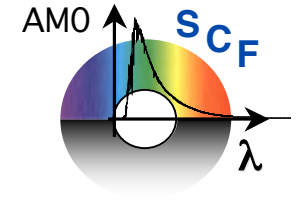
# Truly international, inter-agency effort



- Overall study proposed to ASI for 2006-2008 by this LNF group
- International collaboration with:
  - International Laser Ranging Service (ILRS): E. C. Pavlis et al
  - Univ. of Maryland at College Park (UMCP): D. Currie et al
  - NASA-Goddard Space Flight Center (GSFC): J. McGarry et al
  - NASA-Jet Propulsion Laboratory (JPL): S. Turyshev et al
- GNSS
  - GPS-3: formal collaboration with ILRS, NASA-GSFC, UMCP
  - GALILEO: we did not contact the GALILEO consortium, yet !  
We'd like to do it with ETRUSCO. If we succeed in performing the climatic/optical characterization of their retro-reflectors, it would also be a major, winning, high-profile technological application of INFN research. GALILEO is worth billion €'s
- Pioneer/DSGP
  - Collaboration with NASA-JPL, Germany, France, ASI
  - DSGP mission to be proposed to ESA by Jan 2007



# Truly interdisciplinary effort



- The ETRUSCO project requires knowledge of:

- General relativity
- Space navigation
- Space geodesy
- Cryogenics and vacuum
- Space climate
- Thermography with IR camera and probes
- Lasers and optics analysis
- Precision mechanics and mounting
- Finite element, finite difference modeling
- Thermal analysis for satellites
- Physics data analysis

LNF SCF &  
Cryo service

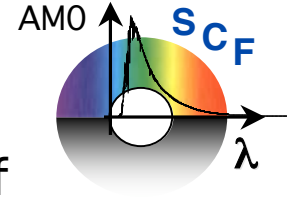
LNF ex-Virgo  
optical bench

LNF Mech.  
service

LARES  
expertise

- With the SCF, our professional international collaborators, plus some extra support from INFN-GR5 ... we can do it !

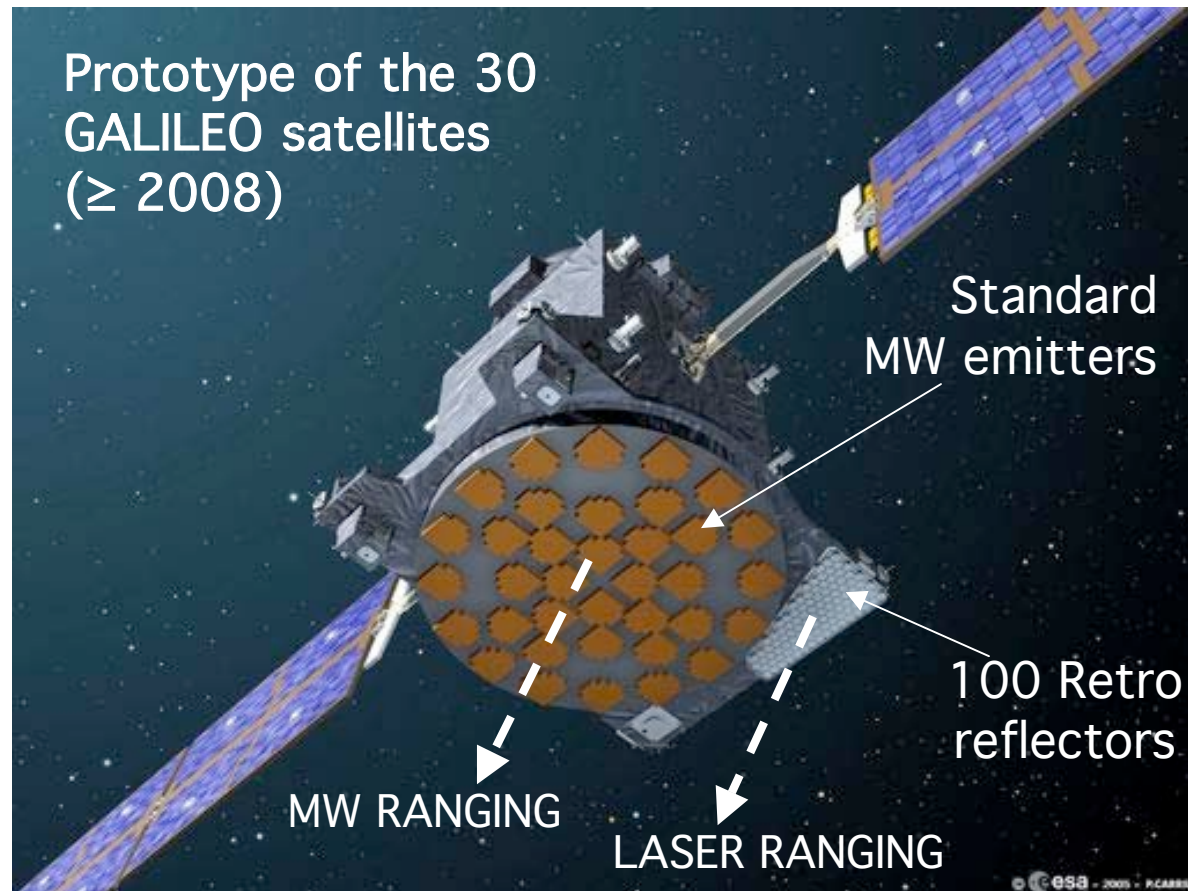
# A Global Positioning Unified Constellation



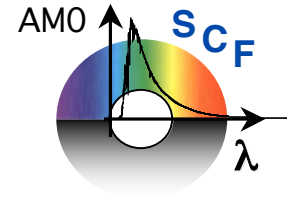
**MW Ranging:** standard measurement of space-time coordinates of a test mass (the “GPS” satellite) with microwaves.  $\sigma \sim 10\text{-}20\text{ cm}$

No long term memory (clock re-calibration) but great for real-time navigation

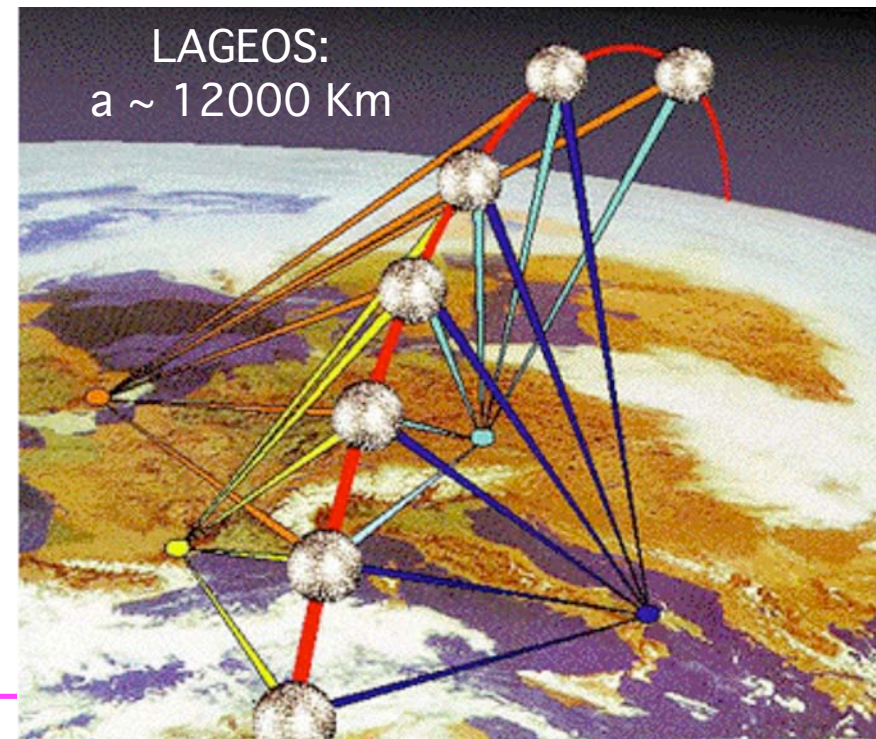
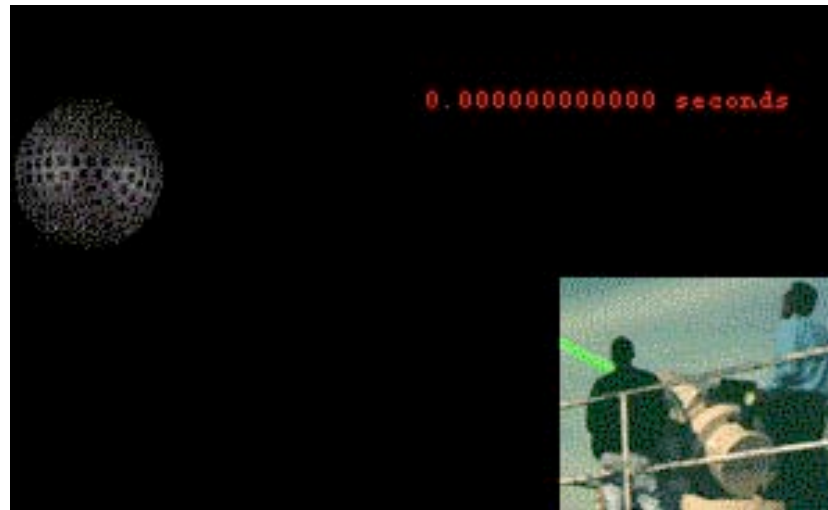
**LASER Ranging:** retro-reflectors (CCR) and optical waves.  $\sigma \sim \text{few mm}$ ,  
absolute position standard, long term stability (tens of yrs)



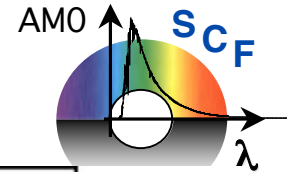
# Satellite Laser Ranging (SLR)



- Orbit accuracy < 1 cm
- Orbits used to define the International Terrestrial Reference Frame (ITRF)
  - Issued by IERS = International Earth Reference Service
  - “The” standard for all civilian, commercial and scientific activities

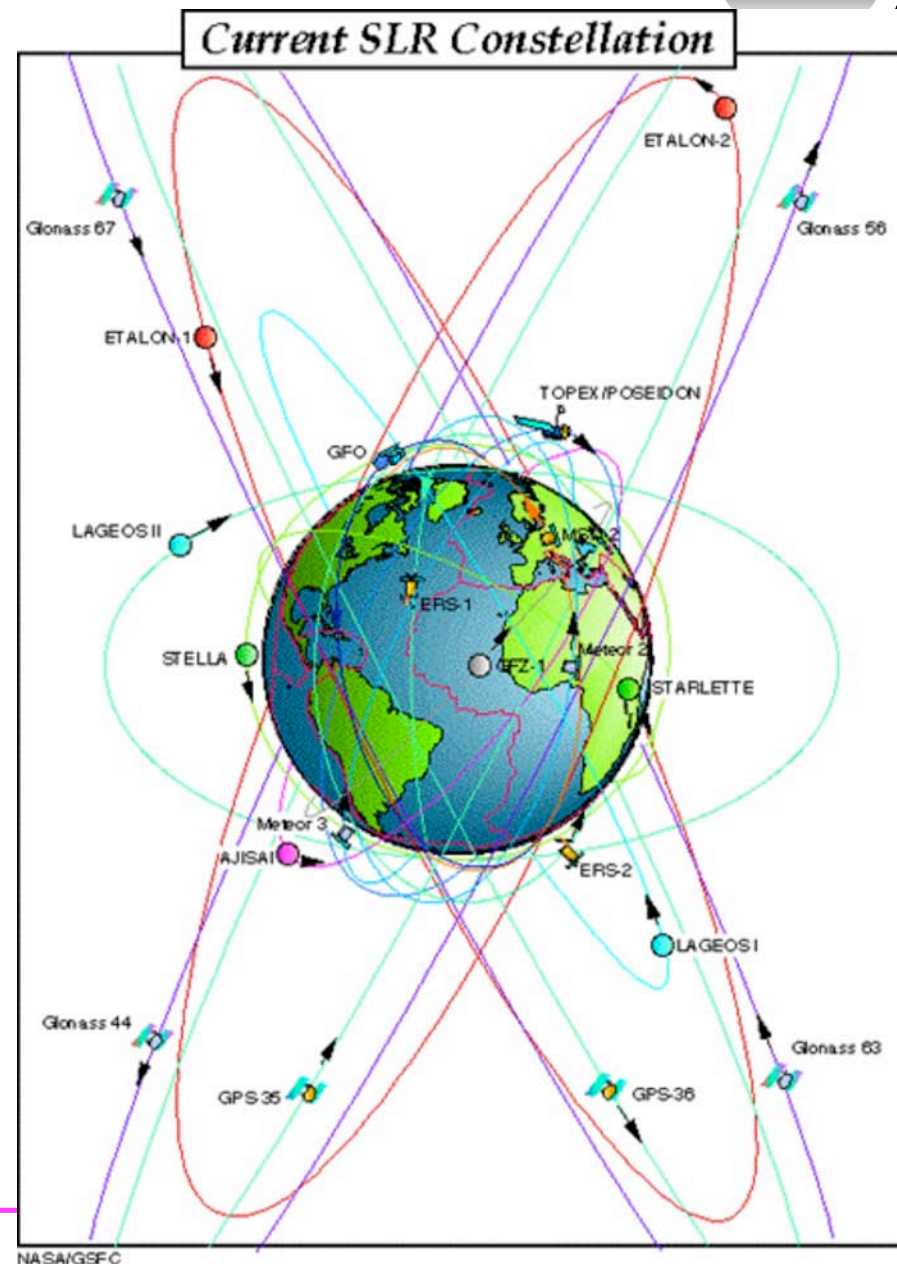


# Current laser ranging constellation



Satellite distances:  
400 Km to **> 20000 Km**  
**for GPS/GLONASS**

Now only 2 GPS-2 (USA) and 4  
GLONASS (Russia) satellites  
have retro-reflectors (traditional  
solid CCRs made of fused silica)





