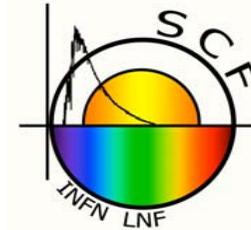


Probing Gravity with 2nd Generation Lunar Laser Ranging



Manuele Martini (INFN-LNF)
for the MoonLIGHT Collaboration

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R. March

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University of California at San Diego, CA, USA

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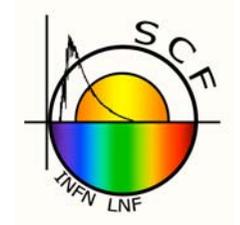
A. Hajian

U. S. Naval Observatory, Washington DC, USA

APOLLO Lunar Laser-Ranging Observatory (T. Murphy et al), Los Alamos, USA

LNF Spring School, May 2008

Outline



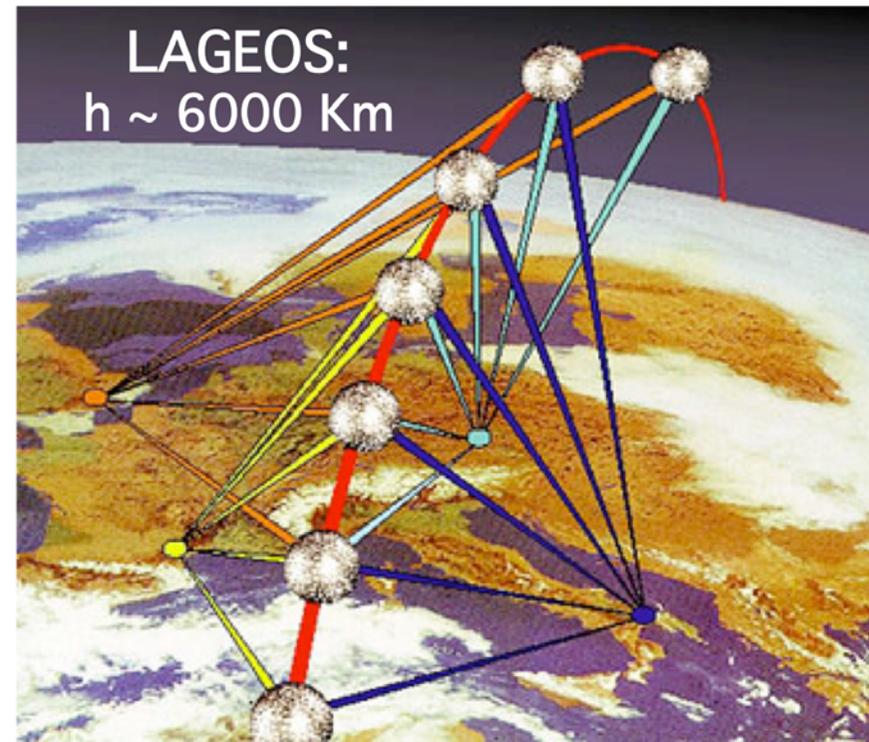
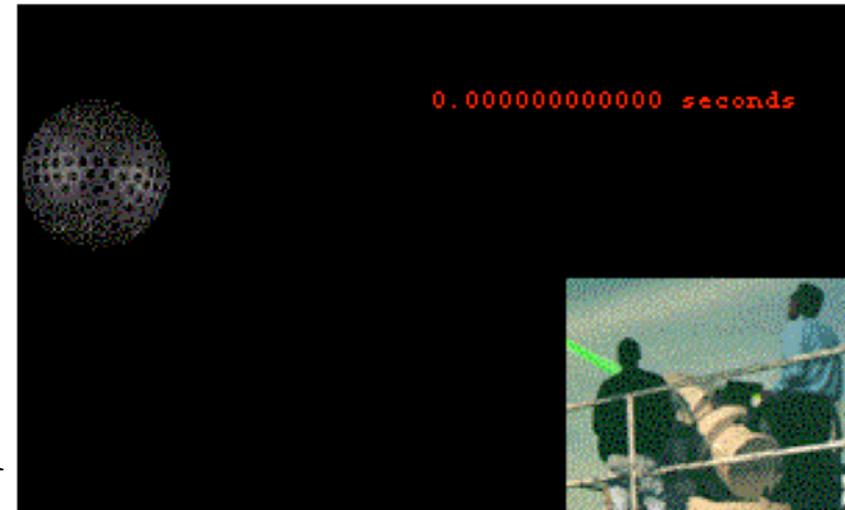
- Science with Lunar Laser Ranging (LLR)
- **MoonLIGHT**: a study for 2nd generation Lunar Laser Ranging (LLR) approved in Aug 2007 by NASA
- Testing new physics
- Conclusions

Satellite Laser Ranging (SLR) Lunar Laser Ranging (LLR) Time of flight measurement



- The Moon as a test mass (1969+, Alley, Bender, **Currie**, Faller ...)
- LAGEOS: “cannon-ball”, point-like test-masses covered with laser retro-reflectors (raw orbit accuracy < 1 cm)

