

# AMS

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- Introduction
  - Constraints / Environment
  - AMS-01
- AMS-02 Detector
- *Preparativi* and launch dates

INFN-Spazio  
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# Alpha Magnetic Spectrometer (AMS)

- The AMS is a **large acceptance** ( $\sim 0.5 \text{ m}^2\text{sr}$ ) detector designed to operate on the **International Space Station (ISS)**, **version AMS-02**, for a **long duration (3y)** measurement of the cosmic-ray spectra ( $\sim 500 \text{ MeV}$  to several  $\text{TeV}$ ).
- The spectrometer will provide precise measurements of the **cosmic-ray composition up to  $Z \sim 26$** , including **light isotope separation**, and is intended to provide a unprecedented sensitive **search for antimatter and dark matter**.
- A first version of the device (**AMS-01**) was operated successfully on a **10-day shuttle flight** in June, 1998.

# Constraints

- Vibrations and shock loads during launch
- The space vacuum
- The radiation environment (thermal and ionizing)
- **Payload mass**
- Payload power
- Bandwidth for data transmission

# Space Environment

## Cosmic Ray Composition:

~90 % hydrogen nuclei

~9 % helium nuclei

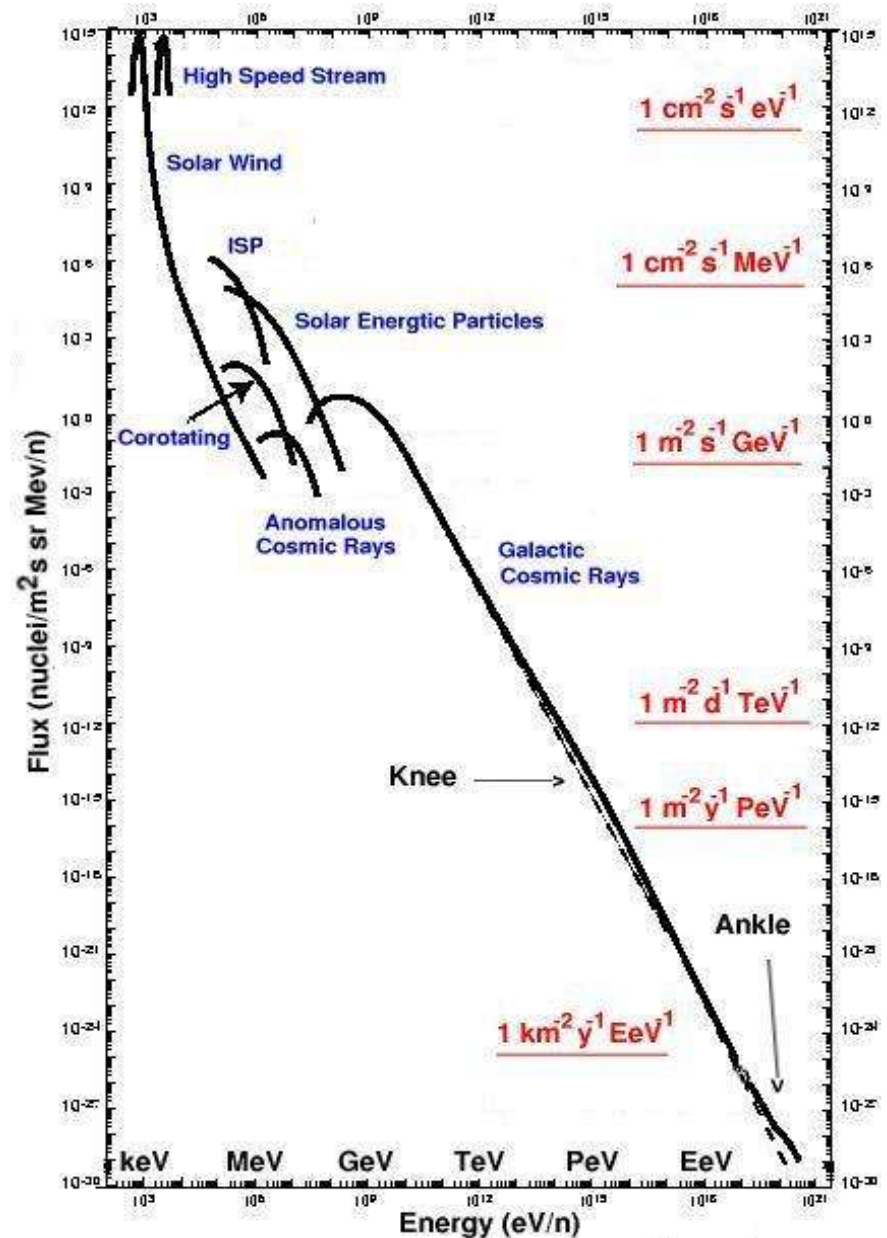
~1 % heavier nuclei

< 1 % electrons

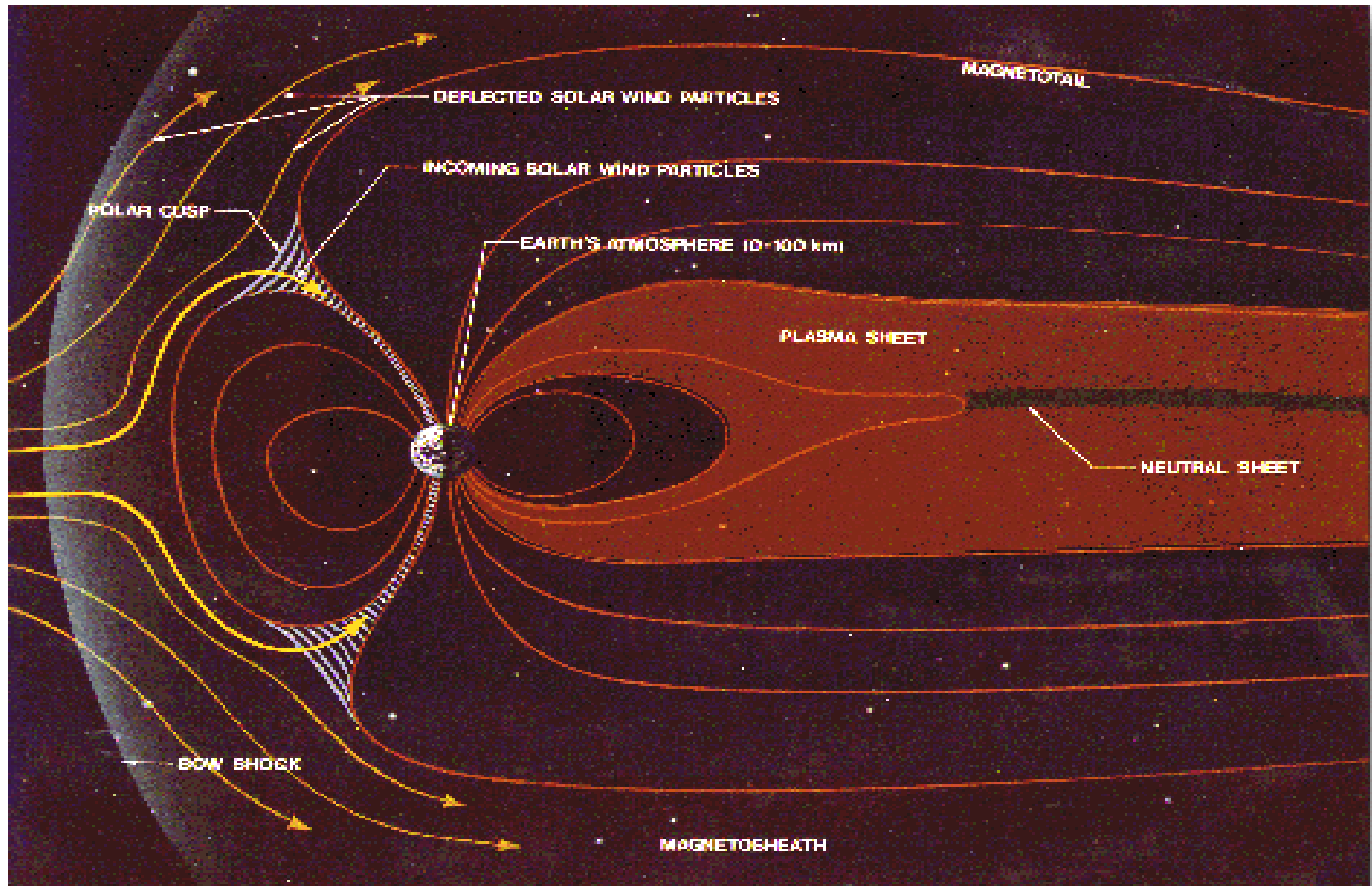
~0.1 % positrons

< 0.1 % gamma rays

<  $10^{-4}$  antiprotons

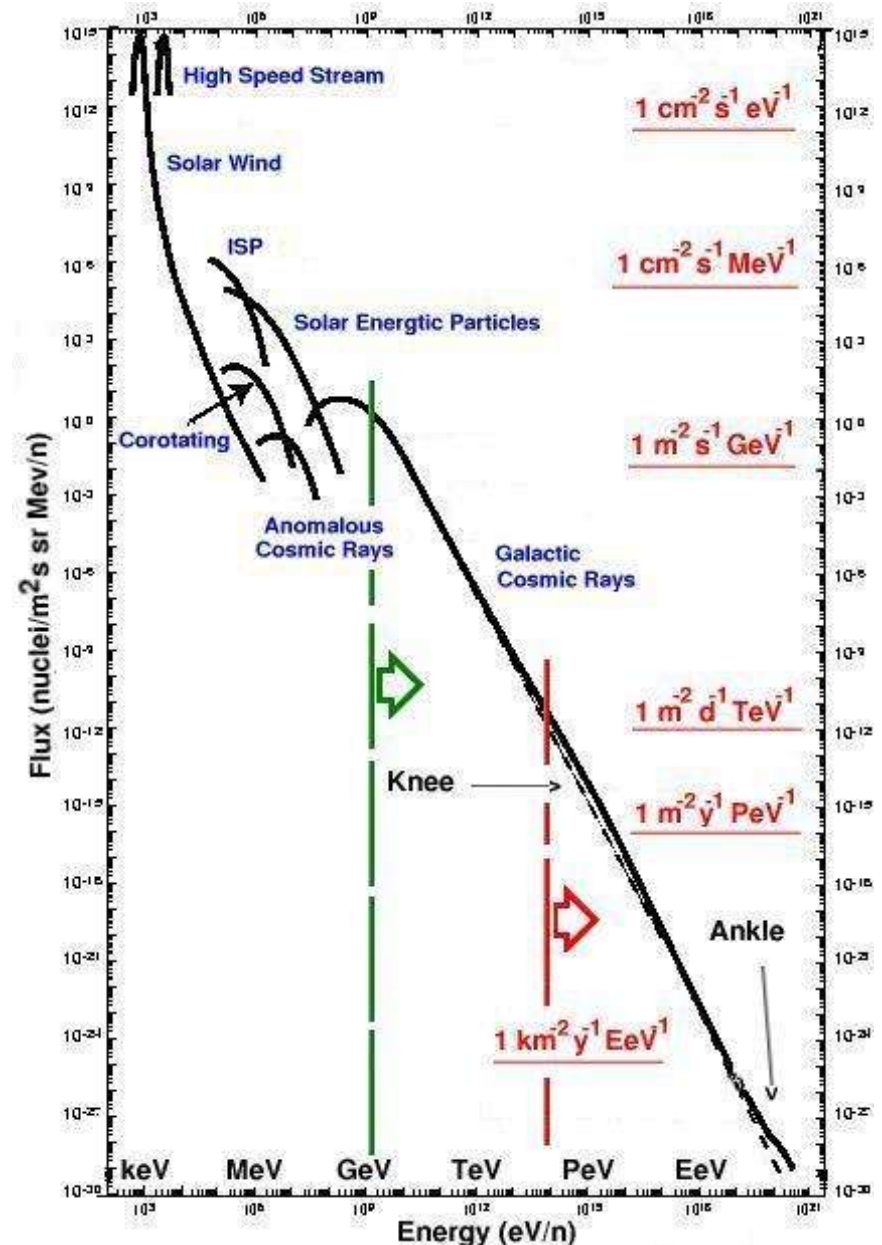


# Space Environment



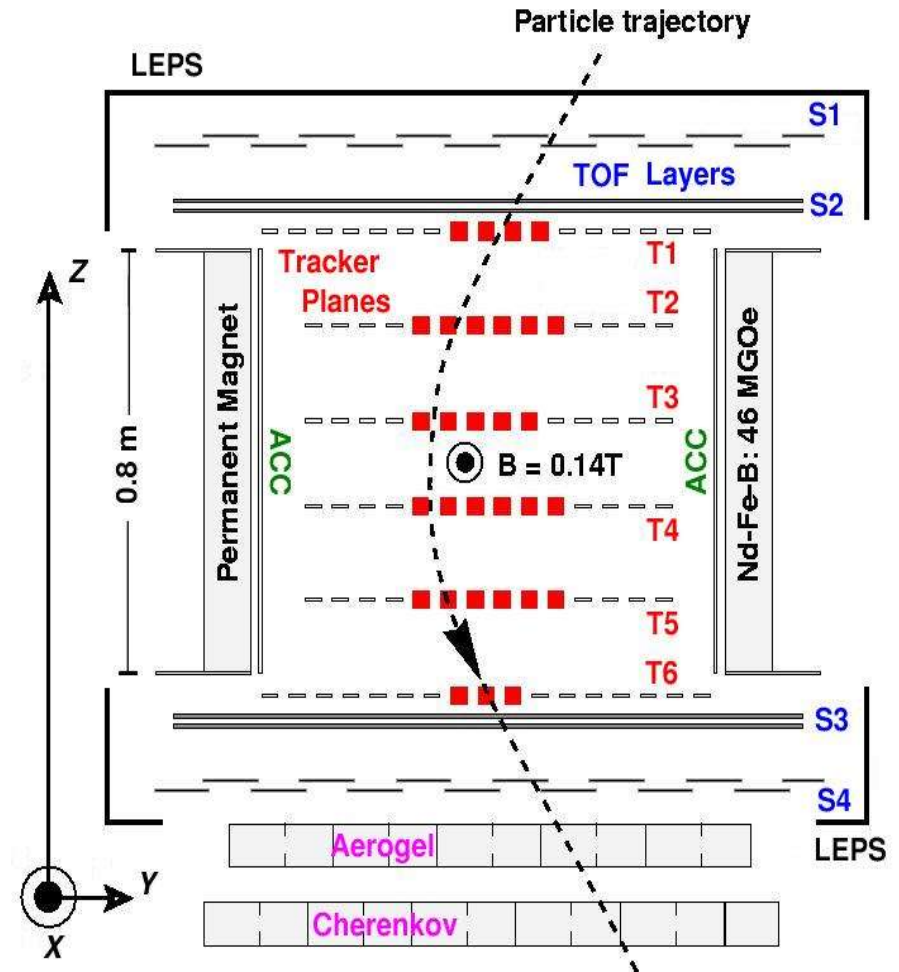
# Cosmic Ray Detection in Space

- At relativistic energies, above  $\sim 1 \text{ GeV}/n$  the detection techniques such as  $dE/dx-E$  or time-of-flight are inoperative or impractical
- A **magnetic spectrometer** provides an alternative, and allows a determination of the sign of the particle charge.
- Balloon-borne magnetic spectrometers are used to study cosmic rays, limited by flight times (currently 10-20 days)