# GNSS solid retroreflector arrays

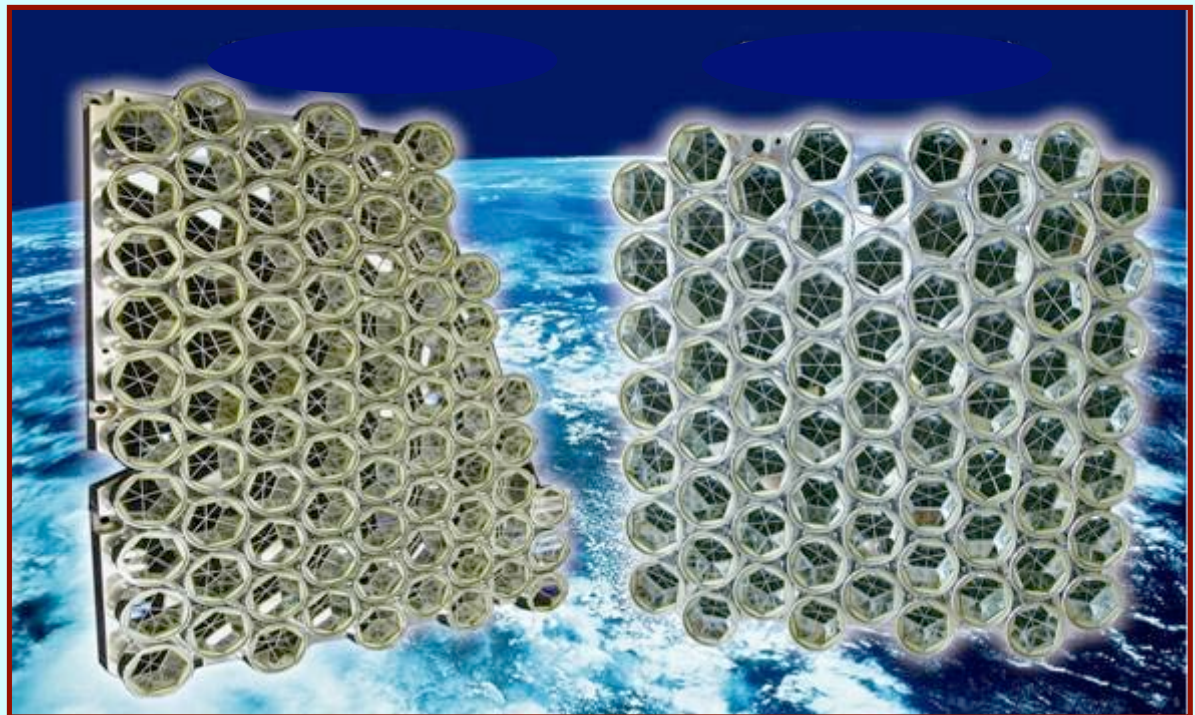
V. Vasiliev, IPIE-Moscow; talk at **FPS-06**, Frascati, March 06  
(see <http://www.lnf.infn.it/conference/fps06/>)

## ***GALILEO TEST satellites***

*Orbit:  $h = 23200$  km,  $i = 56^\circ$*

***GPS-35*** } *Orbit:  $h = 20200$  km,  $i = 54^\circ$*   
***GPS-36*** } *Number of CCR's: 32*

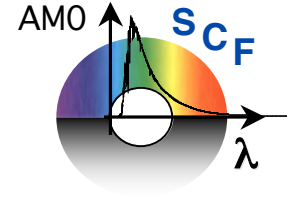
**GIOVE-A (76 CCRs) GIOVE-B (67 CCRs)**





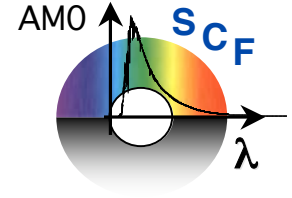
# GNSS = Global Navigation Satellite System

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- MW ranging among
  - the constellation (27 satellites + 3 spares), the fixed ground stations (over 250) and you, the final end user
- GNSS = 3 constellations of 30 satellites each
  - GPS (USA), GLONASS (Russia), GALILEO (Europe)
- Enormous impact in many fields
- IGS (International GNSS Service)
- GPS/GLONASS are proprietary/military at heart

# GALILEO ( $\geq 2008$ ) and GPS-3 ( $\geq 2011$ )

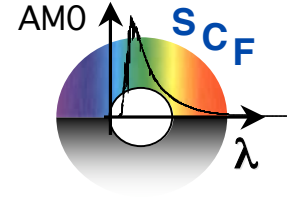


- European GALILEO (E.U. + ESA) changes whole picture
  - Atomic clocks better than GPS ones, **commercial** constellation, **civilian use** only. “Unified”:
    - 100 CCRs on each and every one of the 30 satellites !
    - CCR approximate cost (my estimate): 5K€/CCR  $\Rightarrow$  **15 M€**
- Addition of CCRs
  - improves space accuracy by factor  $\sim 20$ ; improves performance for space geodesy and for commercial services of enormous €-value
- GALILEO puts pressure on US for GPS-3
- ILRS wants to equip GPS-3 with **hollow CCRs**
  - Lighter and smaller

**PLX**



# GNSS observation with laser ranging

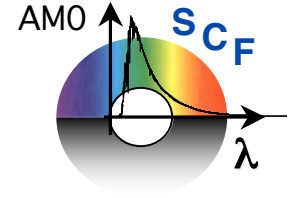


- HOLLOW CCRs: long term stability and performance in space environment to be proven
  - Agreed plan: structural analysis by NASA-GSFC, climatic test by LNF
- SLR+MW: the best of both worlds to map the NEO space-time

Calculations by D. Arnold, ILRS meeting at EGU, April 06, Vienna

Simulations at Galileo altitude for Effective Cross Section of 100 million sq. meters.				
Design	# of cubes	Diam. (inch)	Approx. Area of the cornercubes (sq cm)	Approx Mass of the cornercubes (gm)
uncoated	50	1.3	428	1000
coated	400	0.5	508	460
→ hollow	400	0.5	508	201
→ hollow	36	1.4	356	400
Present GPS cubes	160	1.06	1008	1760

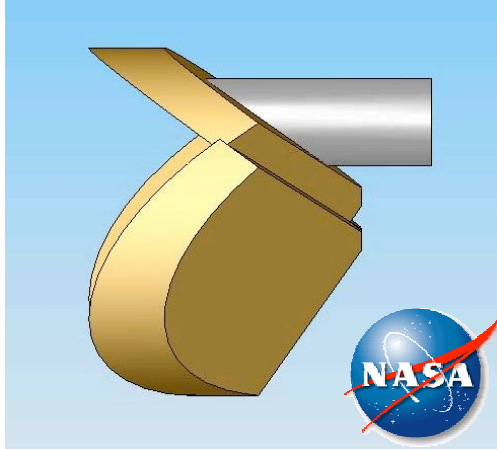
# Beryllium Hollow CCR candidate for GPS3



"Fake"  
spacecraft

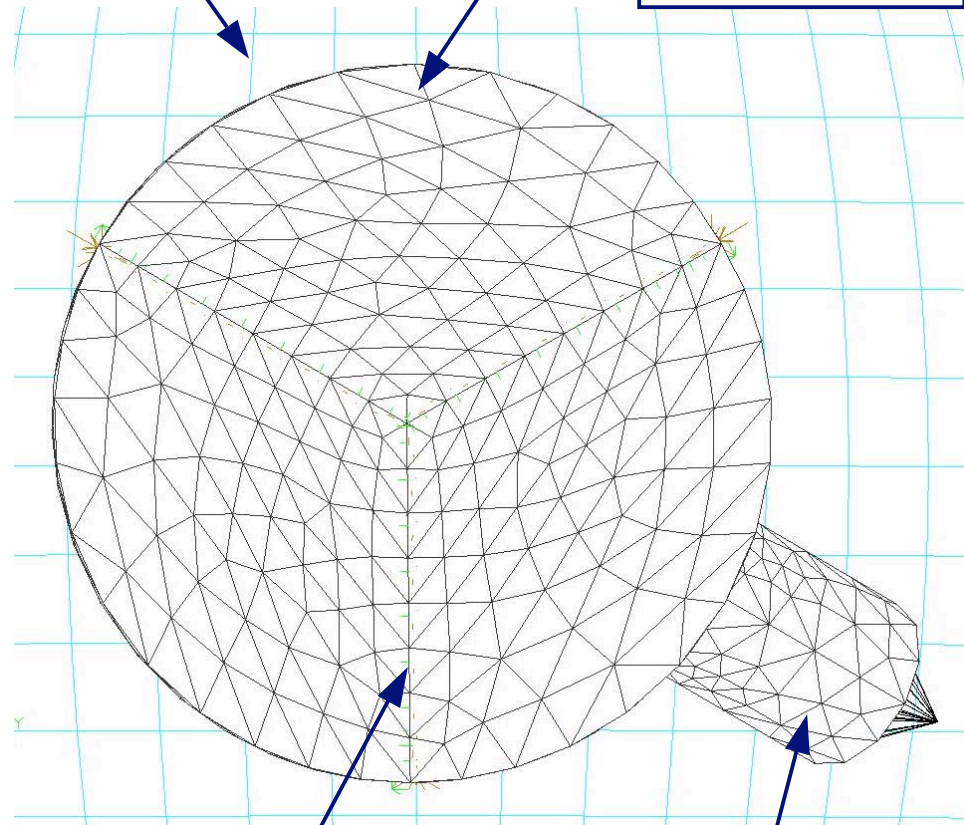
3 Be planes

CCR modeled with ThermalDesktop;  
bonding effects between the 3  
planes and the post modeled  
Very crude spacecraft model: an Al  
sphere surrounding the CCR



Stycast bonding  
(10W/K)

Post

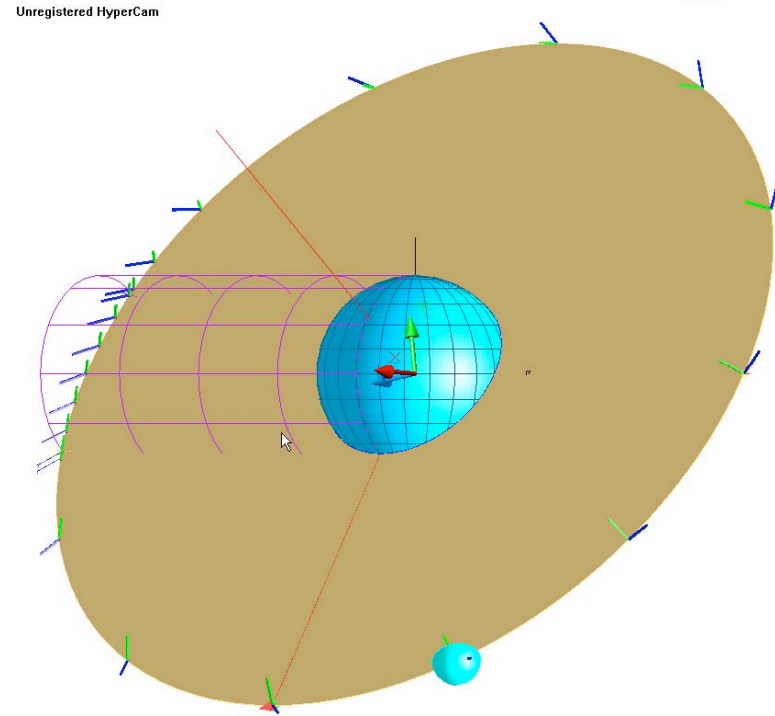
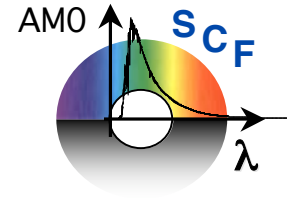
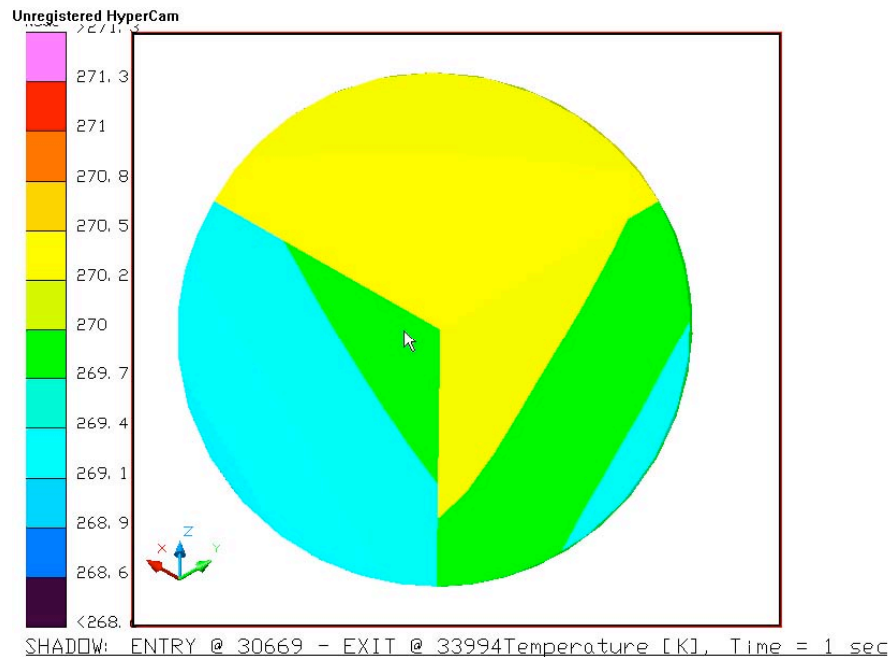


# Climatic simulation of GPS-3 hollow CCR

Hollow CCRs made of 3 pieces, glued (Be) or bolted (Al).

Most severe problem is CCR warping in space climate, which could prevent laser light return. T variation of CCR:

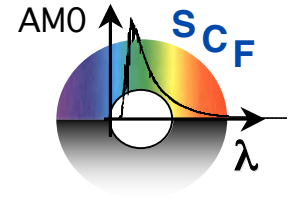
- 300-400 K if CCR thermally isolated from spacecraft
- Much smaller gradient if CCR thermally linked to spacecraft.