S
L
R

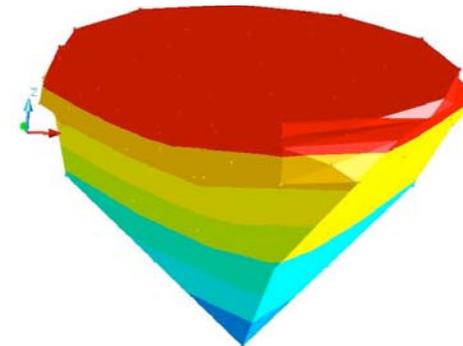


Very critical experimental issues

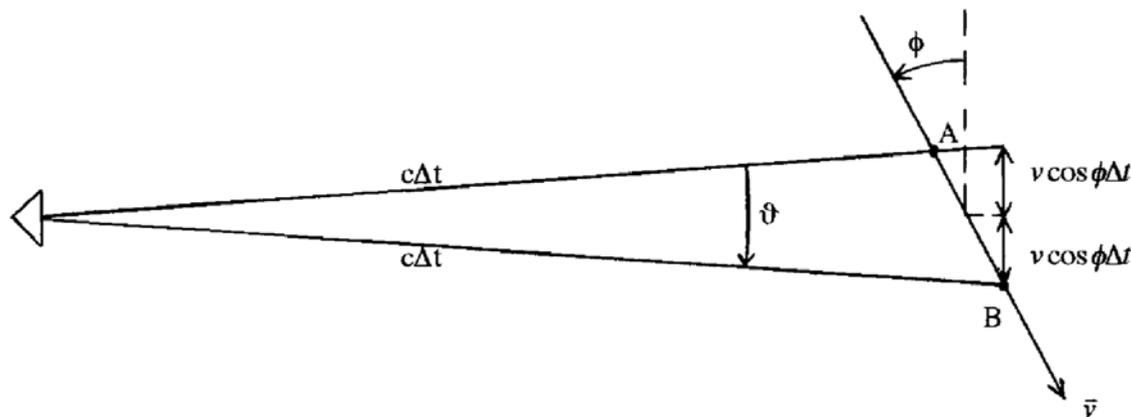


- **Velocity aberration.** Relative station-satellite velocity requires non-zero expensive CCR dihedral angle offsets w/ **0.5 arcsec accuracy**
- Design to control **thermal perturbations** on optical performance
- Characterize “as-built” thermal and optical performance of the INFN dedicated facility