- A fundamental constraint for a space-borne device is the small fluxes at highest energies (above  $\sim 1 \text{ PeV}$ ).



# AMS-01

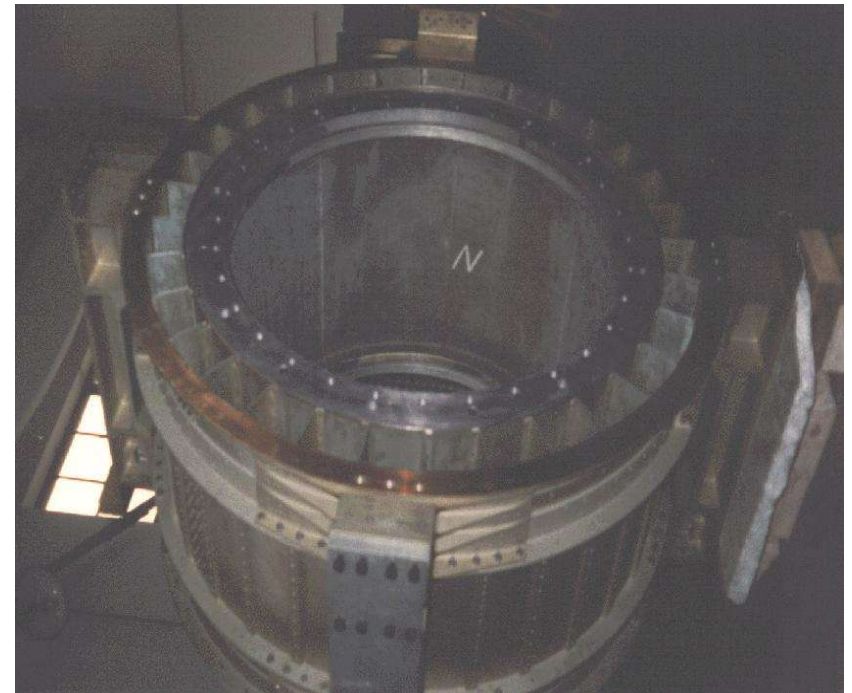
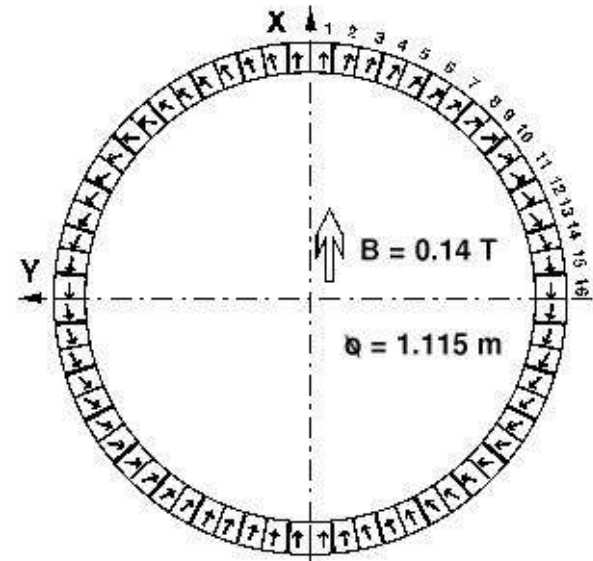
- TOF layers **S1-S4** provided the velocity and  $dE/dx$
- The tracker planes **T1-T6** yielded the rigidity,  $dE/dx$ , and the sign of the charge
- **Cherenkov** counter discriminated between positrons (electrons) and protons (antiprotons) below  $3.5 \text{ GeV}/c$
- The anti-coincidence counter **ACC** was used as a veto
- LEPS, low energy ( $< 5 \text{ MeV}$ ) particle shield





# AMS-01 Permanent Magnet

- Cylindrical form to optimize acceptance/weight
- Composed of 640, 2''x2''x1'' high grade NdFeB blocks
- Dimensions:
  - inner dia. 1.115 m
  - outer dia. 1.298 m
  - length 0.80 m
- Weight: 2.2 t
- 0.14 T field at center
- Negligible dipole field,  
 $B < 3 \text{ G}$  at 2 m from center