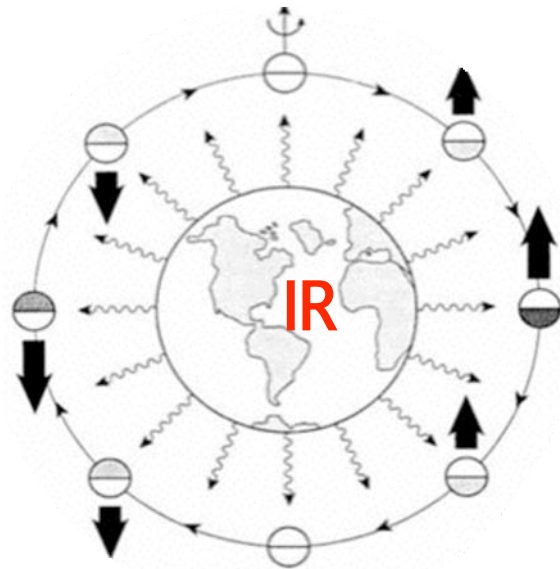
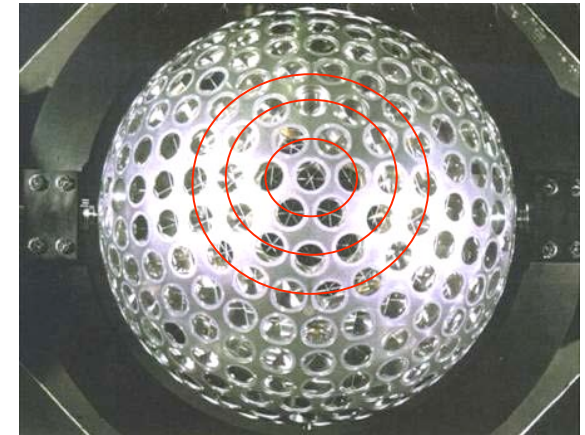
G. Delle Monache. Preliminary.



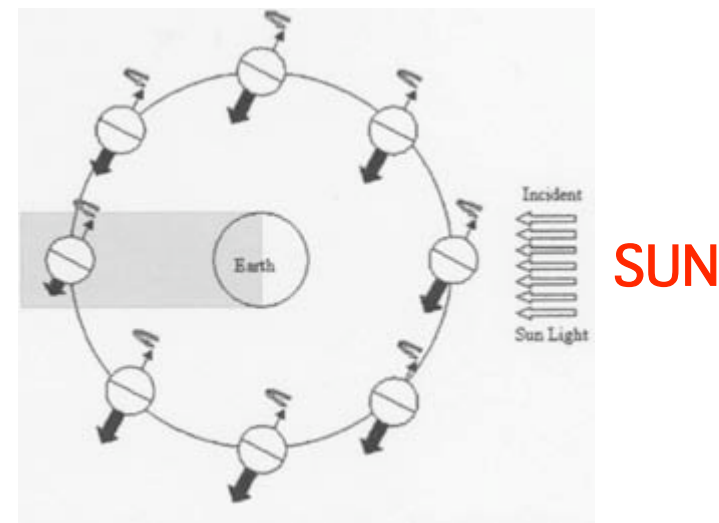
# The (existing) LNF Space Climatic Facility



- We do have a climatic facility, built for thermal test of LAGEOS/LARES prototypes of INFN-GR2
- Measurement of optical properties with lasers on a separate optical bench
- Merge of climatic and optical test TO BE DONE

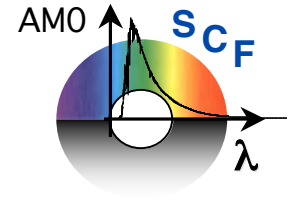


Earth Infrared Yarkovsky effect

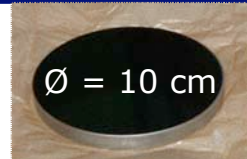


Solar Yarkovsky effect

# The LNF Space Climatic Facility (SCF)

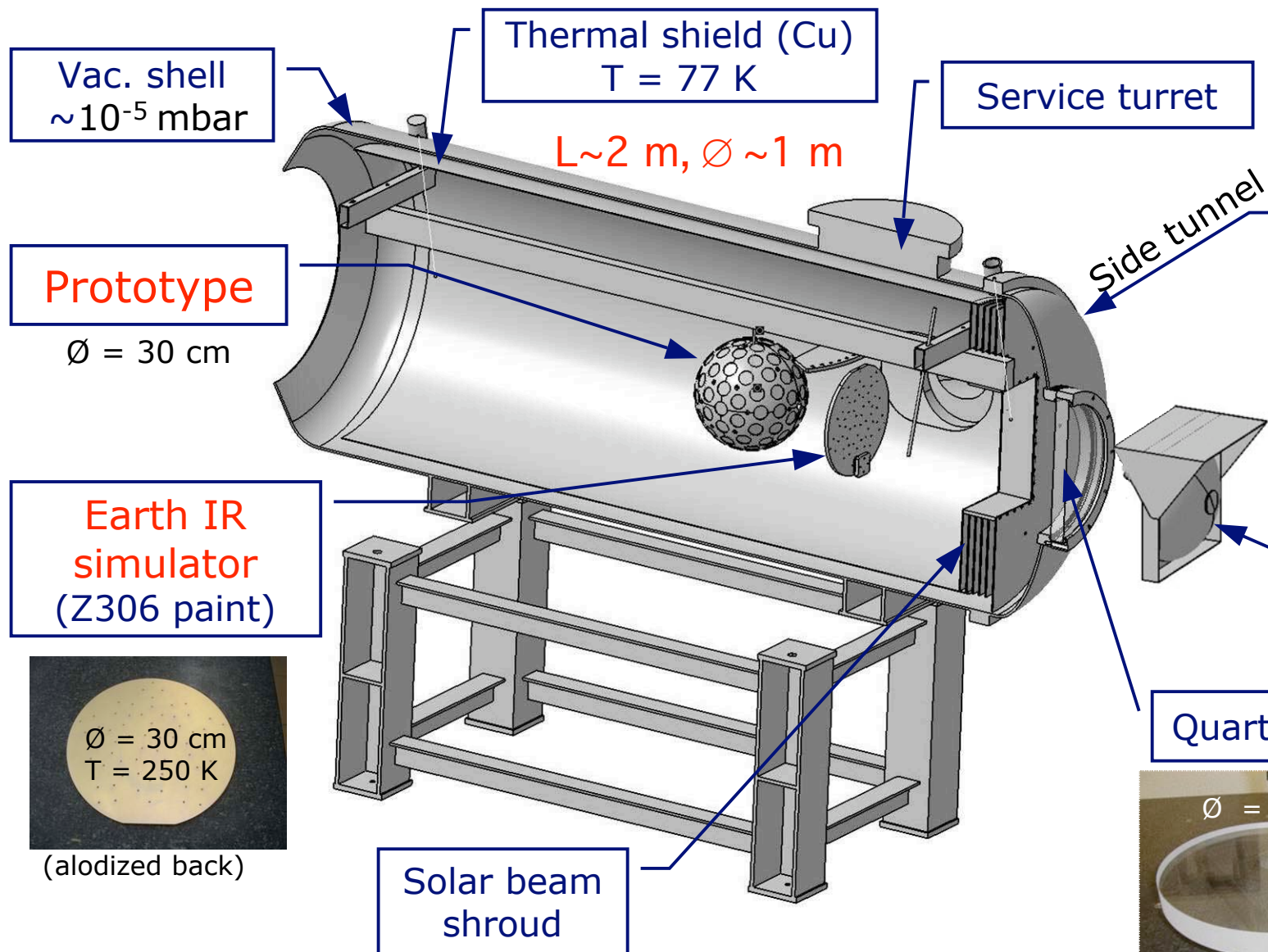
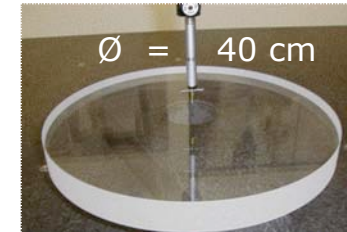


IR camera  
Ge window

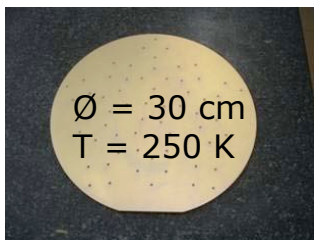


Solar NEO  
simulator

Quartz window



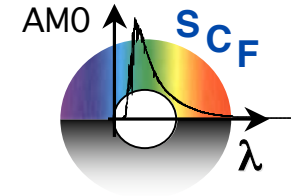
Earth IR  
simulator  
(Z306 paint)



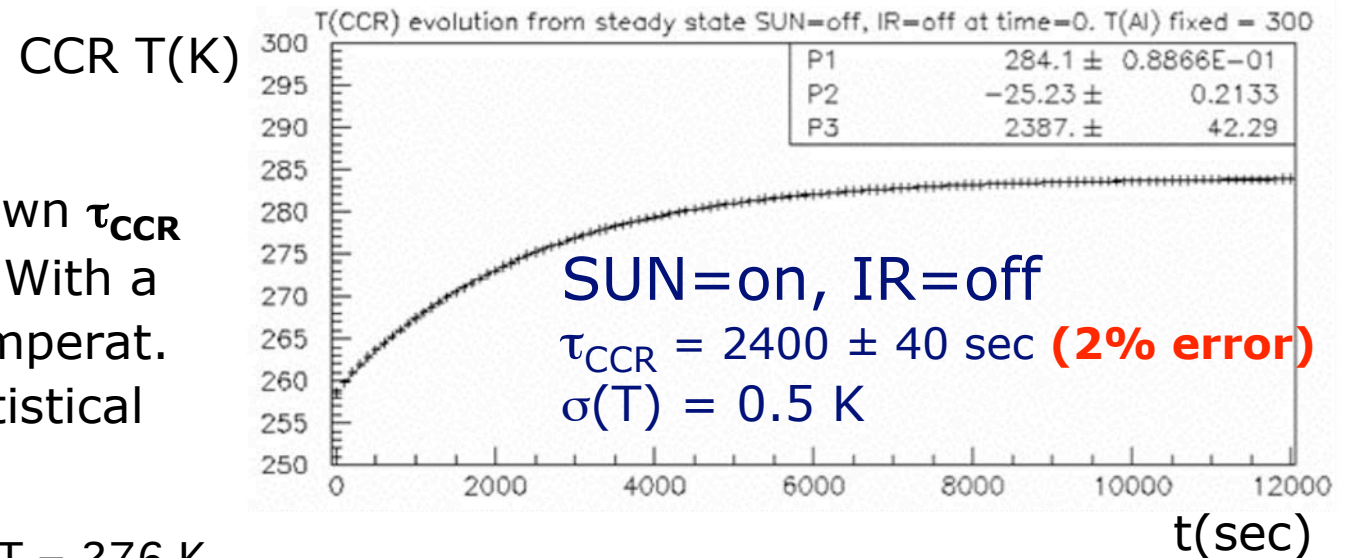
(alodized back)

Solar beam  
shroud

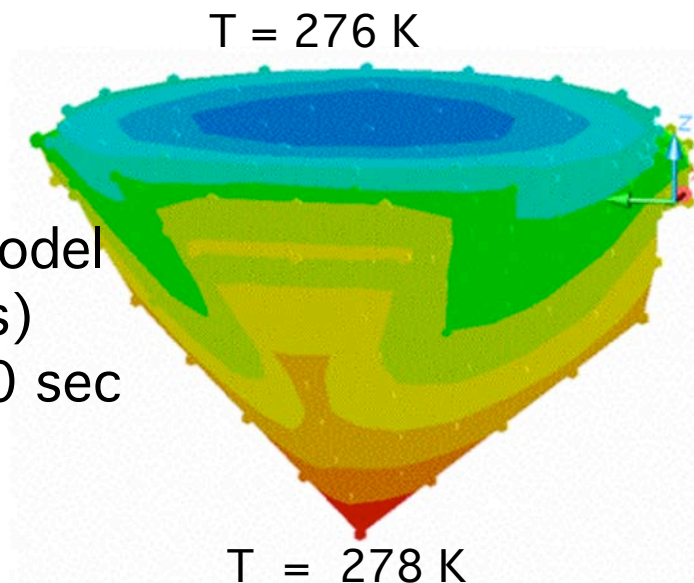
# Simulation of CCR warm-up/cool-down times



Goal: measure unknown  $\tau_{\text{CCR}}$  at  $\leq 10\%$  accuracy. With a 0.5 K accuracy on temperature, this is well within statistical reach

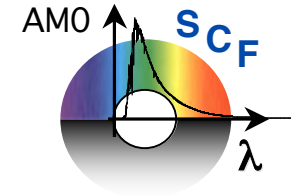


CCR FEM model  
(250 nodes)  
at t = 2800 sec





# Thermometry measurements with IR camera



- **Indoor, in-air test** at room temperature to measure emissivity ( $\epsilon$ ) and reflectivity ( $\rho$ ) of Al and retro-reflectors
- $Q_{\text{camera}} = Q_{\text{emission}} + Q_{\text{reflected}}$
- $T_x$  w/thermocouple
- $T_{\text{bkg}}$ : black disk with controlled temperature = 10 °C or 50°C