$T(\text{center}) = 139.2 \text{ K}$



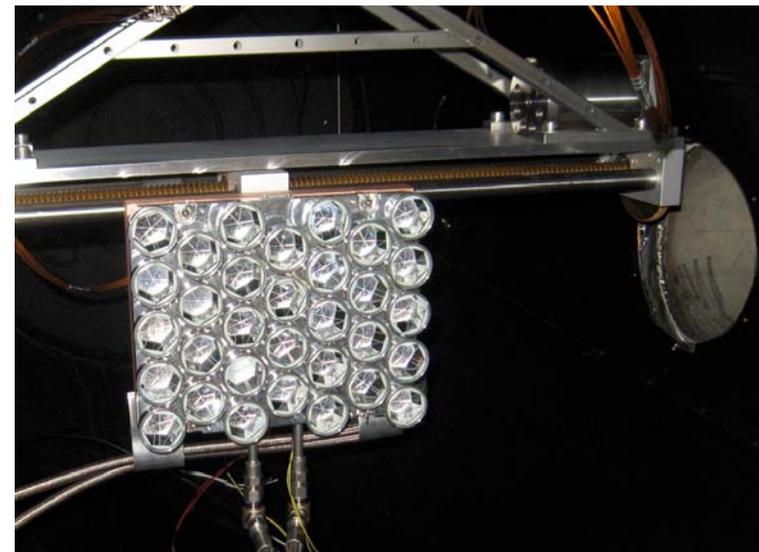
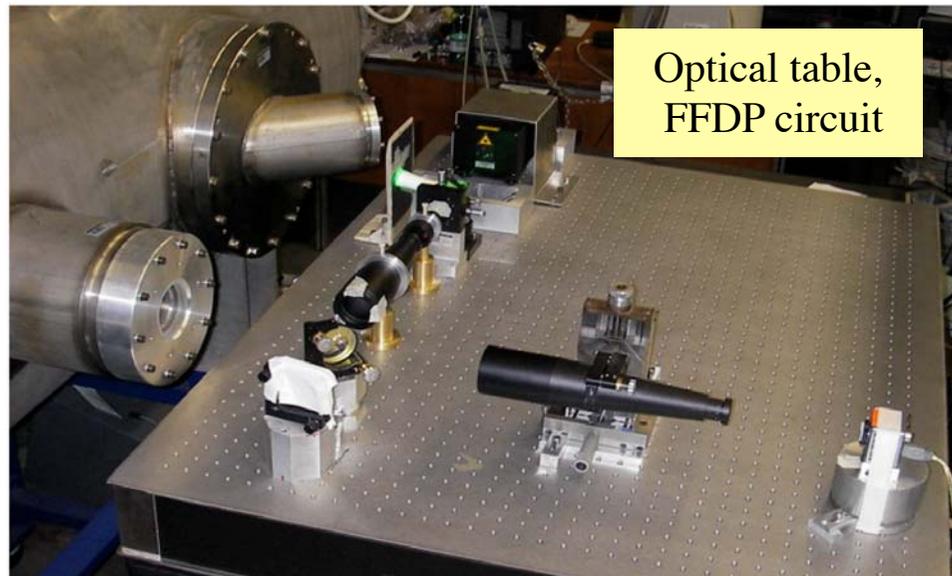
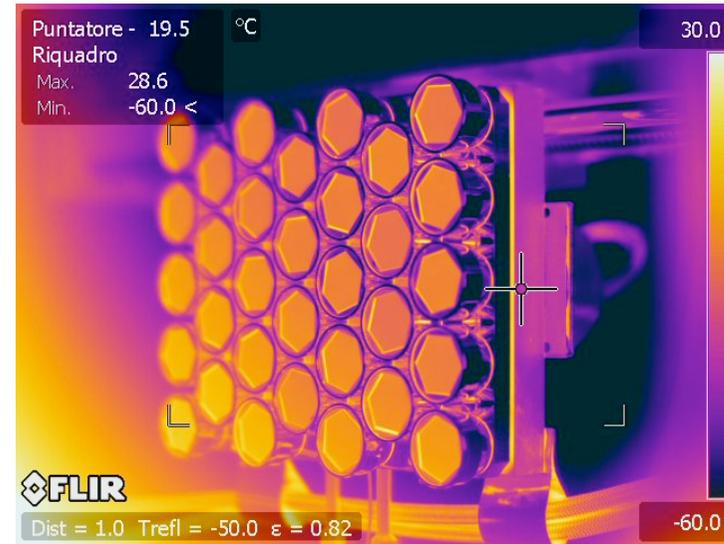
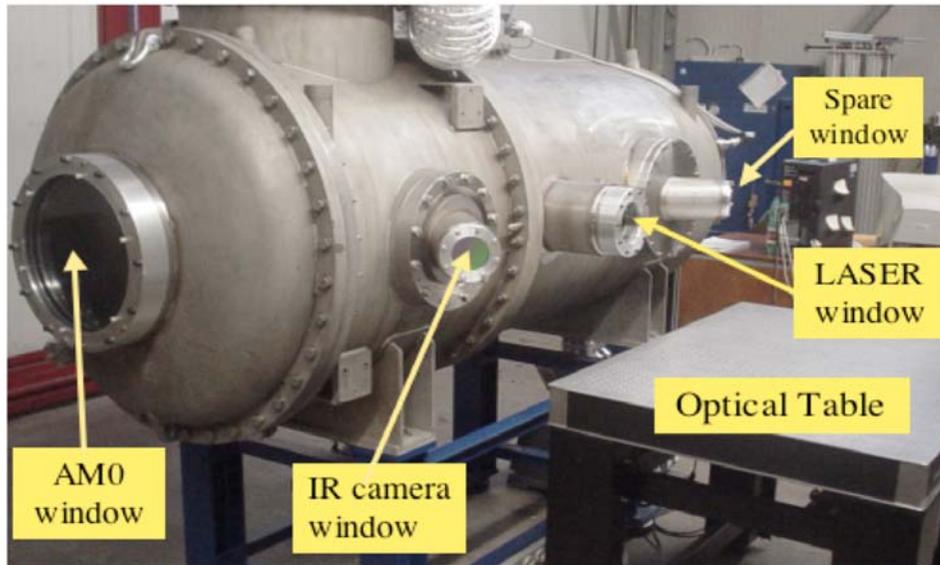
$T(\text{tip}) = 137.4 \text{ K}$