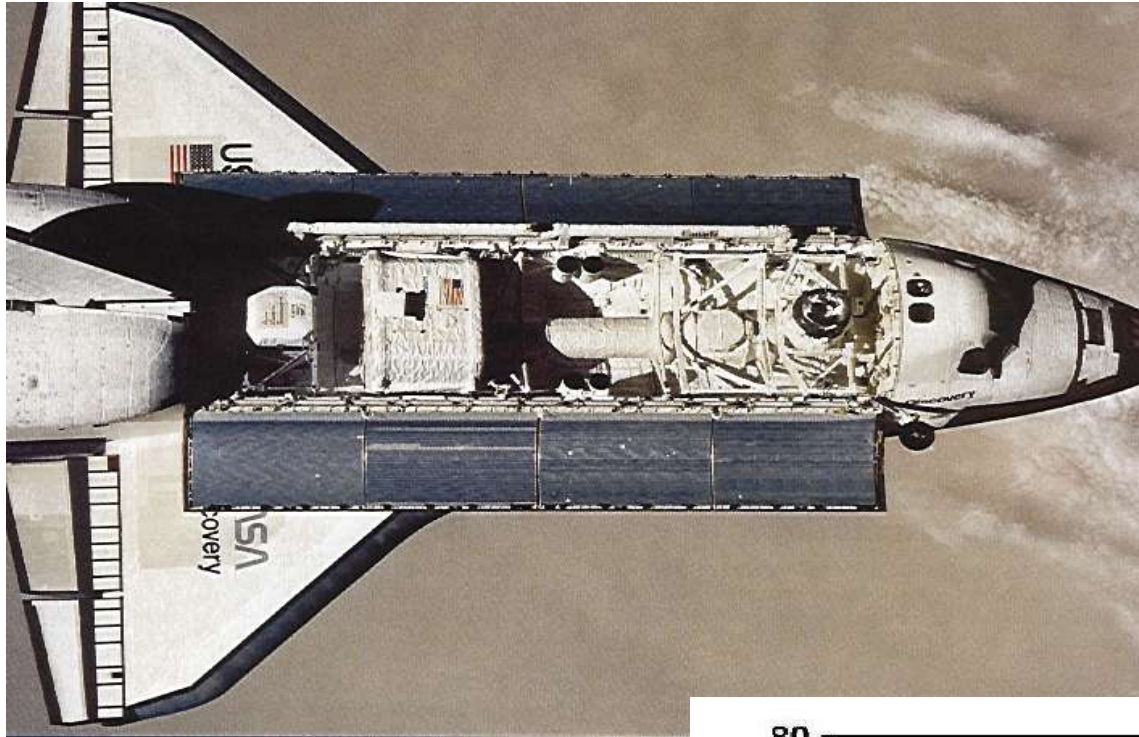


# AMS-01 Silicon Tracker

- Tracker (T1-T6) composed of  $4.136 \times 7.205 \times 0.03 \text{ cm}^3$  double-sided, silicon micron-strip sensors
- implantation/readout strip pitch  
p-side: 27.5/110 microns (y)  
n-side: 26.0/208 microns (x)
- Point measurement resolutions  
 $\sim 10$  and  $\sim 30$  microns
- 57 ladders ( $2.1 \text{ m}^2$  of silicon),  
geometric acceptance  $\sim 0.3 \text{ m}^2\text{sr}$
- 58,368 analog channels read out  
with a low-noise, low-power  
( $0.7 \text{ mW/ch}$ ), high dynamic  
range (75 mips) front-end chip