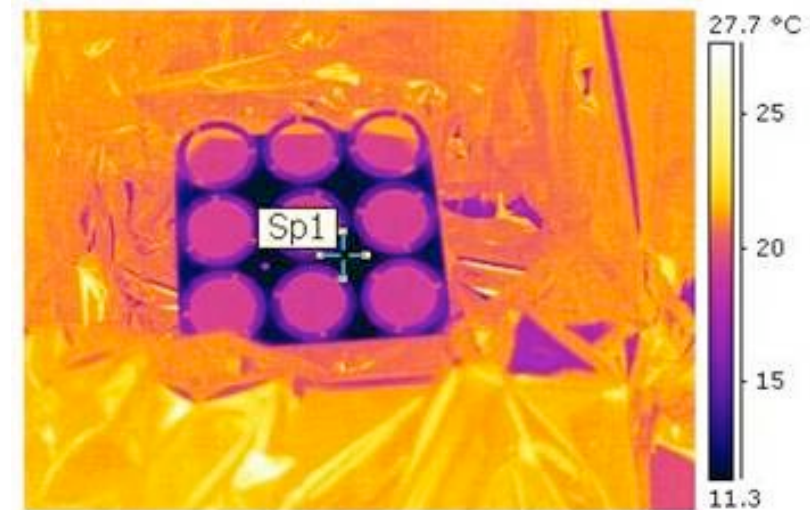
$$\epsilon_{\text{IR}}(\text{CCR}) \sim 0.82$$

$$\rho_{\text{IR}}(\text{CCR}) \sim 0.18$$

$$\epsilon_{\text{IR}}(\text{Al}) \sim 0.15$$

$$\rho_{\text{IR}}(\text{Al}) \sim 0.85$$

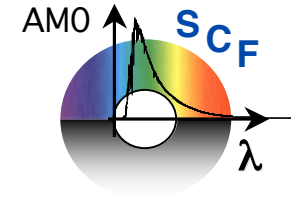
IR pictures of the LAGEOS array



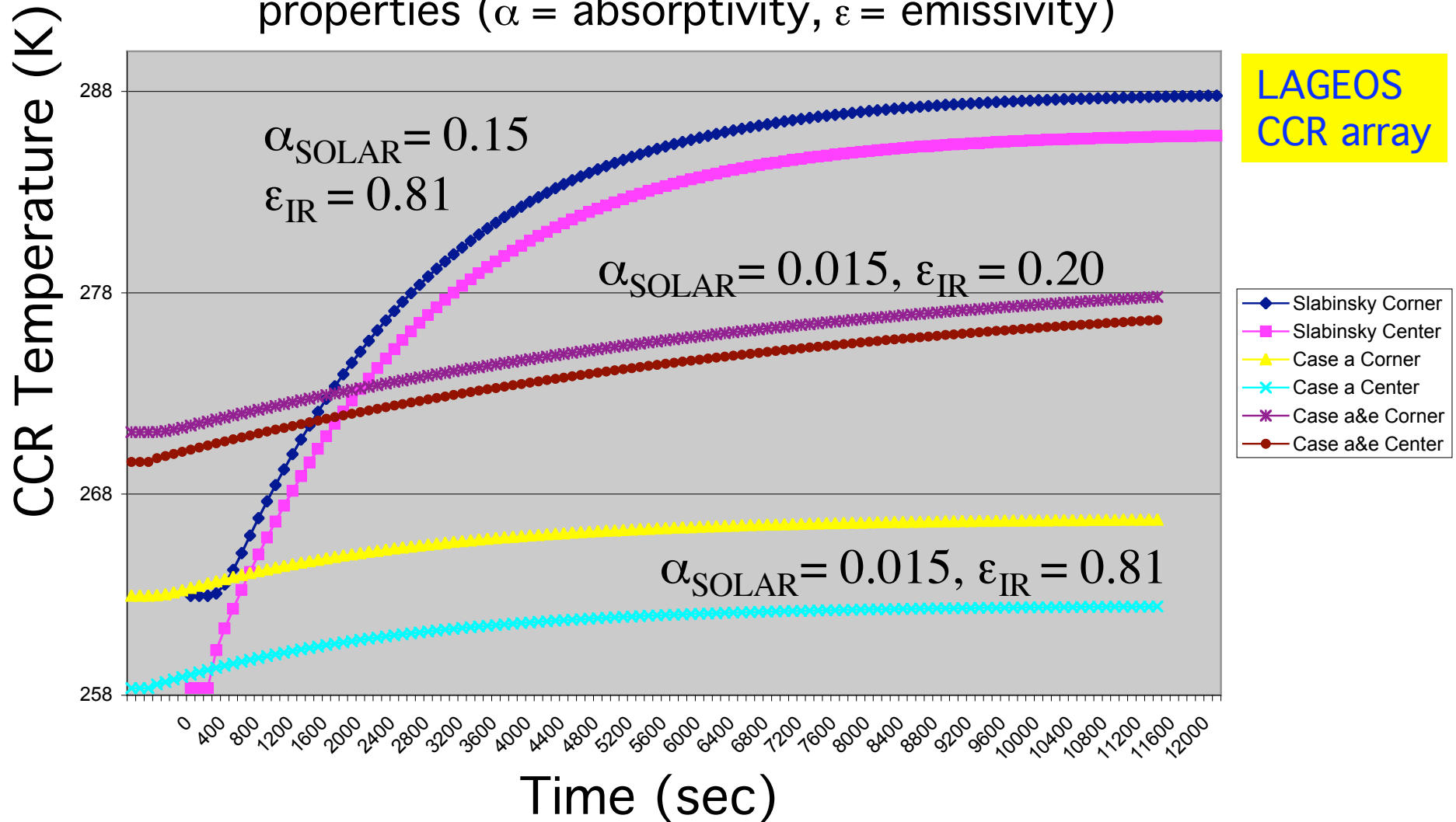
Black disk  
At 10 or 50 °C

LAGEOS array

# Warm-up time vs thermo-optical properties

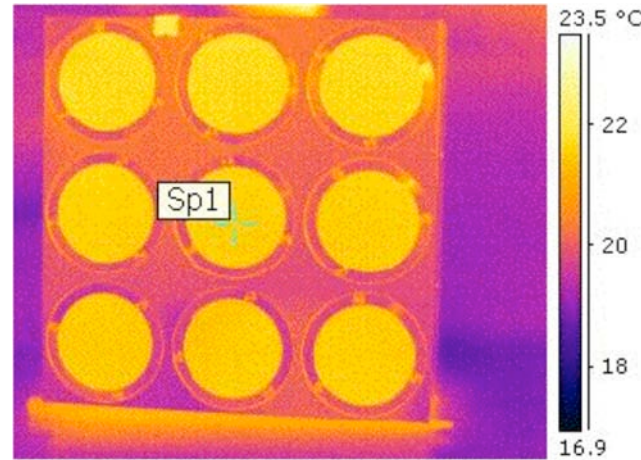
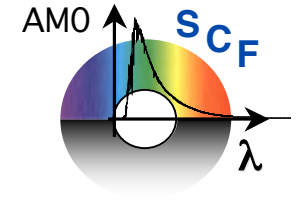


Simulation for different suprasil (CCR) thermo-optical properties ( $\alpha$  = absorptivity,  $\varepsilon$  = emissivity)

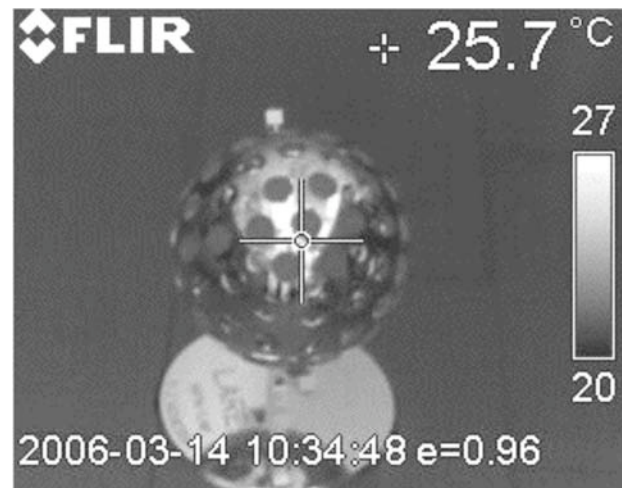




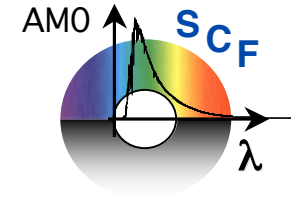
# LAGEOS matrix, LARES 1:2 proto built at LNF



IR images

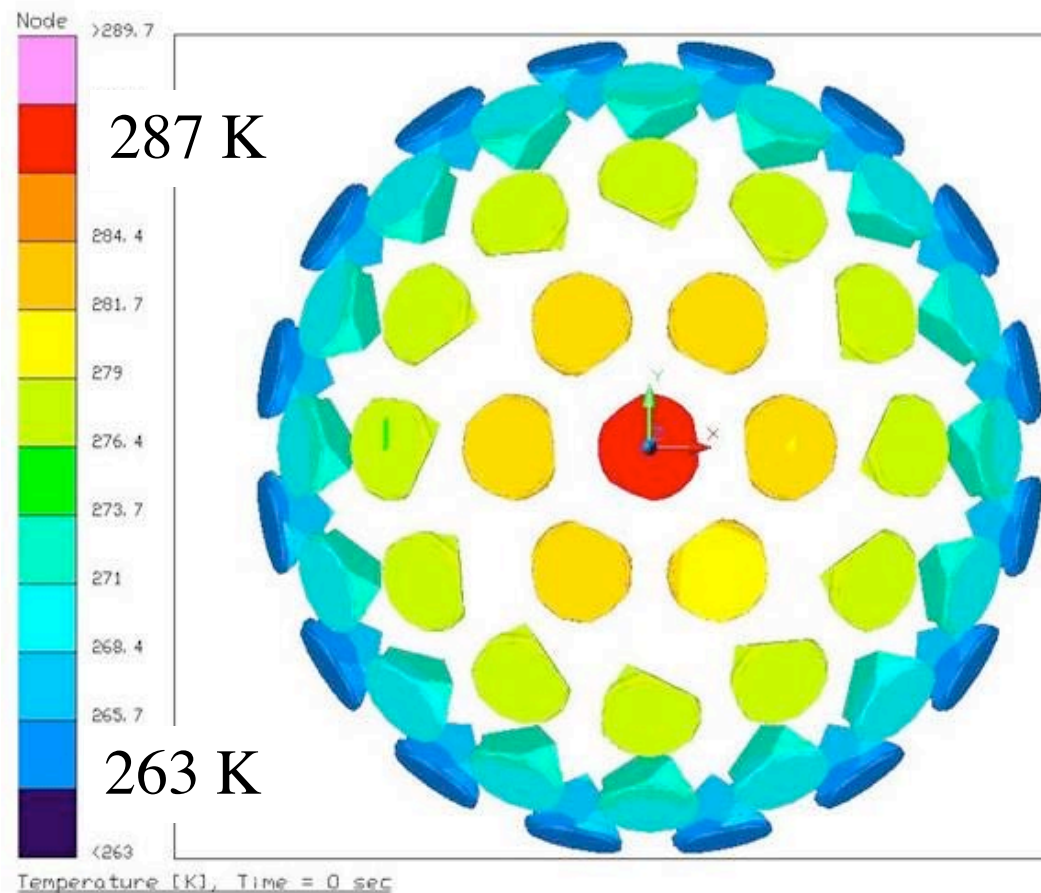


# FE model & thermal simulation of LARES

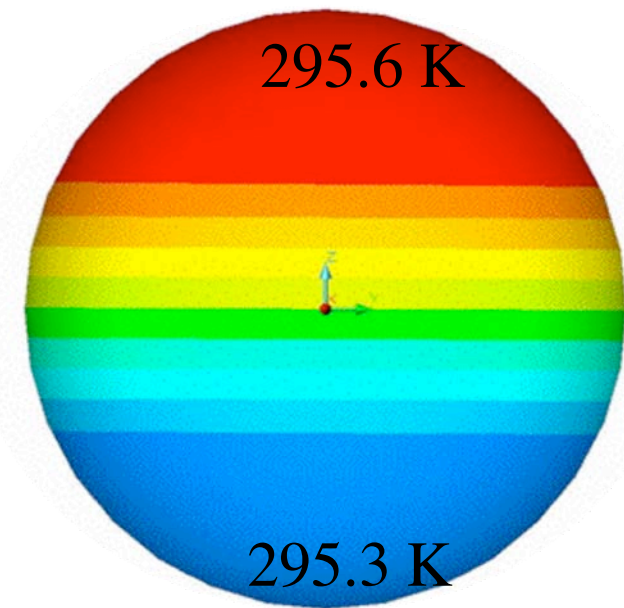


- New shell-over-the-core design
- Model with 15000 nodes. Being optimized
- Steady state with LARES in front of a solar lamp

CCRs, front view

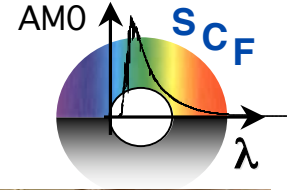


Core, side view

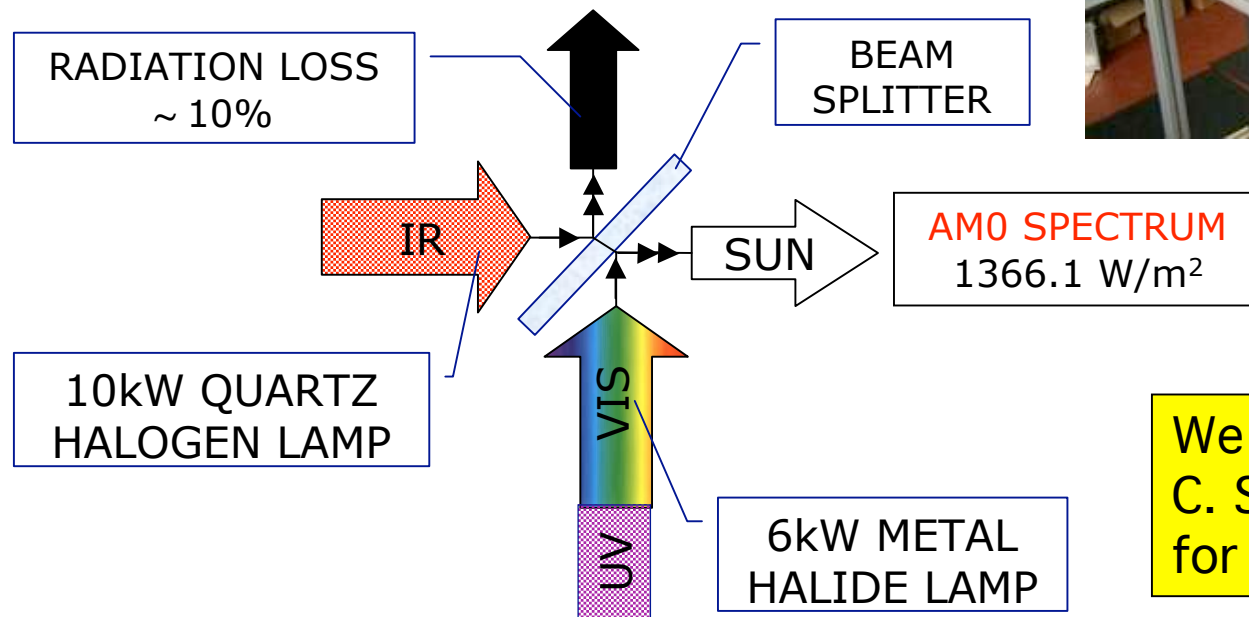
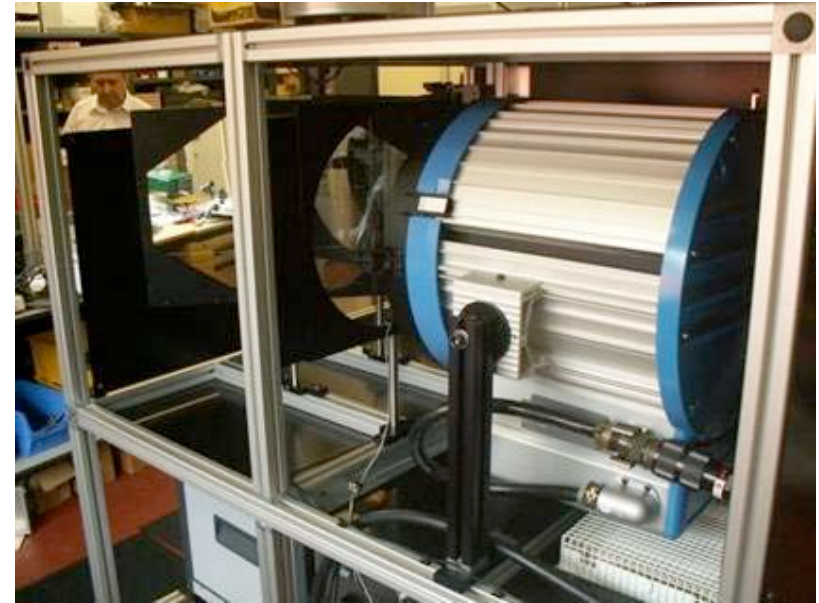