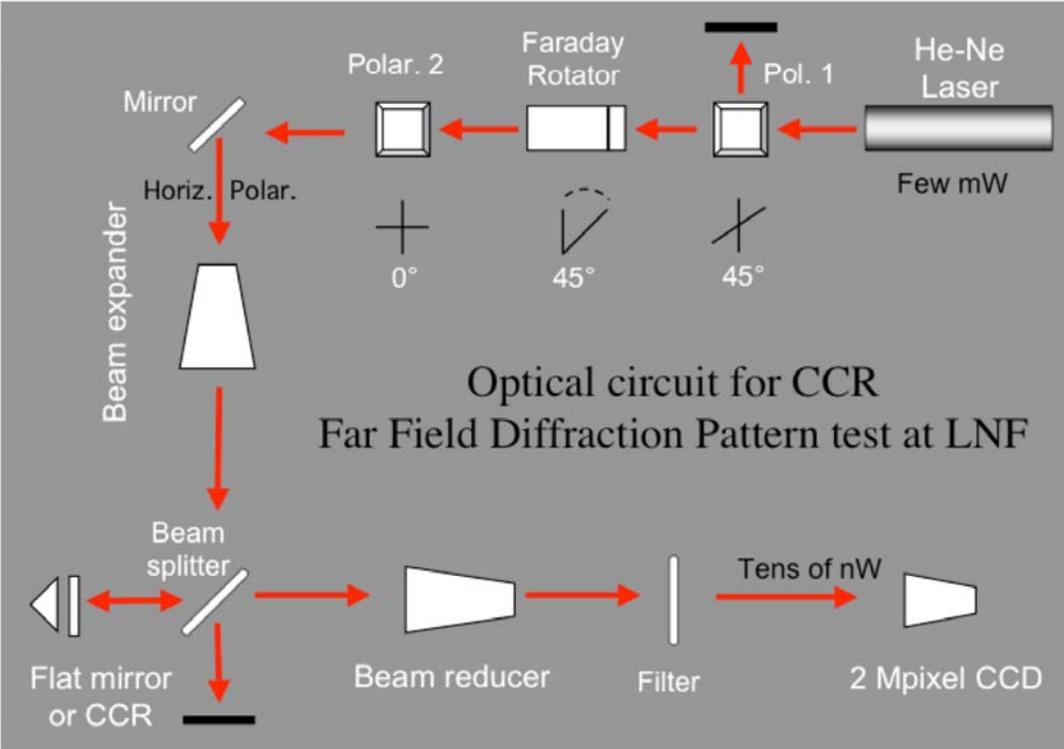
SCF-Test of GPS-2 flight model from USA

The INFN-LNF Satellite Laser Ranging Characterization Facility

Thermal and laser tests **never performed before in space conditions**



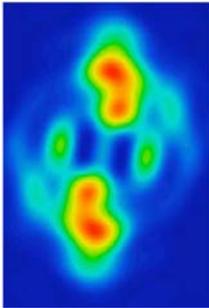
Measured & simulated Far Field Diffraction Pattern



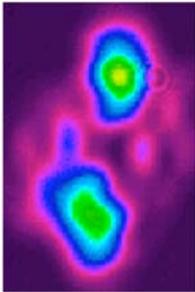
Simulated with CodeV

Measured @ LNF

GPS-2 proto
has dihedral angle
offset = $2 \div 3$ arcsec



Good agreement
Peak distance is $50 \mu\text{rad}$



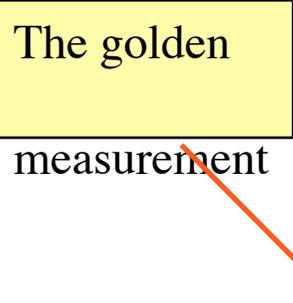
General Relativity Science Objectives

(for up to factor 100 improvement over current LLR)

Table by T. Murphy (U. of California at San Diego), who operates the APOLLO LLR station at Los Alamos

Phenomenon	Current limit	Limit with 1 mm ranging	Limit with 0.1 mm ranging	Measurement timescale
Weak Equivalence Principle ($\Delta a/a$)	10^{-13}	$\sim 10^{-14}$	$\sim 10^{-15}$	2 yr
Strong EP (Nordvedt param.)	4×10^{-4}	$\sim 10^{-5}$	$\sim 10^{-6}$	2 yr
Gdot/G	$10^{-12}/\text{yr}$	$\sim 10^{-13}/\text{yr}$	$\sim 10^{-14}/\text{yr}$	4 yr
Geodetic Precession (PPN parameter β)	$\sim 5 \times 10^{-3}$	5×10^{-4}	$\sim 5 \times 10^{-5}$	6-10 yr
Deviations from $1/r^2$ (Yukawa param. α)	$10^{-10} \times \text{gravity}$	$\sim 10^{-11}$	$\sim 10^{-12}$	6-10 yr

The golden
measurement



Limits on non-newtonian gravity



Current limits on additional Yukawa potential:

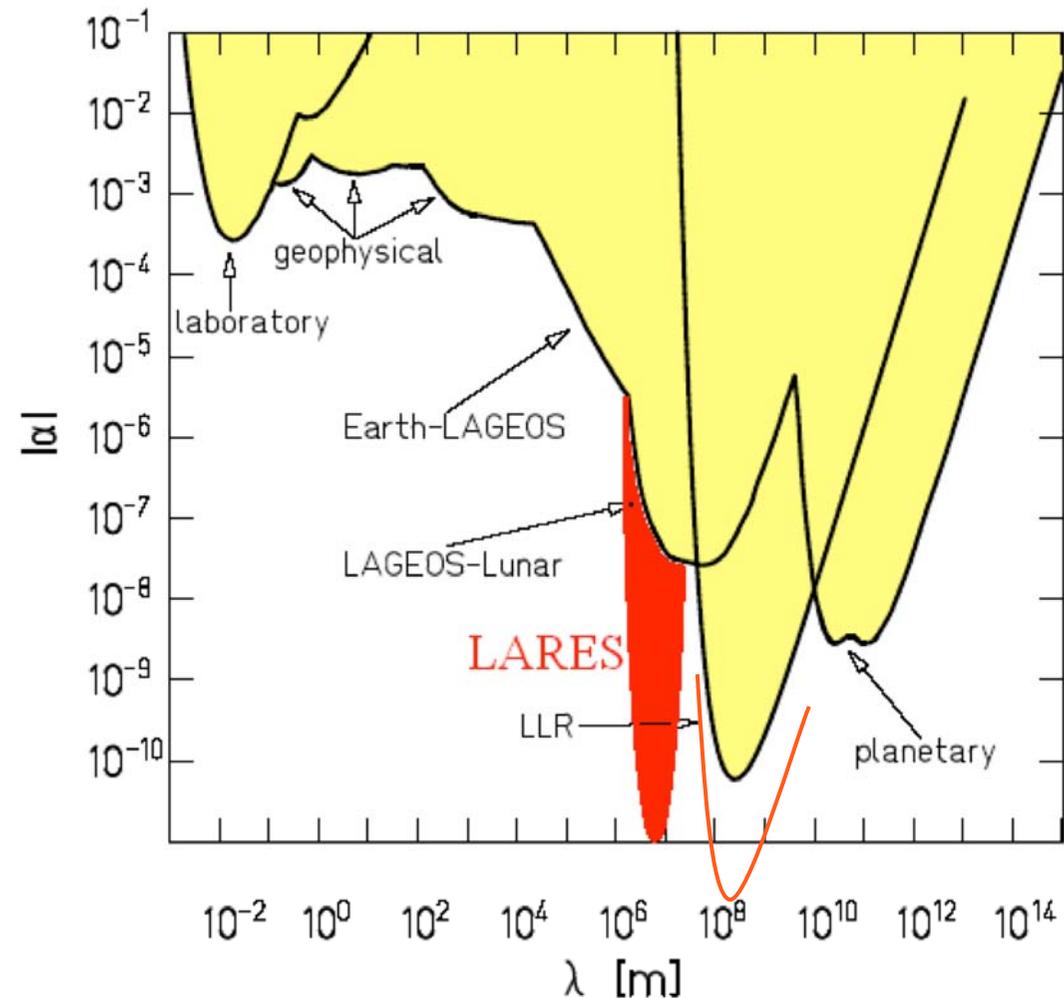
$$\alpha \times (\text{Newt-gravity}) \times e^{-r/\lambda}$$

Expected limit on α for the MoonLIGHT proposal for a ranging accuracy of 100 μm

Expected limit on α set by the LARES mission at an orbit radius of about 8000 Km

LARES limit by I. Ciufolini

MoonLIGHT limit: current LLR limit $\times 10^{-2}$



“BraNe new world”: a model beyond General Relativity

PHYSICAL REVIEW D **68**, 024012 (2003)