# AMS-01 Shuttle Flight STS-91

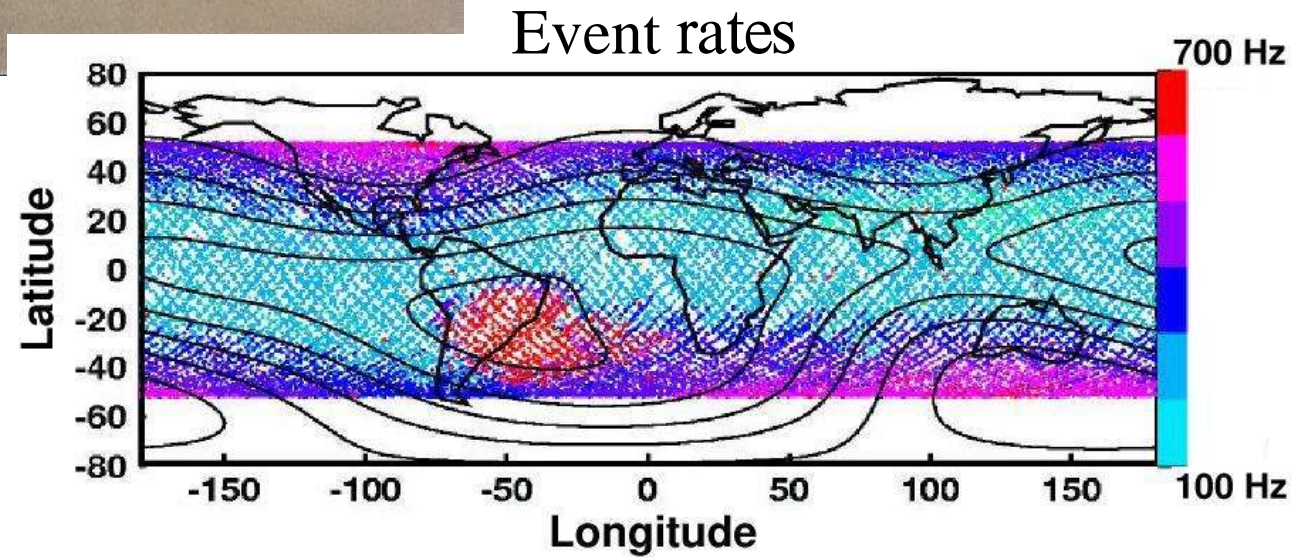


10-day flight in June, 1998

51.7° orbit at altitudes between 320-390 km

Last rendez-vous with Mir

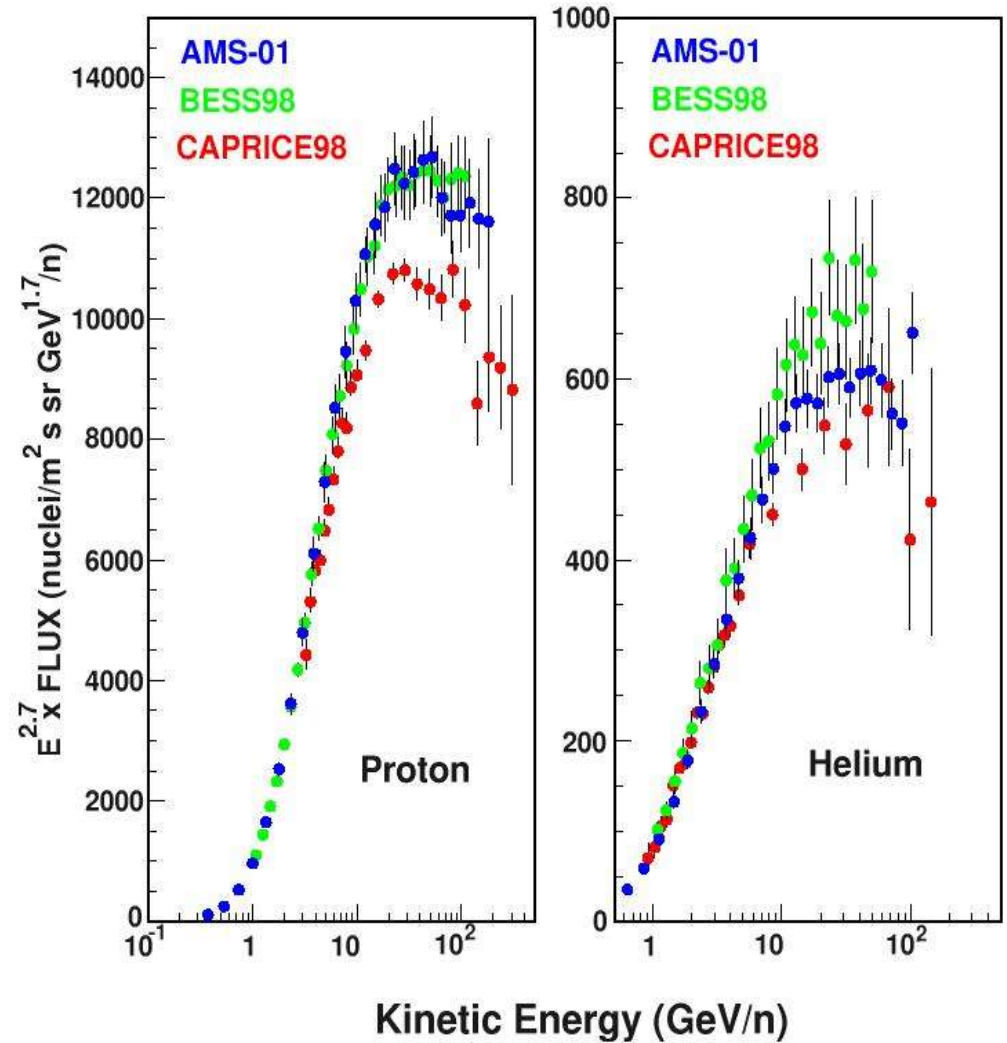
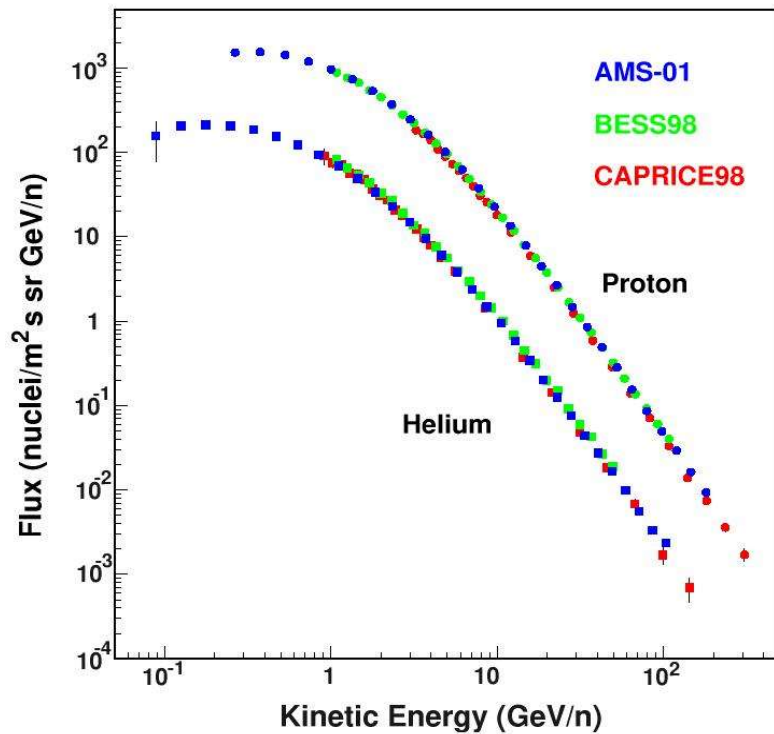
10<sup>8</sup> events recorded





# AMS-01 Results

## *Cosmic-Ray Fluxes (p, He)*

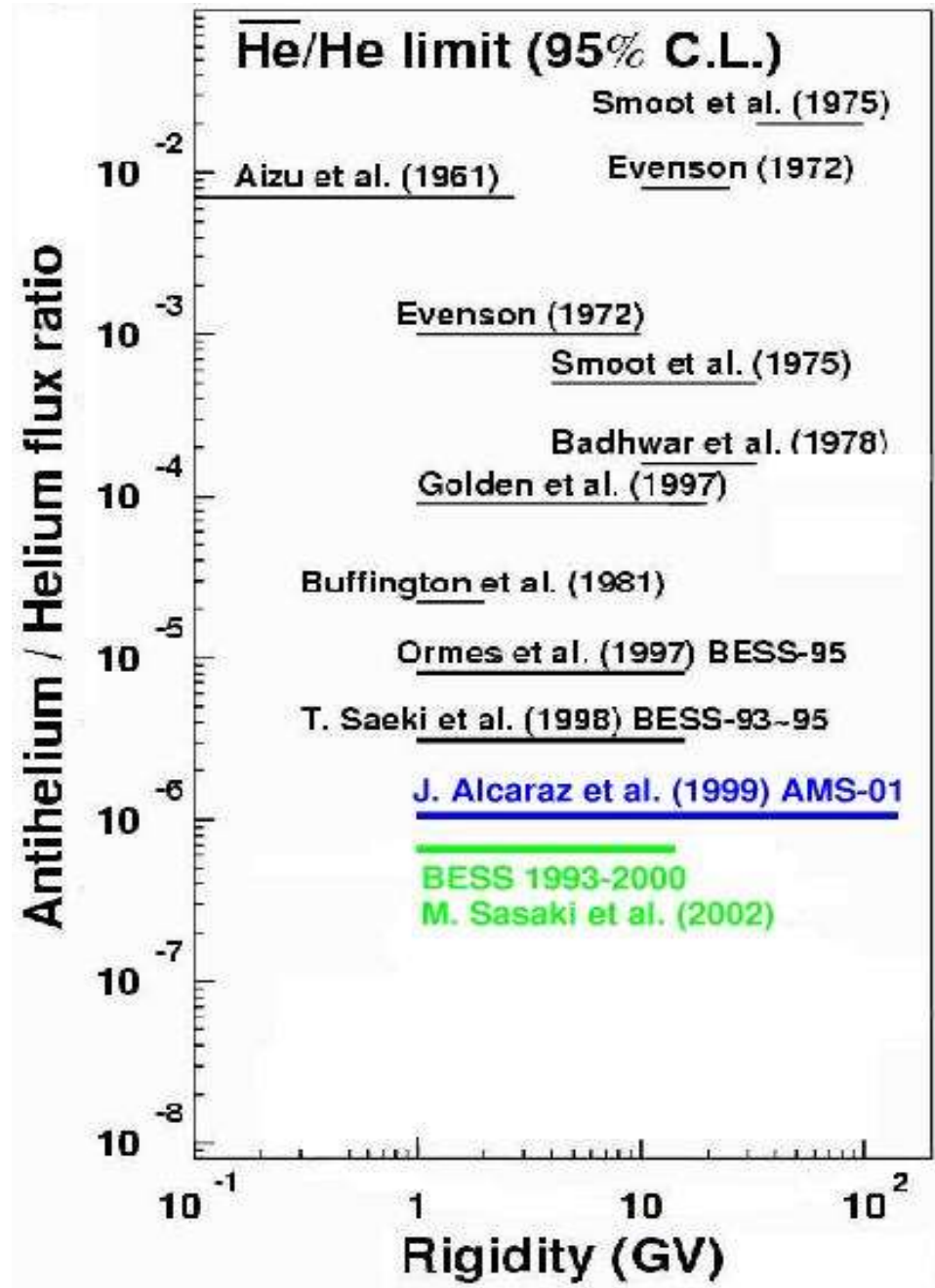


CAPRICE: May 1998  
34.5° N, 110° W (4.3 GV)