# Status of the SCF

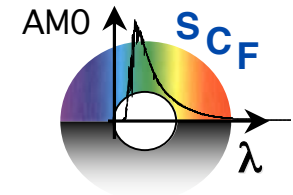


- Solar simulator acceptance test at TS-Space (UK) successful; being shipped to LNF
- Outgassing performed, Nitrogen transfer line installed



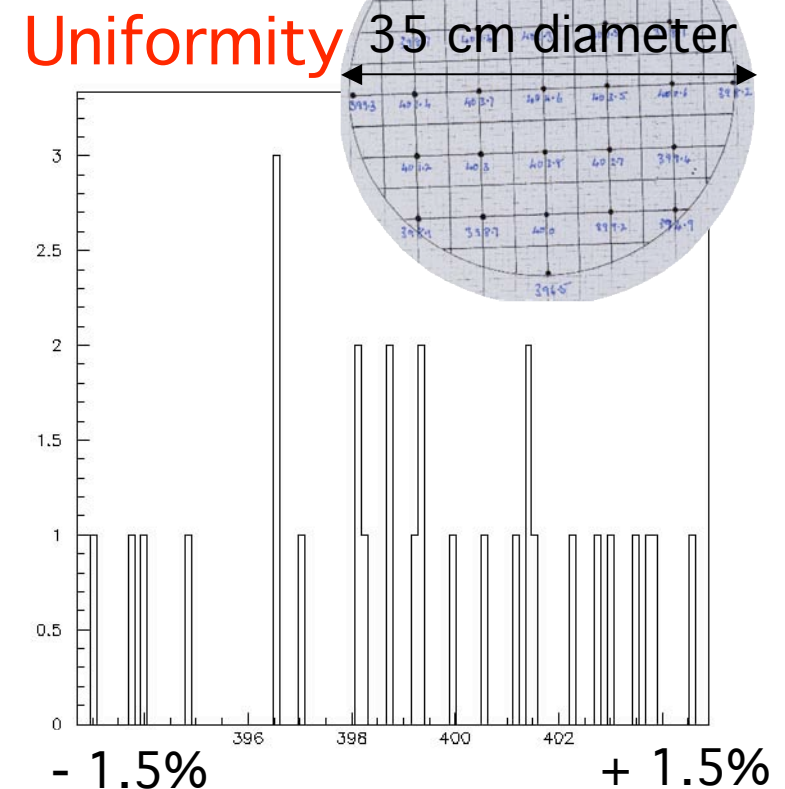
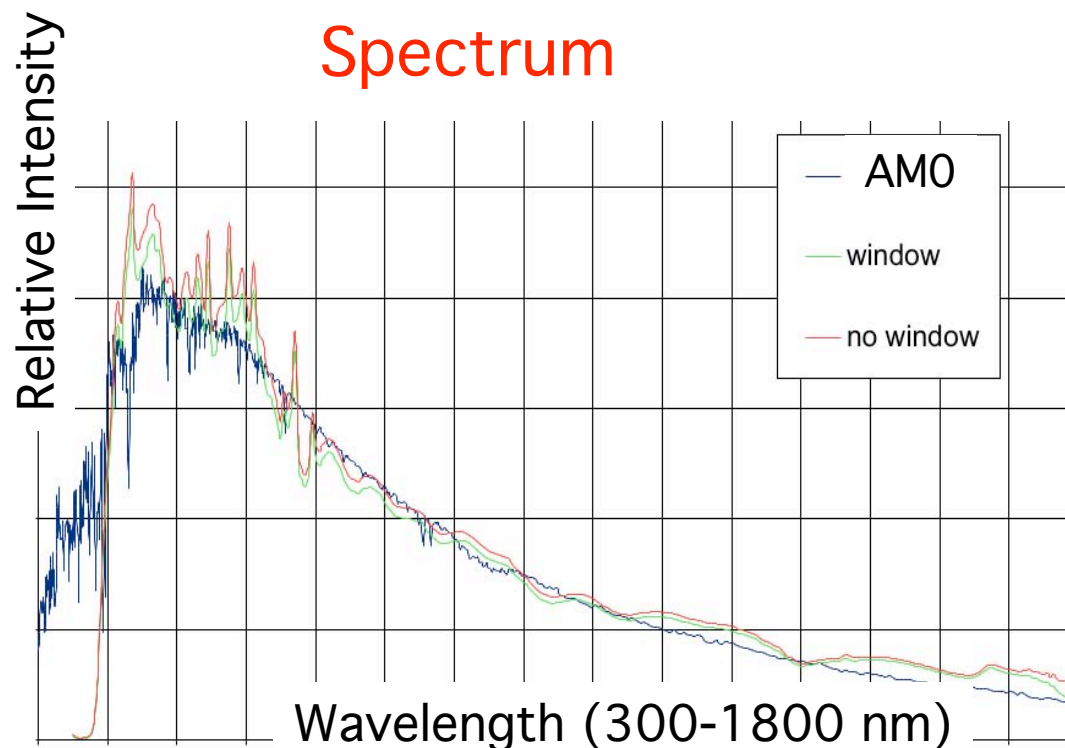
We deeply thank M. Calvetti, C. Sanelli, M. Curatolo of LNF for their support to the SCF !

# Measured Solar Simulator spectrum & uniformity

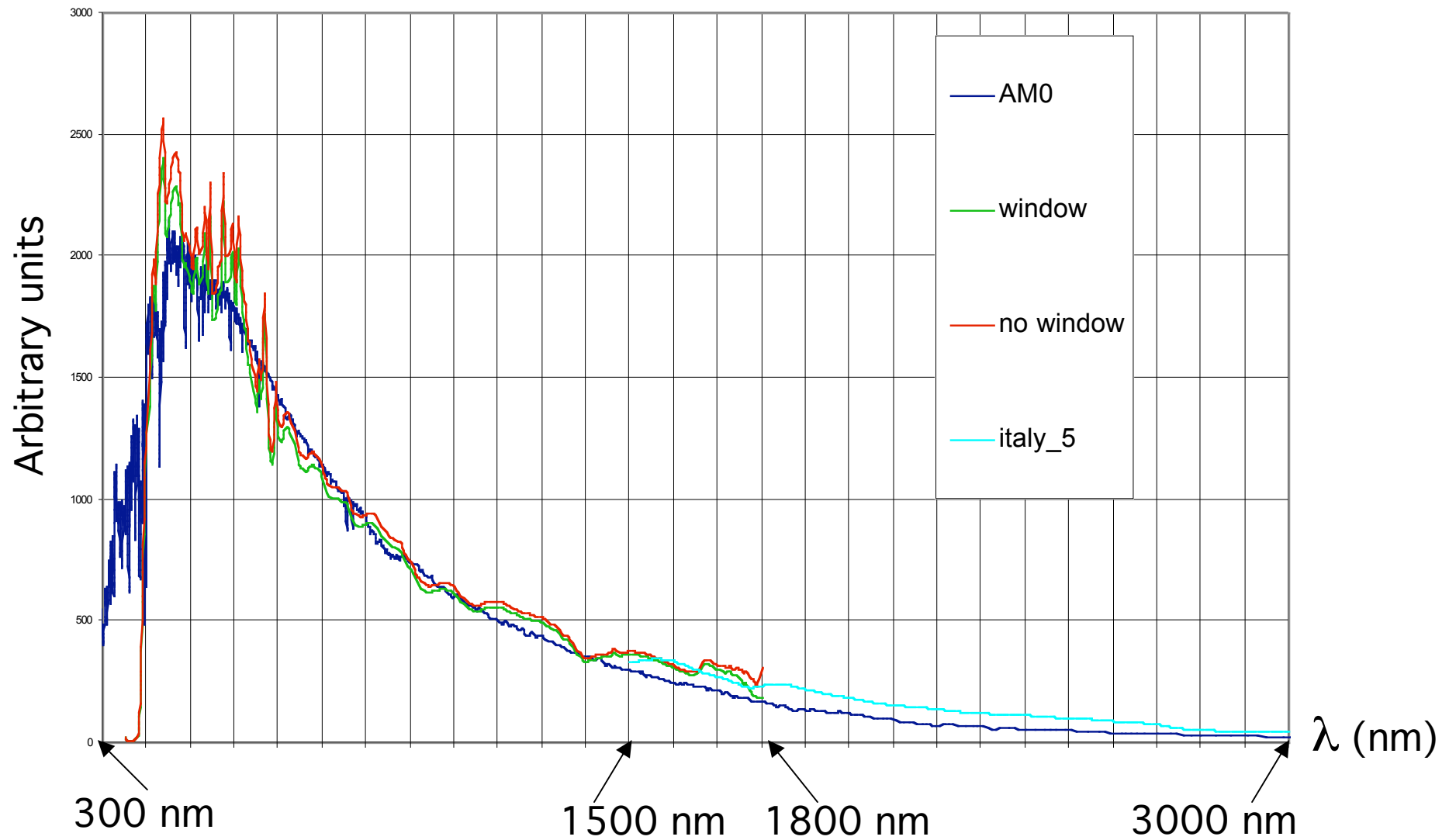
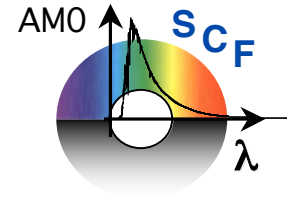


- “AM0” standard spectrum (400-3000 nm)
- Absolute calibration @1% w/Solarimeter
- HV adjusted for lamp ageing w/PIN diode

ACCEPTANCE TEST  
MAY 29, 2006 OK.  
Being shipped to LNF

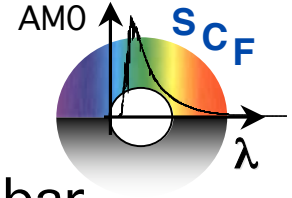


# Extended **AM0** spectrum (300 - 3000 nm)

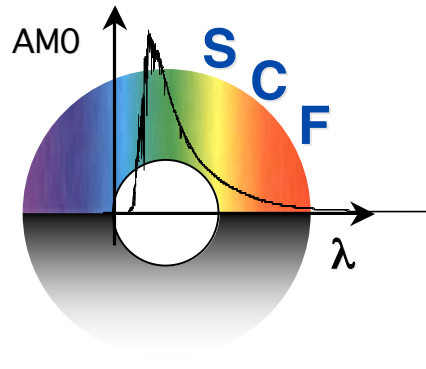




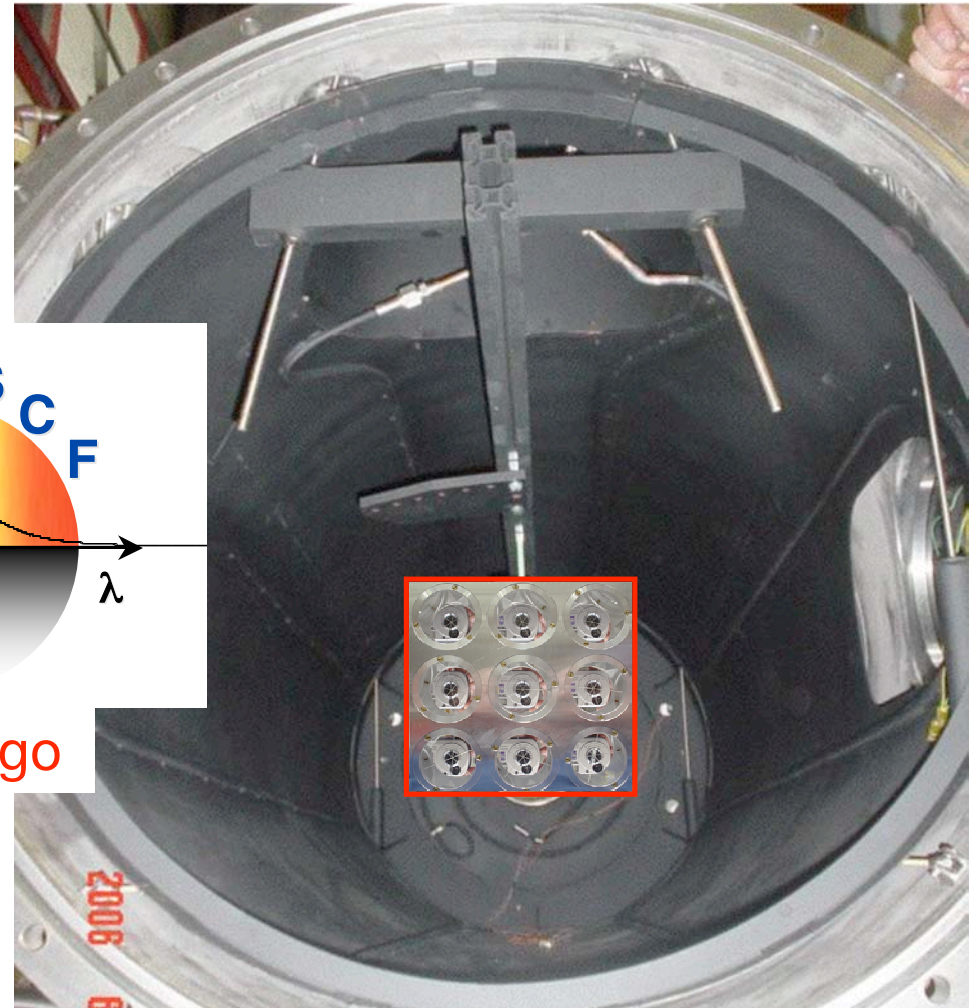
# LNF SCF commissioning well underway



T probes installed, outgassing performed, vacuum @  $10^{-5}$  mbar.  
Liquid N<sub>2</sub> TL connected this week, then real MEASUREMENTS !!!