The accelerated universe and the Moon

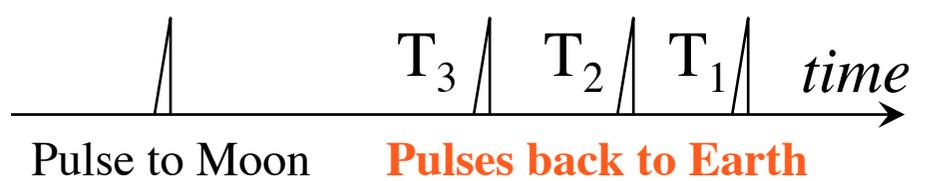
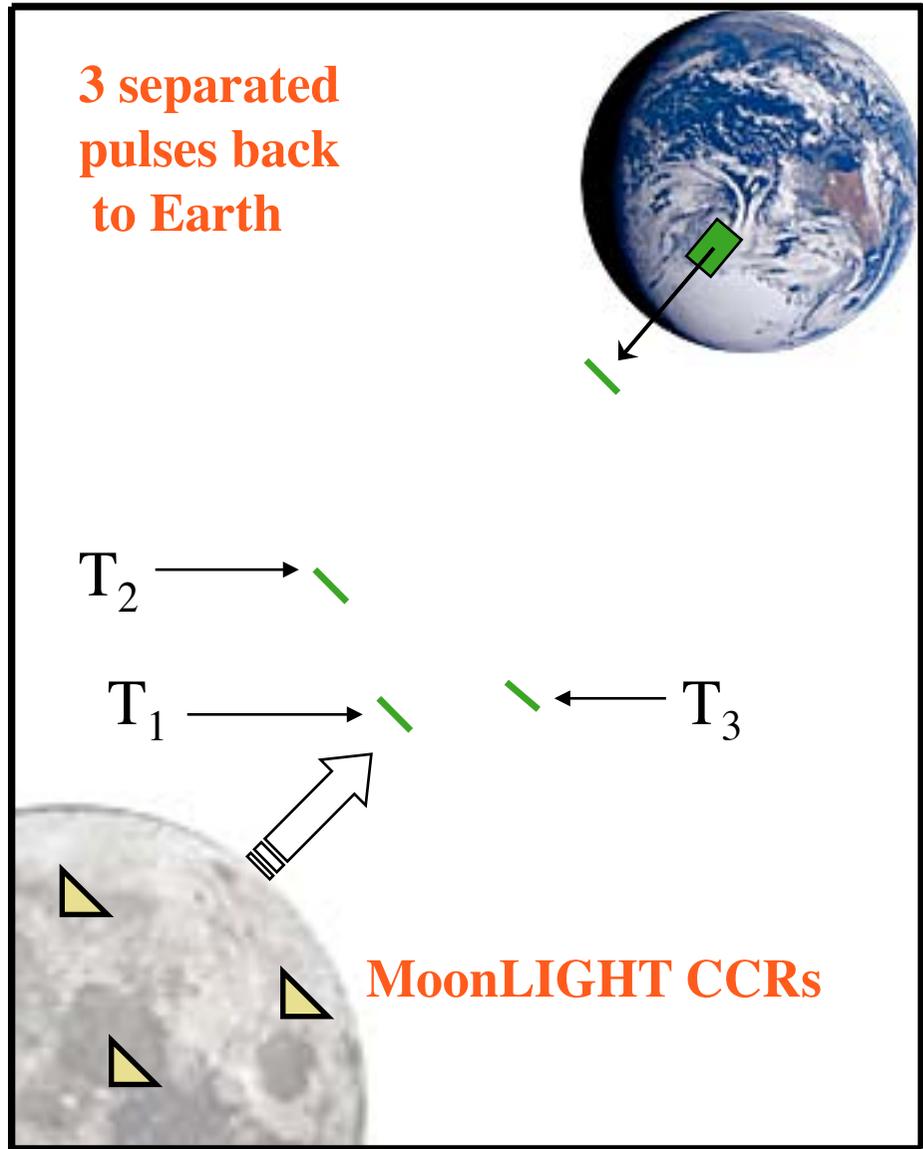
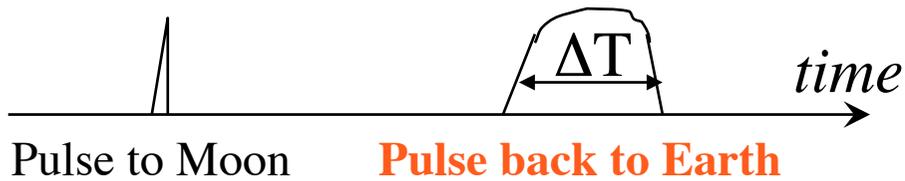
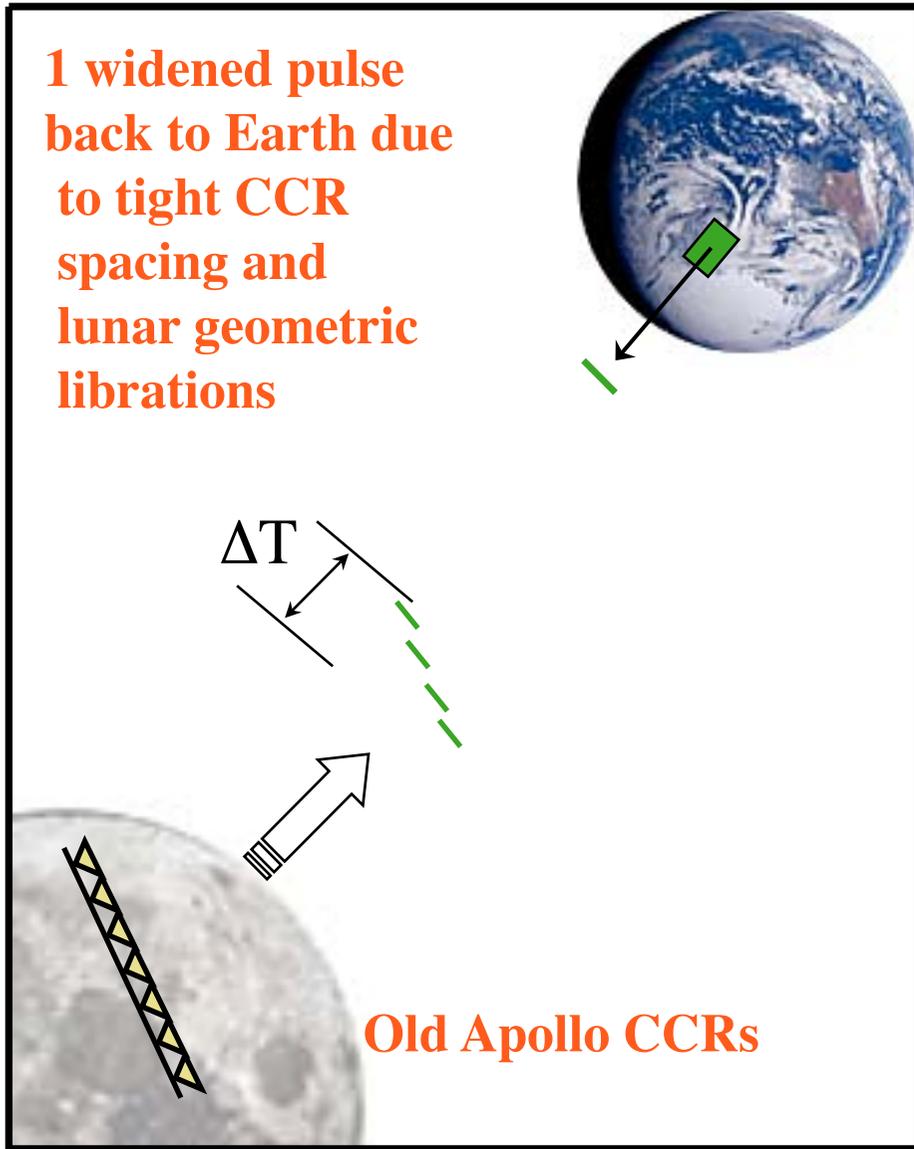
Gia Dvali, Andrei Gruzinov, and Matias Zaldarriaga

for Cosmology and Particle Physics, Department of Physics, New York University, New York, New York 10003