BESS: July 1998  
57° N, 101° W (0.5 GV)

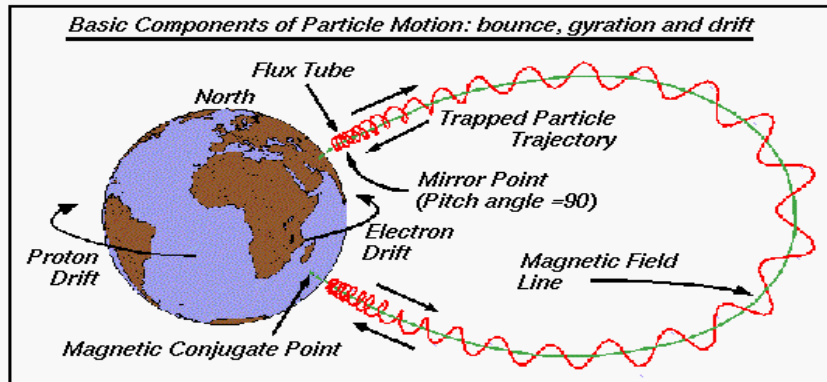
# AMS-01 Results

## *Antihelium Search*



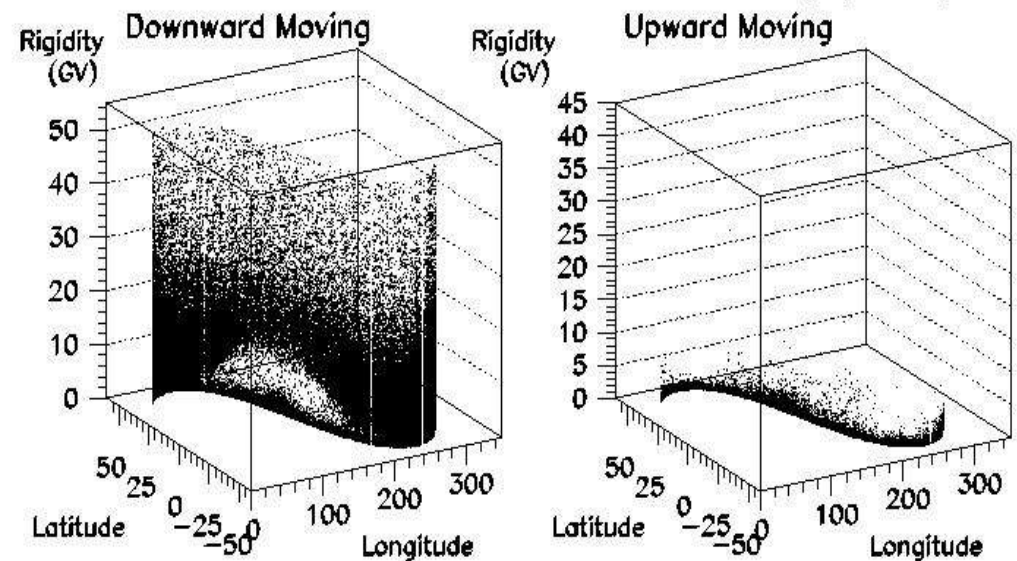
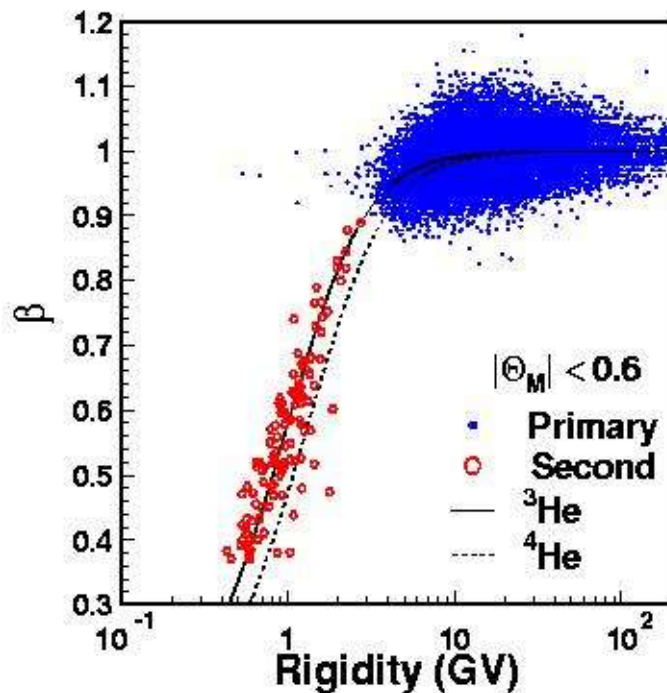
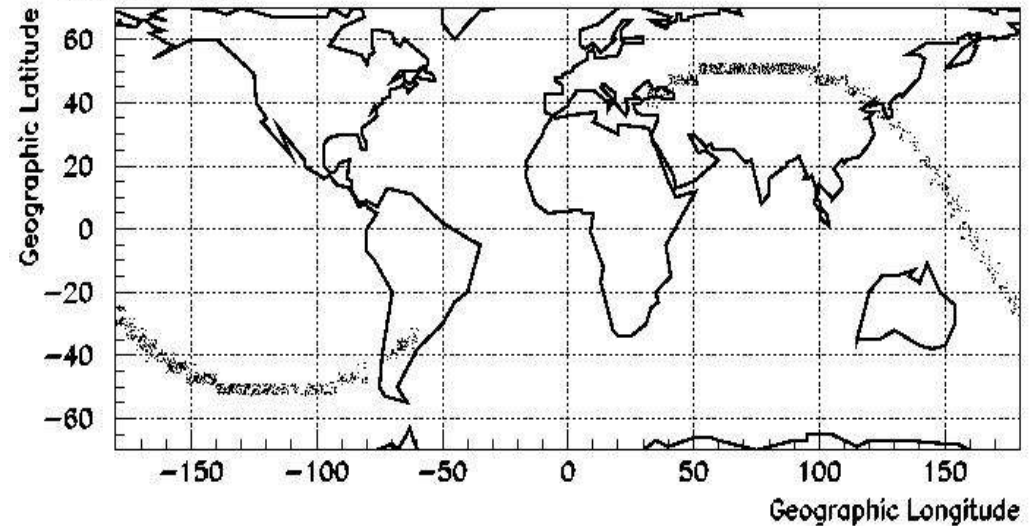
# AMS-01 Results