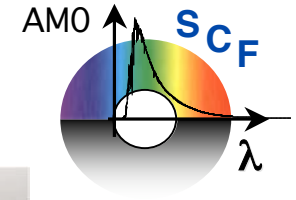


SCF logo





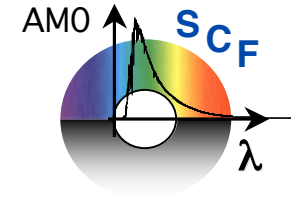
# LNF SCF commissioning



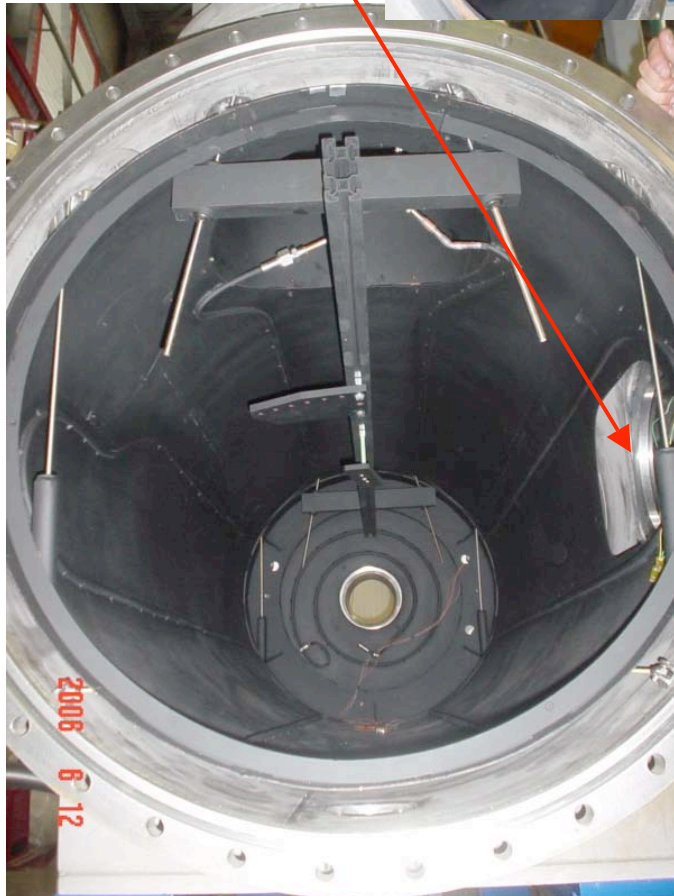
Liquid N<sub>2</sub> TL  
connected



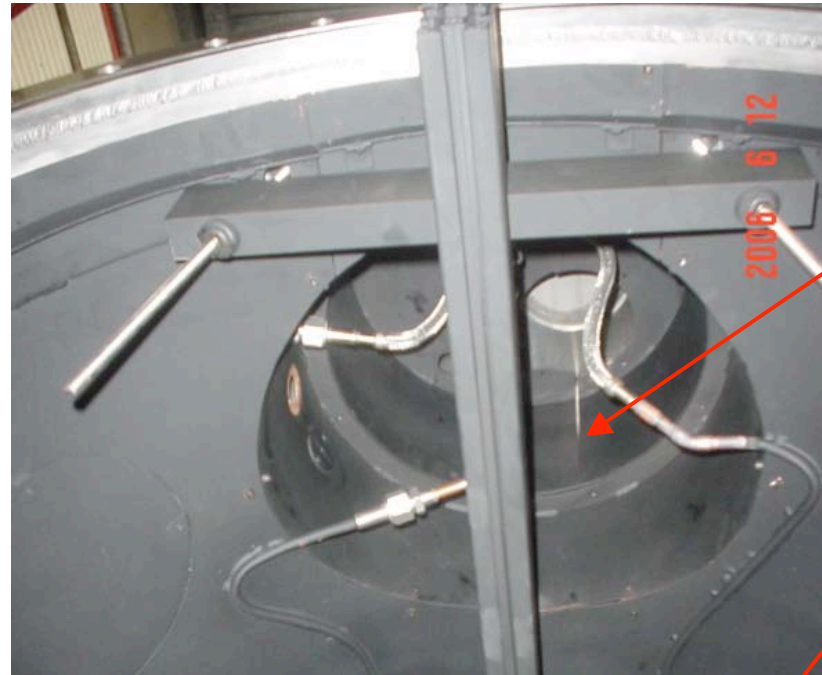
# Inside the LNF SCF



Side tunnel  
for IR  
camera



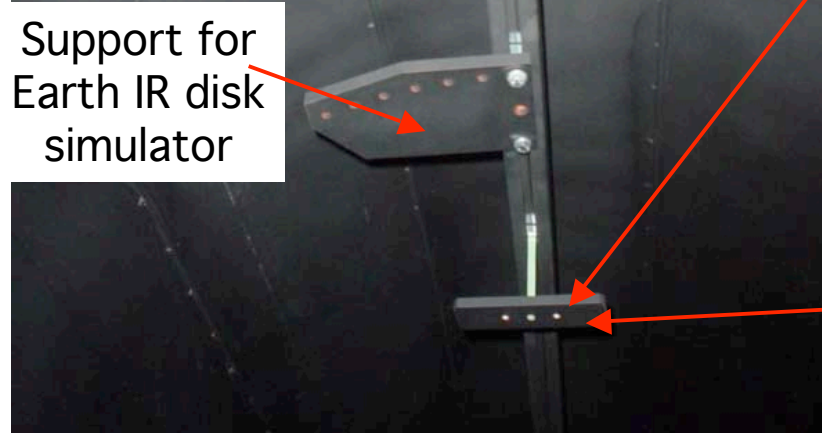
Service  
turret



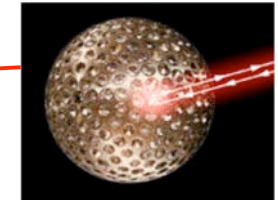
Support for  
GNSS array...



Support for  
Earth IR disk  
simulator

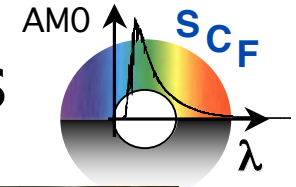


... or DSGP  
spherical  
test-mass





# Optical characterization of retro-reflectors

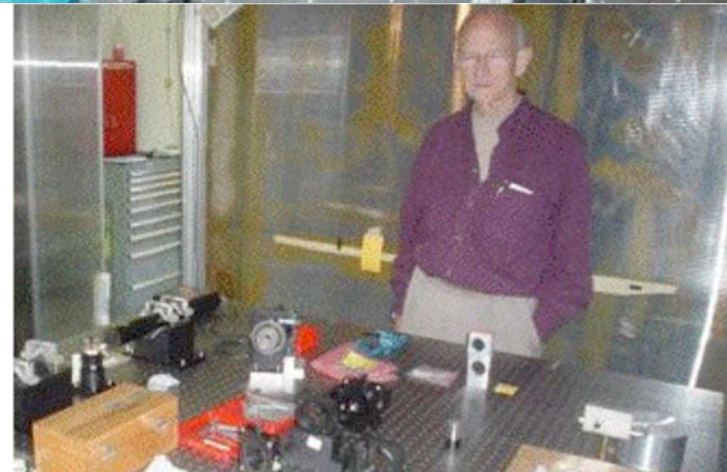


## Test 1: Far-Field Diffraction Pattern (FFDP)

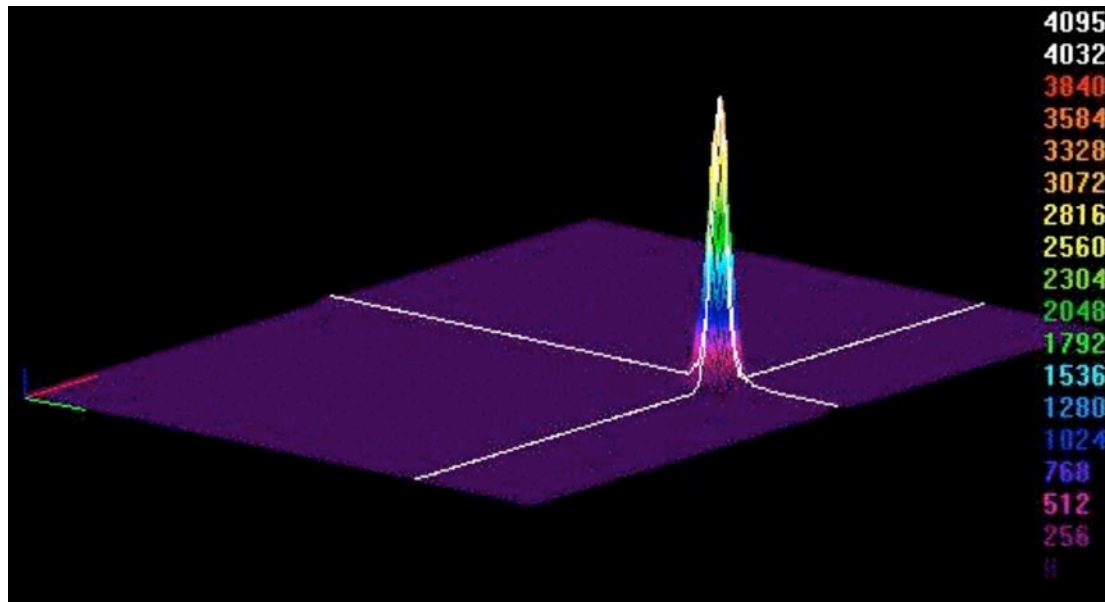
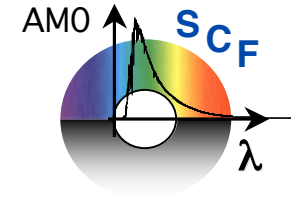
- “Optical FLAT” (mirror) for normalization
- CCDs as laser beam profilers

Repeat test inside the SCF

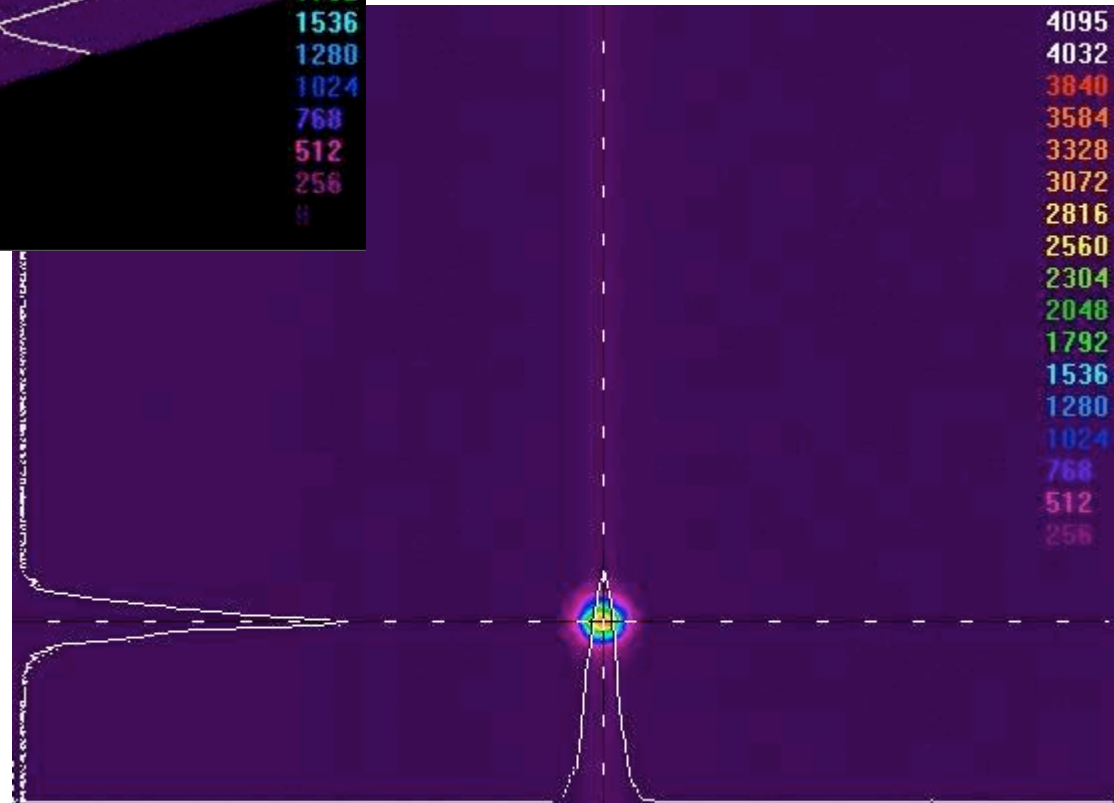
Thanks to John Degnan, Dave Arnold, Erricos Pavlis, Jan McGarry for advise and to Doug Currie (in photo) for help on setting up the optical tests at LNF