(Received 20 December 2002; published 8 July 2003)

Cosmologically motivated theories that explain the small acceleration rate of the Universe via the modification of gravity at very large, horizon, or superhorizon distances, can be tested by precision gravitational measurements at much shorter scales, such as the Earth-Moon distance. Contrary to the naive expectation the predicted corrections to the Einsteinian metric near gravitating sources are so significant that they might fall within the sensitivity of the proposed Lunar Ranging experiments. The key reason for such corrections is the van Dam–Veltman–Zakharov discontinuity present in linearized versions of all such theories, and its subsequent absence at the nonlinear level in the manner of Vainshtein.

- This (mem)Brane world theory gives anomalous precession of the Moon of \sim **1 mm/orbit**, in addition to standard GR geodetic precession
- LLR accuracy now \sim **1 cm**. New laser station APOLLO is reaching **few mm**
- This model can be fully tested by MoonLIGHT with **100 μ m** (or less) accuracy, i.e. w/factor 100 (or more) improvement over current LLR



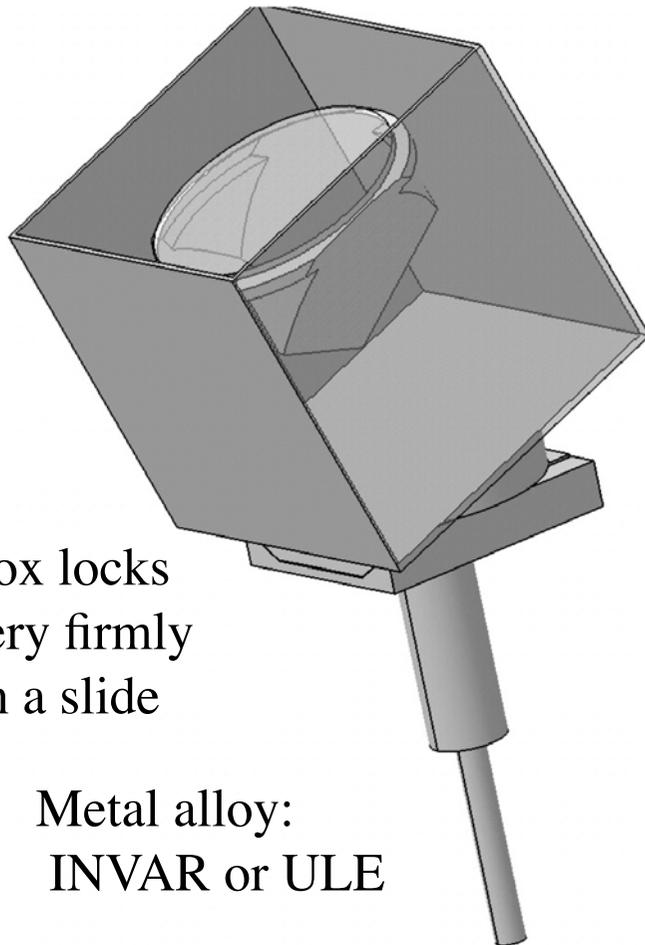
NASA call: "Suitcase" science to the Moon



Concept by Astronaut Roberto Vittori. Manned missions on 2015-18

Retro-reflector: 10 cm diam.

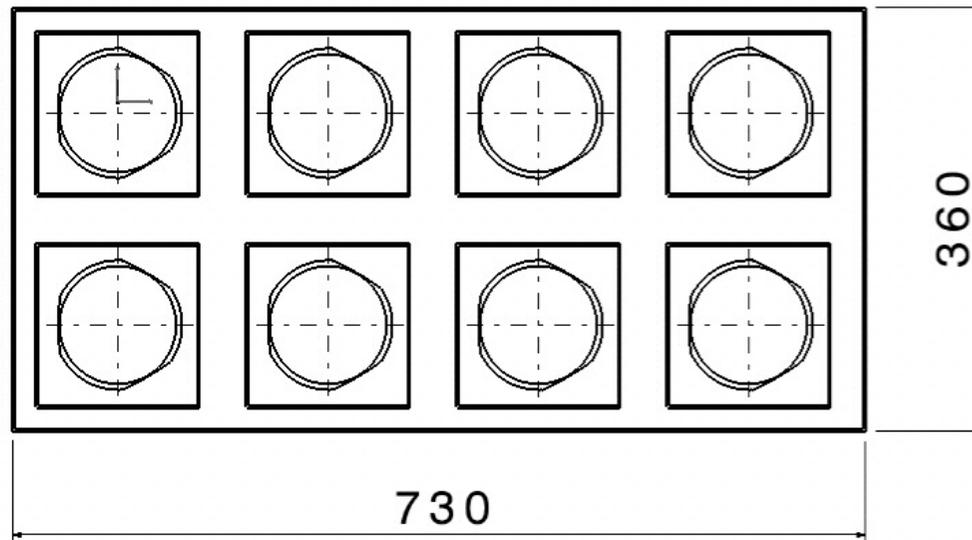
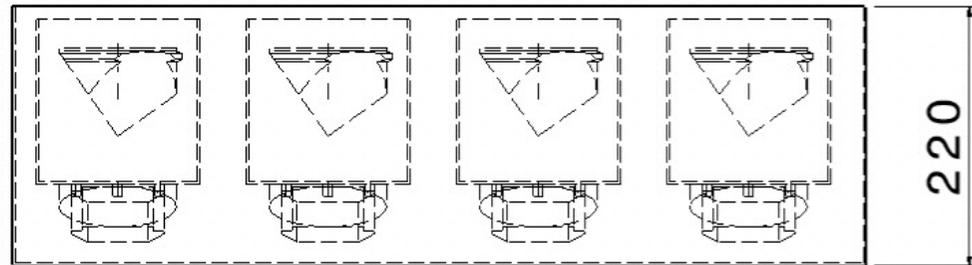
Box: 14 cm side