## *Undercutoff Particle Populations ( $p, e^-, e^+, {}^3\text{He}$ )*



STS-91  
MET : Day 6.4  
Duration : 70 min

Detected Particle Location and Rigidity





# AMS-02

w.r.t AMS-01

Modified:

magnet

tracker

TOF

ACC

Replaced:

Cherenkov  $\Rightarrow$  RICH

Added:

TRD

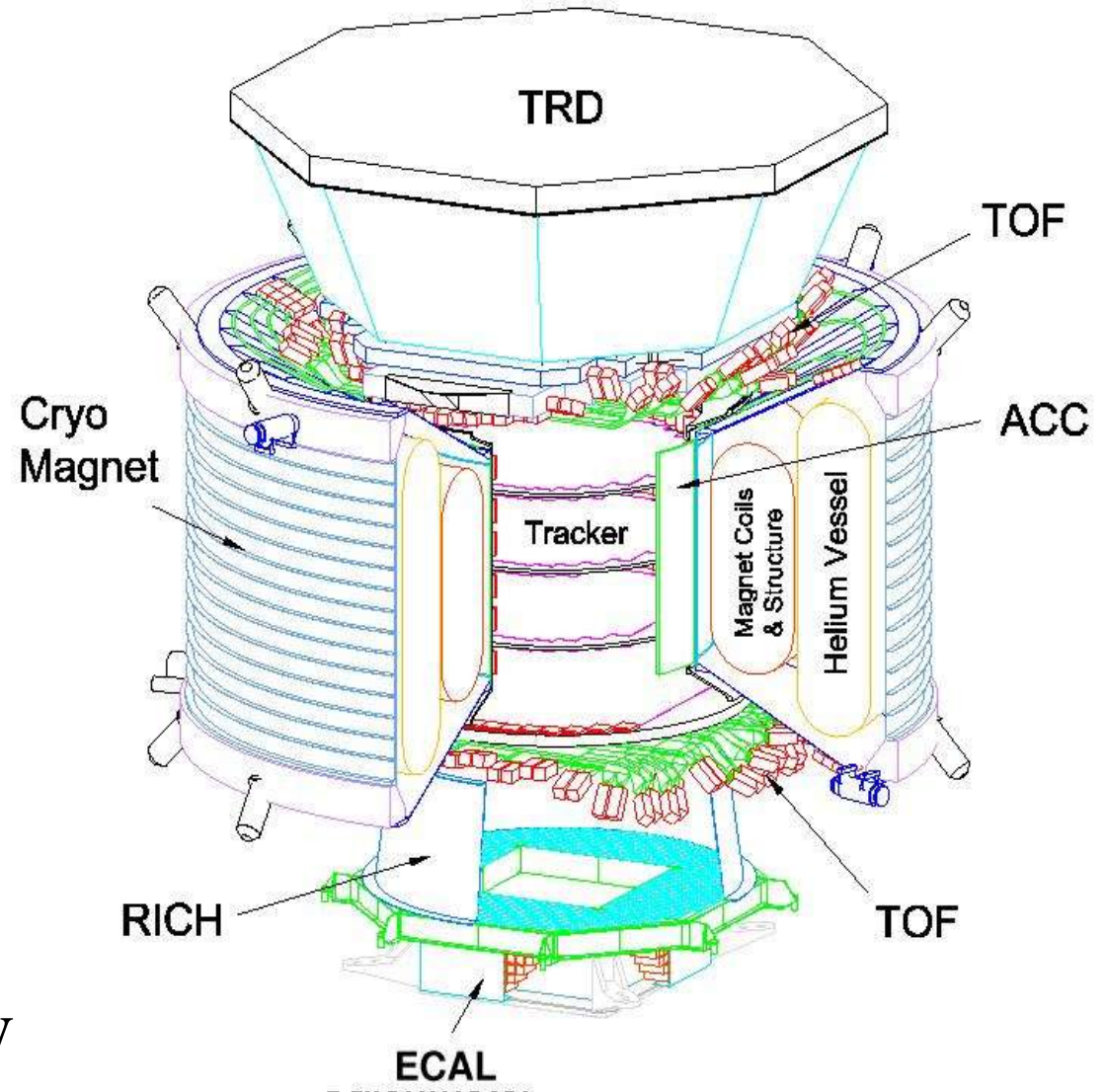
ECAL

Resources:

mass 3.2  $\rightarrow$  7.0 t

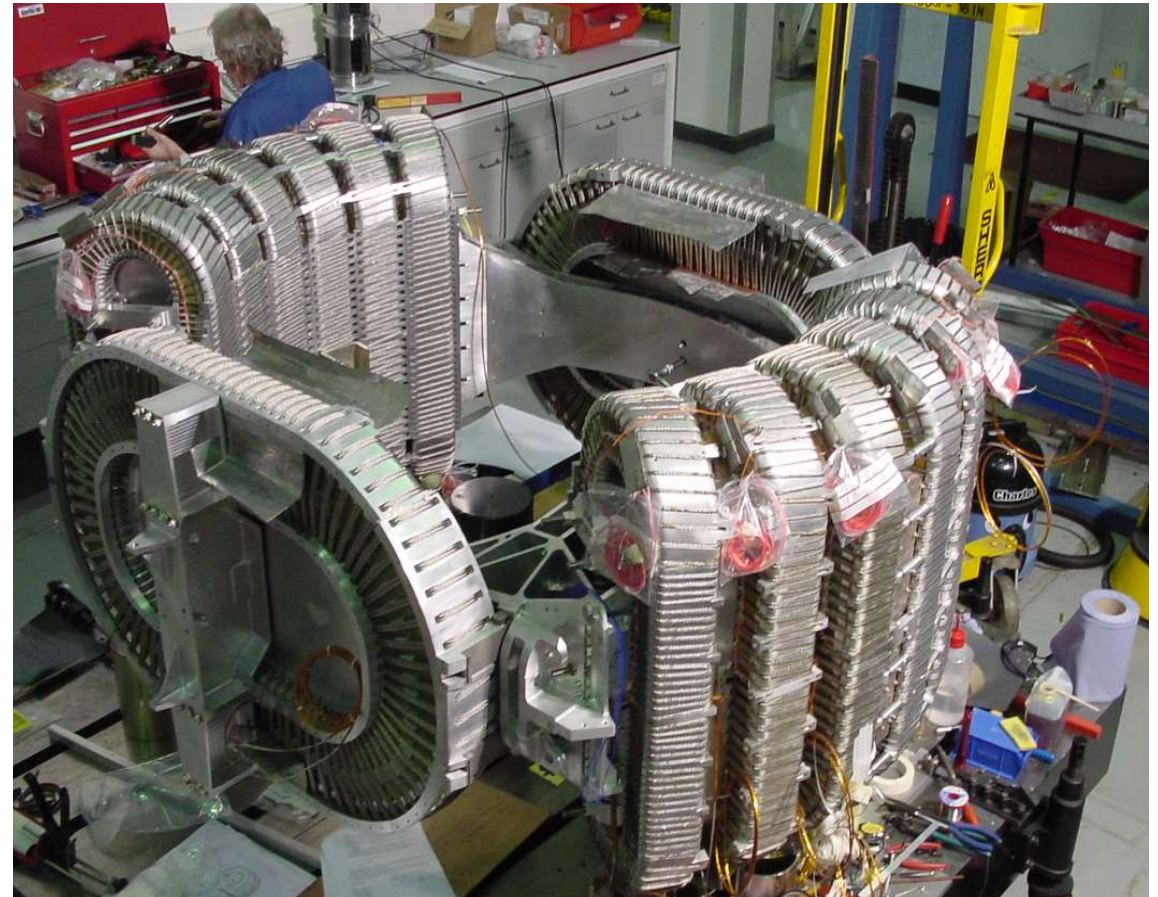
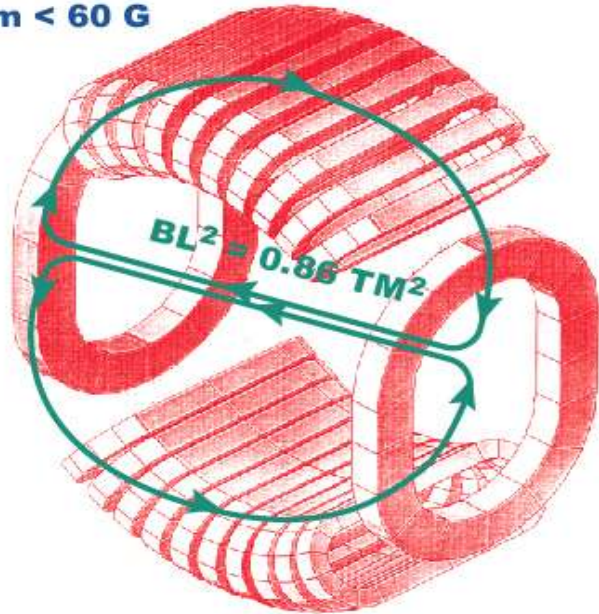
power 1.0  $\rightarrow$  2.0 kW