# He-Ne laser beam readout by CCD

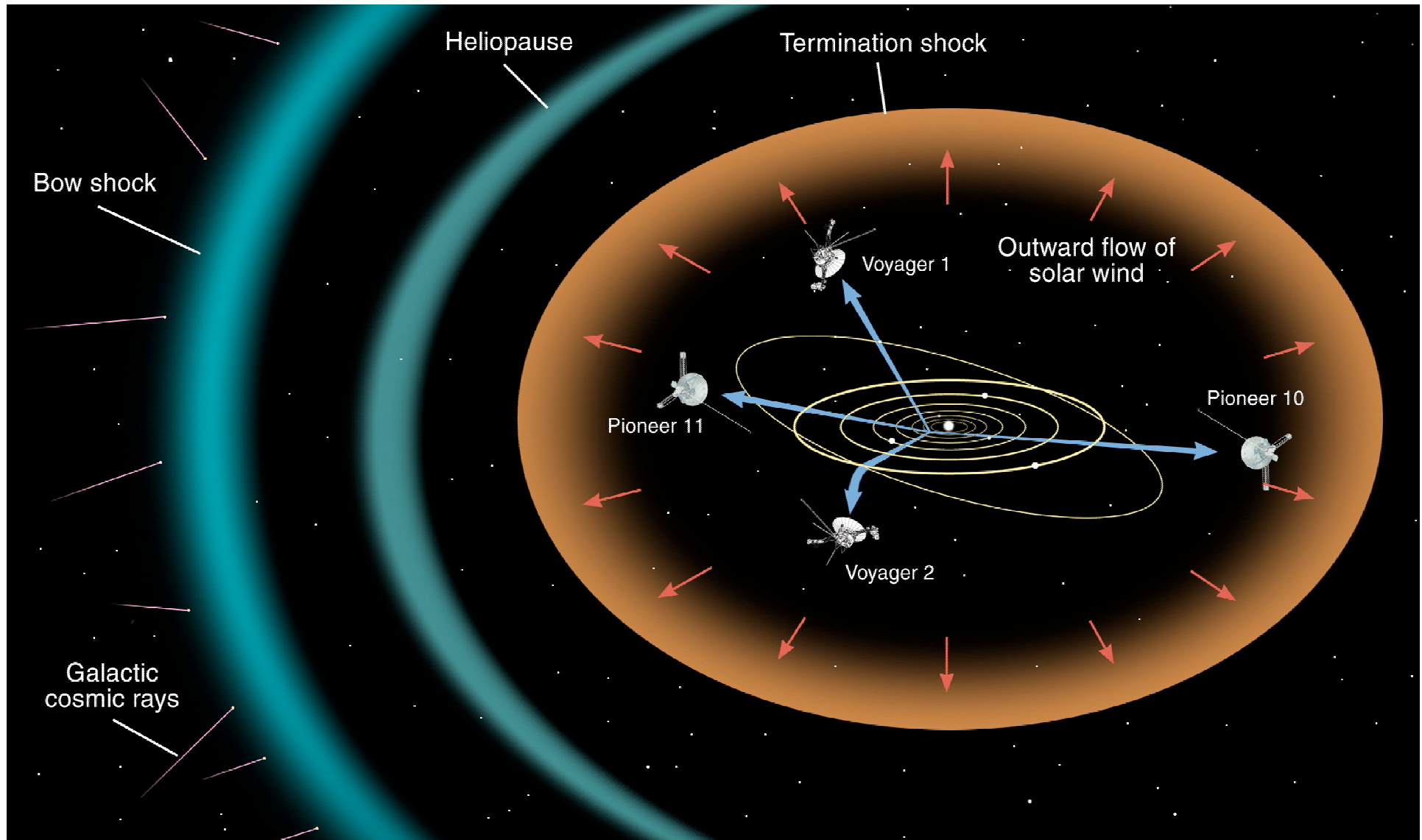
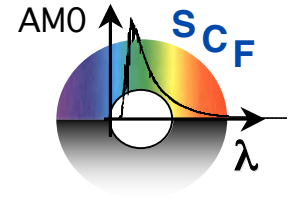


A. Lucantoni, Rome I Univ.  
(LARES student of A. Paolozzi)

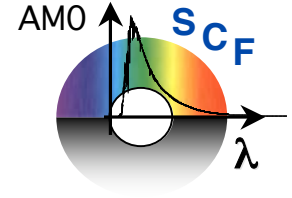




# Enough with NEO. Let's go to Deep Space



# Probing Gravity in Deep Space: the Mystery of Pioneer Deceleration



- In the outer SS the probes with the most accurate and robust navigation capabilities are the **PIONEERS**. Why ? Because:
  - Pioneers were spin-stabilized => months without firing thrusters
  - **VOYAGERS**: factor 50 less accurate, because 3-axis stabilized
  - **GALILEO**: less accurate and up to Jupiter only
  - **CASSINI**: analysis not accurate enough
  - Outer planets: less accurate than Pioneers
- Pioneer doppler data ('87-'98, 40-70.5 AU) provide clear anomalous deceleration

$$a_{PI0} = (8.74 \pm 1.33) \times 10^{-10} \text{ m/s}^2$$

~10 x maximum **LAGEOS** thermal accelerations !!

- Effect of asymmetric thermal forces ?
  - forward-backward asymmetry in thermo-optical parameters ?
- Was it the **RTGs** ? Or is  $1/r^2$  violated by **New Physics** ?

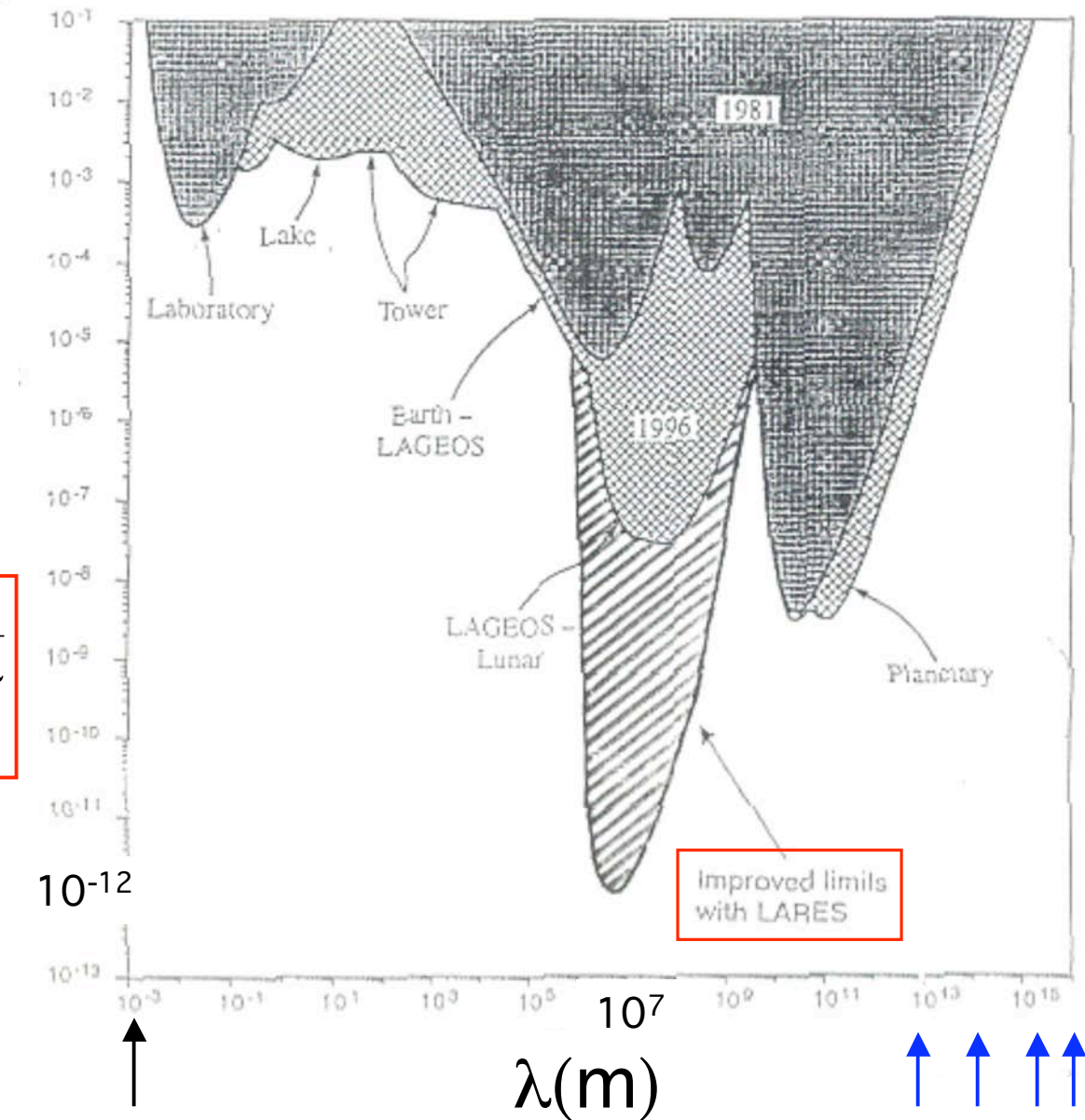


Radioisotope  
Thermoelectric  
Generators (RTGs)

Test of  $1/r^2$  law and  
of new long range  
interactions, ie, a  
Yukawa Potential

$$V_{YUK} = -\alpha \frac{GM_{earth}}{r} e^{-\frac{r}{\lambda}}$$

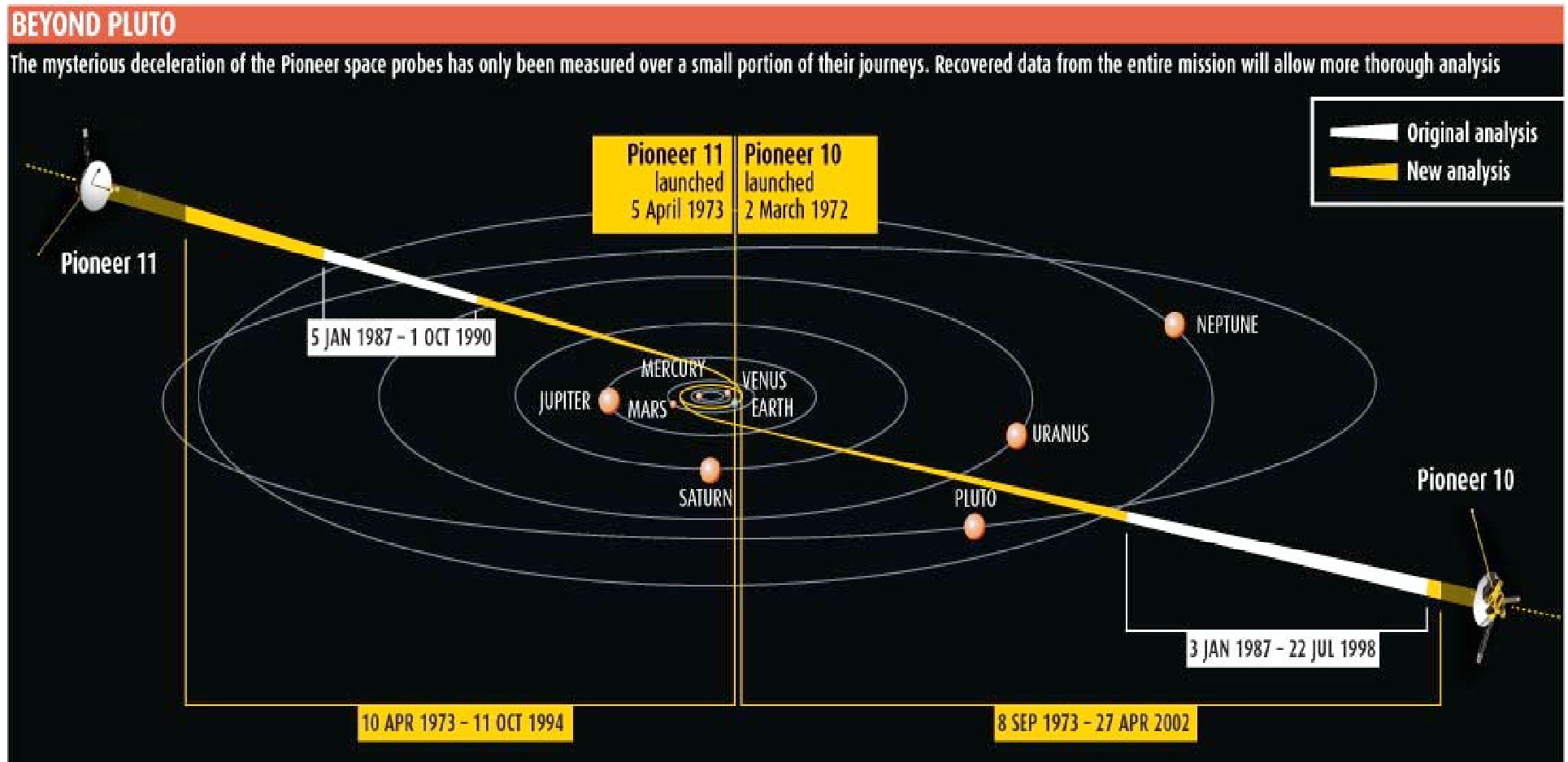
$\alpha$



Eot-Wash  
experiments

Pioneer  
region

# Status of the analysis of Pioneer data



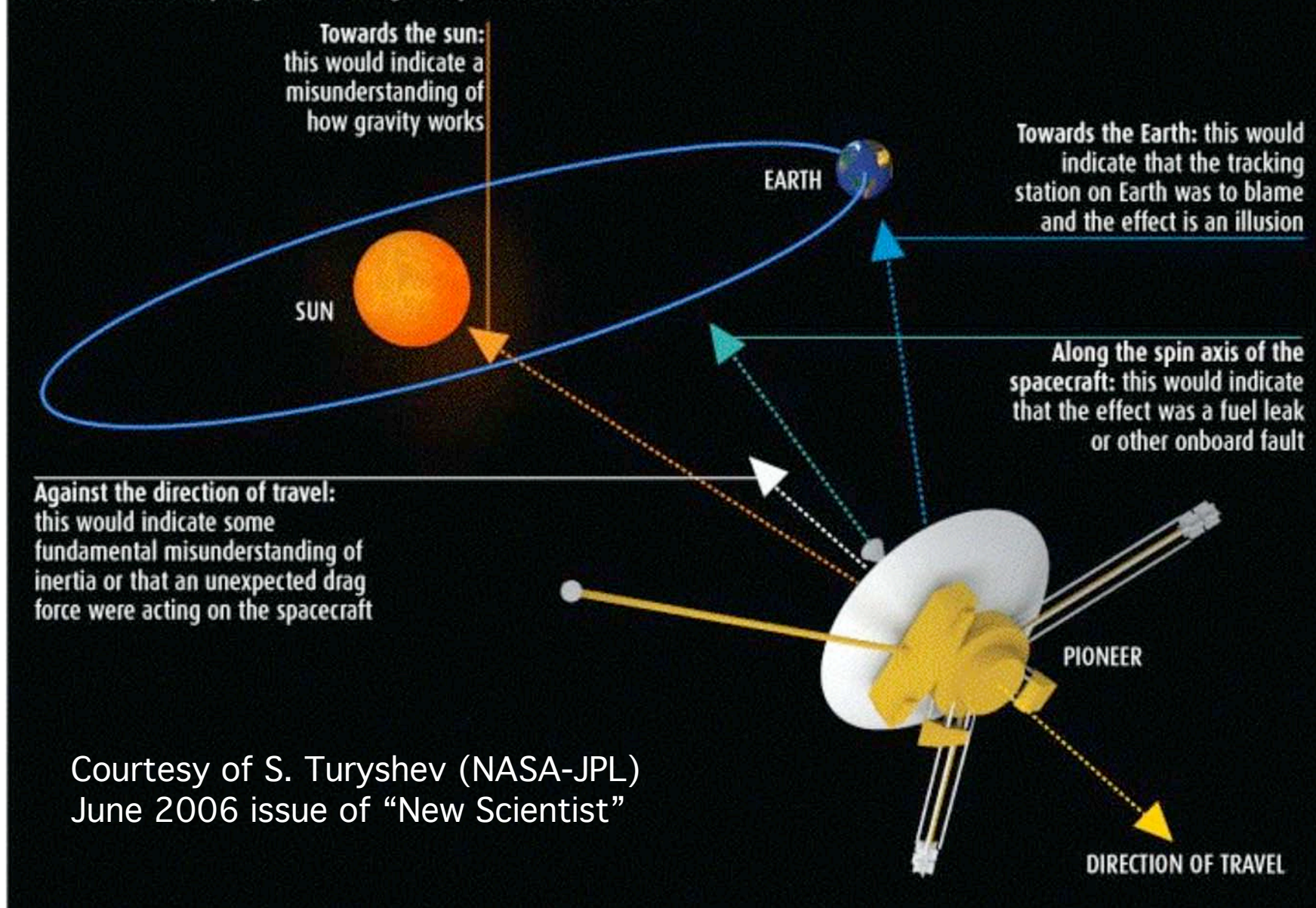
Courtesy of S. Turyshev (NASA-JPL) - June 2006 issue of "New Scientist"