Box locks
very firmly
on a slide

Metal alloy:
INVAR or ULE

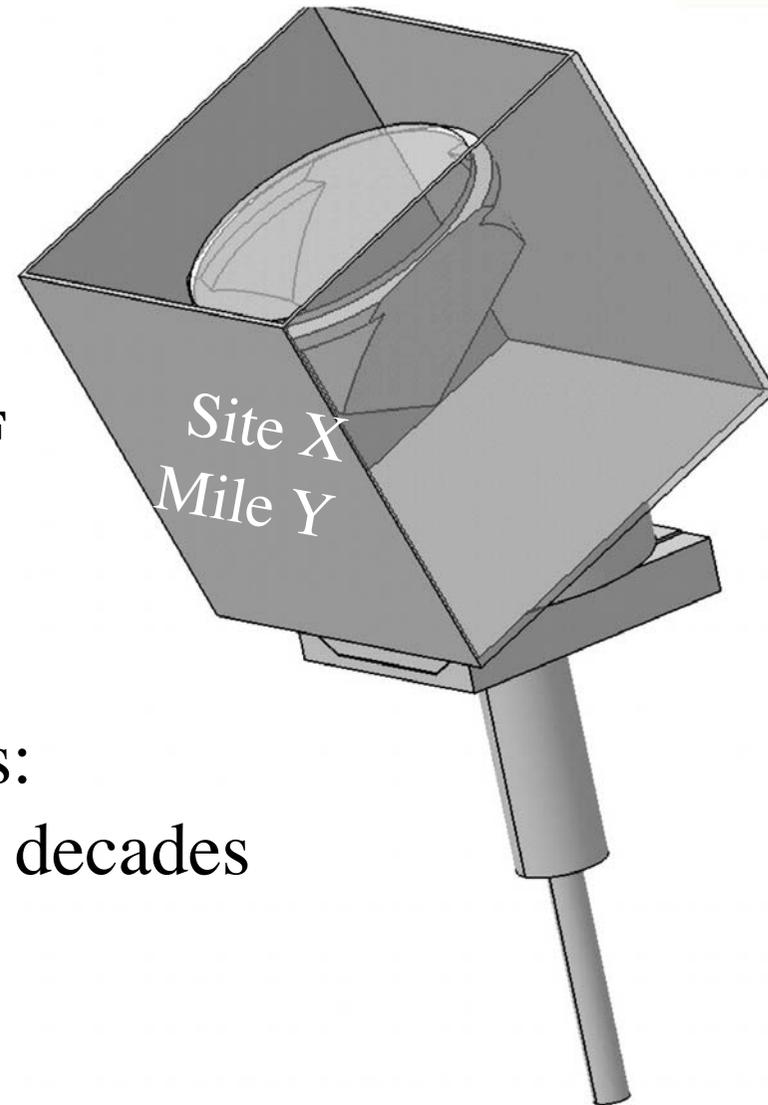
Suitcase for the CCR boxes (mm)



Moon land-marks



- MoonLIGHT units can also be used to **land-mark sites and pathways of the new lunar exploration**, like the Roman “milestones”
- Can be used to build the IMRF
 - Intern. Moon Reference Frame



MoonLIGHT landmarks:
light, passive and last for many decades

Conclusions

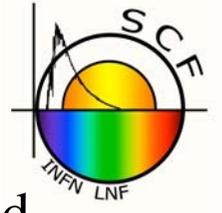


- SLR: precise and cost-effective way to probe gravity with the Earth-Moon system
(Millimeter) .AND. (0.1 M€ to M€)
- Not just physics
 - Space geodesy
 - GALILEO has SLR on all satellites
- Three main contributions to science error budget
 - SLR measurements, Earth grav. models, thermal perturbations
 - **The INFN SCF fills the void of non-existent of space characterization of SLR**
- With SLR/LLR we love to test:
 - General Relativity
 - Yukawa deviations from $1/r^2$, Brane Worlds

THE LNF SCF GROUP



Preliminary thermal analysis: CCR



Thermal CCR conditions look good for integrity of laser signal: cold CCRs work well and T gradient through CCR body is small (< 2 K: variations of the refraction index variations is quite acceptable)