# Status of the analysis of Pioneer data

## PIONEER POSSIBILITIES

The Pioneer anomaly might be working in any one of four directions



Courtesy of S. Turyshev (NASA-JPL)  
June 2006 issue of "New Scientist"





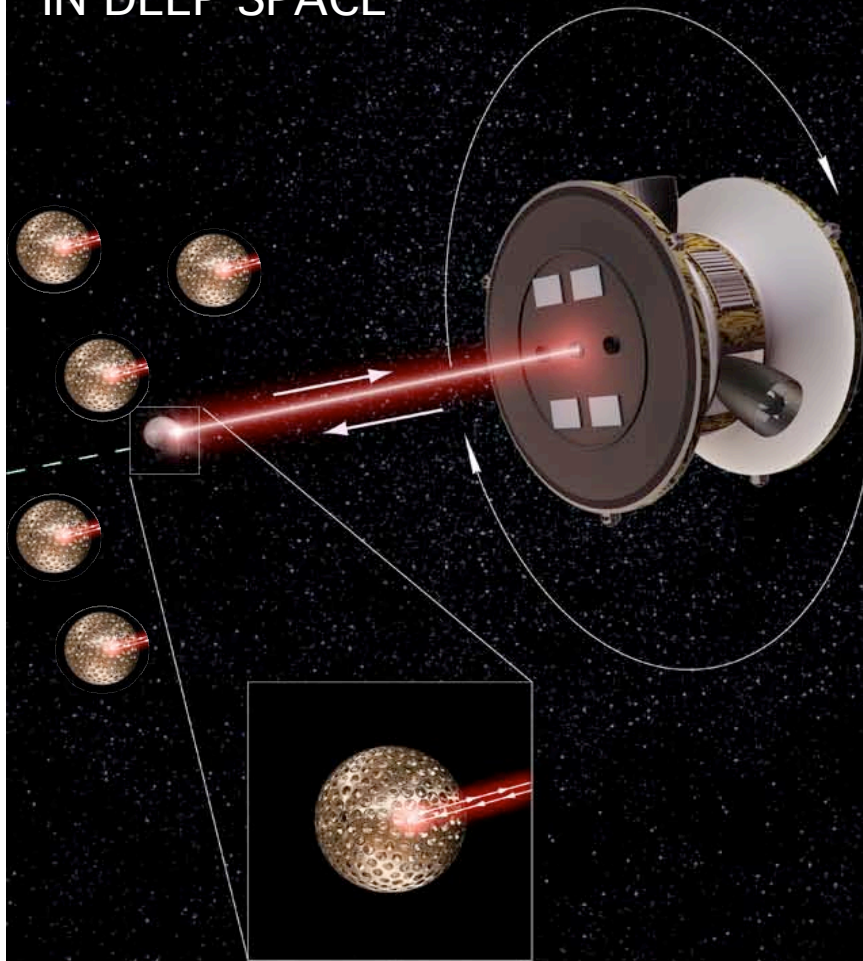
A MISSION TO EXPLORE THE PIONEER ANOMALY



# Measurement Concept: Formation-flying

A CONSTELLATION OF SMALLER “LARES”  
IN DEEP SPACE

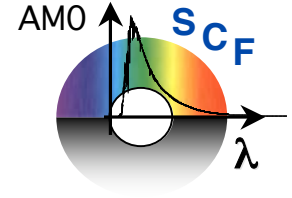
Courtesy of  
S. Turyshev (JPL)



- Active spacecraft and passive test-mass
- **Objective: accurate tracking of test-masses**
- 2-step tracking: common-mode noise rejection
  - Radio: Earth → spacecraft
  - Laser: spacecraft → test-mass
- Flexible formation: distance may vary
- The test mass is at an environmentally quiet distance from the craft, > 250 m
- Occasional maneuvers to maintain formation

# The Deep Space Gravity Probe mission

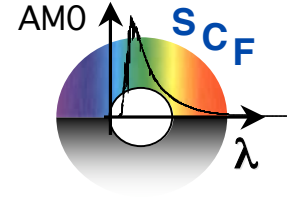
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- Study of the **Pioneer anomaly** with a few lighter “LARES”
- Different masses/materials to test **Equivalence Principle**
- Planet flybys for **planetary science**
- **Thermal forces**; here we can contribute
  - **Solar constants** beyond Saturn  $\leq 10^{-2} \times \text{NEO-AM0}$ .
  - **IR radiation by planets**. Disks with varying  $\varnothing$  and Temperature
  - Measure thermal properties in SCF, then use **orbital simular** and thermal sw for full 10-80+ AU orbit
- LAGEOS thermal accelerations  $\leq 1/10 \times a_{\text{PIO}}$  ! Our expertise with the **high-accuracy thermal** characterization of LARES at the SCF will be extremely useful for the design of DSGP

# Conclusions

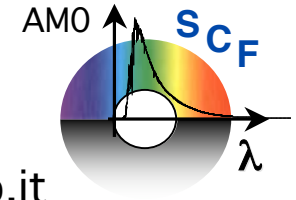
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- ETRUSCO is a **truly international, inter-agency, interdisciplinary** project of space research, in NEO and Deep Space
- **ETRUSCO builds upon the LNF SCF**, which has been funded for the LARES space mission by INFN-GR2 + LNF
- The LNF SCF fills a research “niche” in the field of **experimental tests of General Relativity**
- GNSS project: good potential for **high-tech applied research**. GALILEO is maybe the most important EU/ESA enterprise
- We propose to INFN-GR5 this program and requests funding for:
  - Dedicated software license for thermal analysis
  - GNSS: innovative hollow retro-reflector prototypes
  - Pioneer/DSGP: modest upgrade of SCF for Deep Space conditions (liquid Nitrogen not enough, since at Saturn distances,  $T(\text{test mass}) < 77 \text{ K}$ )
  - Pioneer/DSGP: construction of prototype test mass



# Summary of ETRUSCO LNF resources



Picture from [www.apollodiveio.it](http://www.apollodiveio.it)

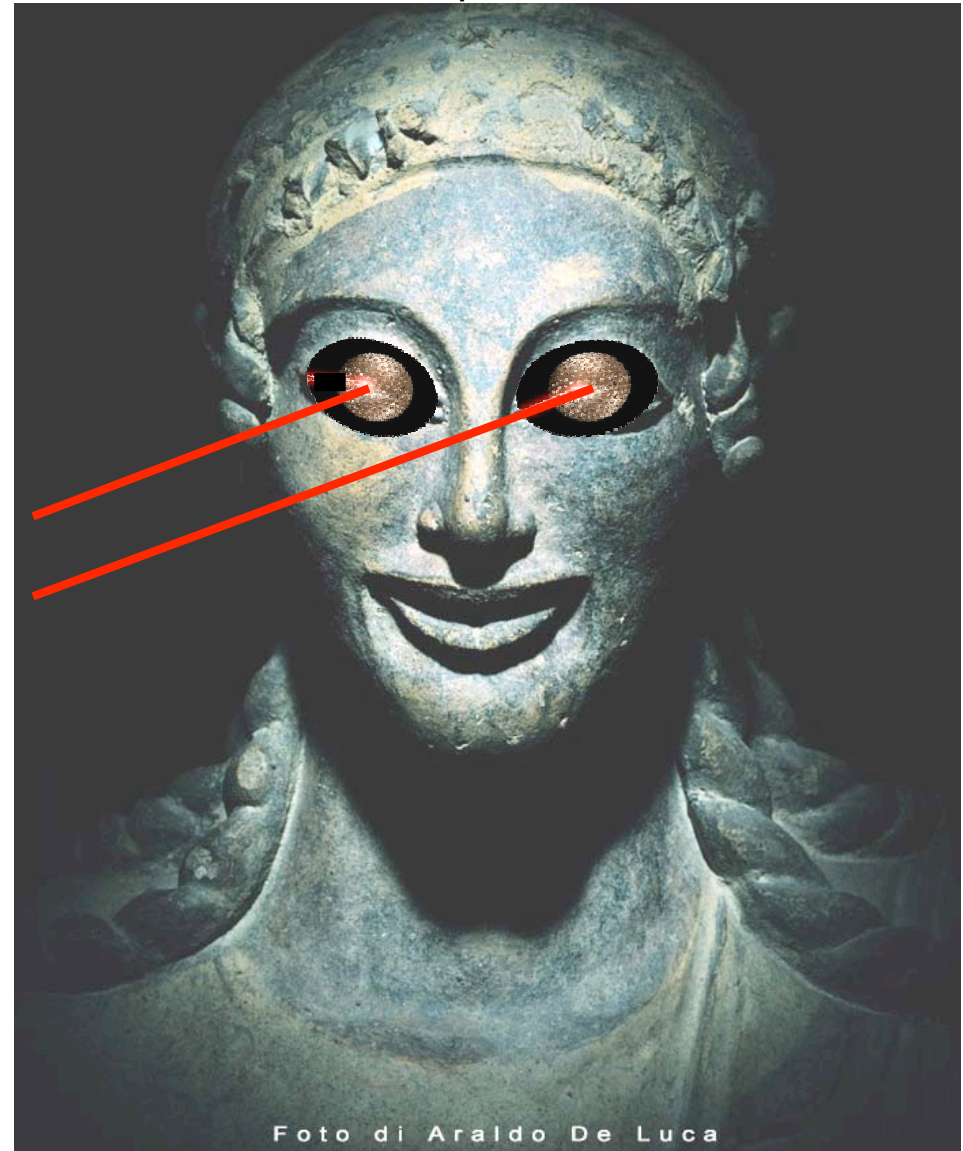
## GROUP (all “associated” to INFN-LNF)

- S. Dell’Agnello, INFN-LNF  
(30%, responsabile)
- G. Delle Monache, INFN-LNF  
(30%, head Cryogenics Service)
- C. Cantone, INFN-LNF  
(30%, borsista INFN)
- M. Garattini, INFN-LNF  
(30%, borsista INFN)
- G. Bellettini, Roma2  
(30%, Prof. Ord.)

TOTAL = 1.5 FTE

## Requests to LNF (per year)

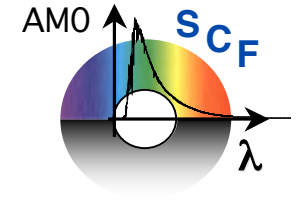
- Cryogenics Service: 3 mesi uomo
- Mechanics Service: 3 mesi uomo





# Financial request to INFN-GR5

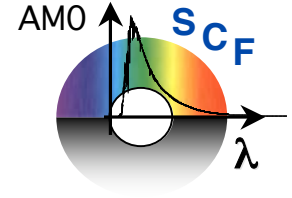
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- Dedicated license of software for thermal analysis (SINDA-Fluint/RadCad/ThermalDesktop by C&R Tech) 15 K€
- GNSS: innovative hollow retro-reflector prototypes (bolted Al and glued types by PLXinc, PROSystems) 15 K€
- Upgrade of NEO SCF for Deep Space
  - $T(\text{test-mass}) < 77 \text{ K}$ , the limit of the current liquid nitrogen system; need to get towards 10 K; SHI Cryocooler SRDK-408D2-W71D by TECO 25 K€
- Deep space test masses prototypes
  - DSGP-specific retro-reflectors 10 K€
- Optics consumables 5 K€
- Mechanics consumables 5 K€

# Missioni interne ed estere

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- Missioni estere

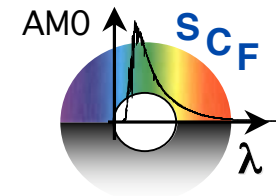
- Meeting e misure per il GNSS coi collaboratori ILRS, NASA-GSFC, UMCP (tutti vicino a Washington DC) 5 K€
- Training sull'analisi dati del Pioneer presso gli esperti del JPL di Pasadena (una volta sola) 5 K€

- Missioni interne

- Meeting con colleghi teorici di INFN-Firenze e del Dip. di Fisica del Politecnico di Torino sull'impatto del GNSS sulla Relativita' Generale 2 K€
- Misure ottiche e test di laser ranging sui prototipi dei LNF col laser impulsato di potenza del Centro di Geodesia Spaziale di Matera dell'ASI (MLRO). Trasporto prototipi e strumenti 3 K€
  - MLRO e' una delle migliori stazioni di laser ranging del mondo

# Total financial requests

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- Inventariabile (retros/test mass protos) 25 K€
- Costruzione apparati (SCF upgrade) 25 K€
- Consumi (software, mechanics, optics) 25 K€
- Missioni interne (ASI-Matera, teorici) 5 K€
- Missioni estere (NASA-GSFC, NASA-JPL) 10 K€

Totale 90 K€