2 Mbit/s downlink



# AMS-02 Superconducting Magnet

$n < 60 \text{ G}$



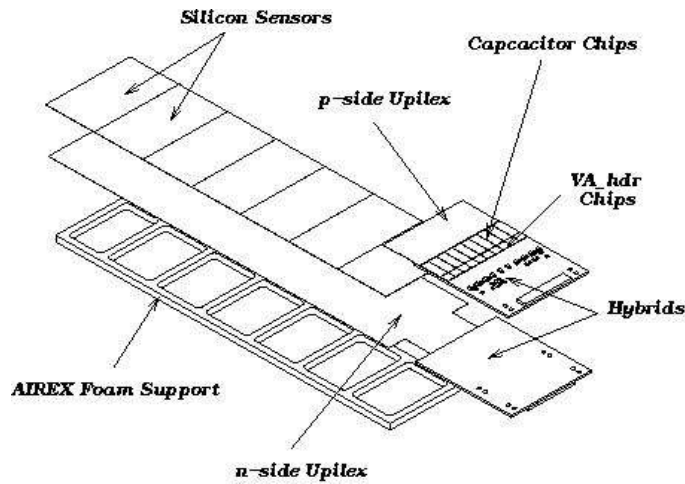
## AMS-02 Coil Characteristics

Parameter	Helmholtz	Racetrack
Height [mm]	1081	826
Width [mm]	681	306
no. of turns	3360	1457
cross section [mm]	88 × 146	54 × 103

$$BL^2 = 0.8 \text{ Tm}^2$$

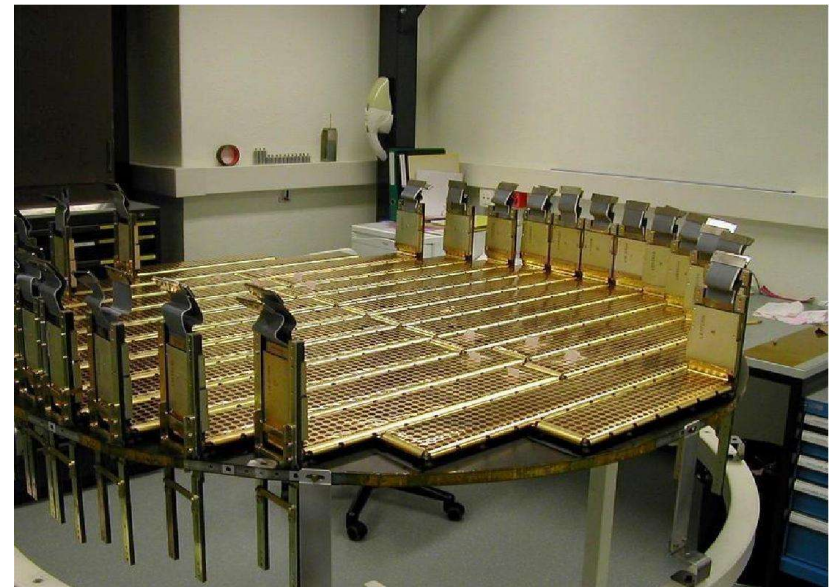


# AMS-02 Silicon Tracker



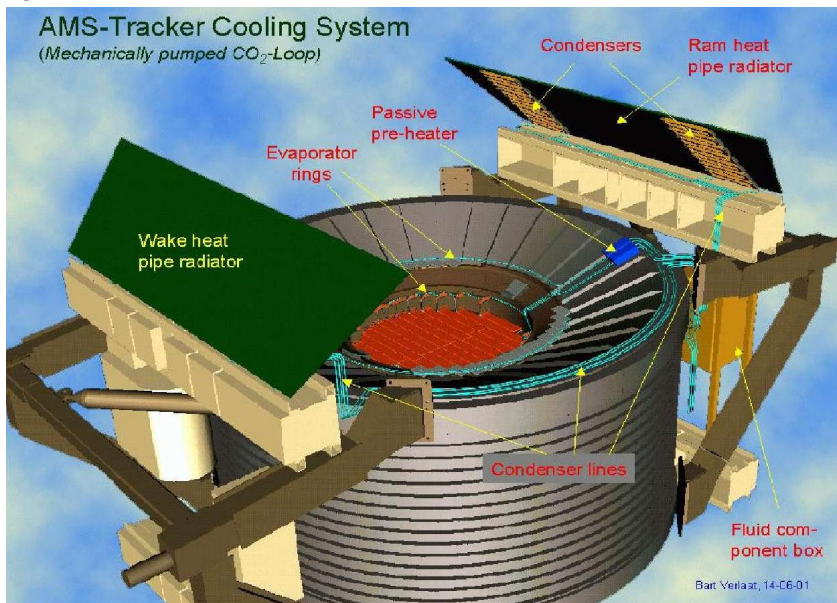
Ladder: implantation strip pitch n-side 26 → 104 microns

Plane: 6 → 5, layers of silicon 6 → 8

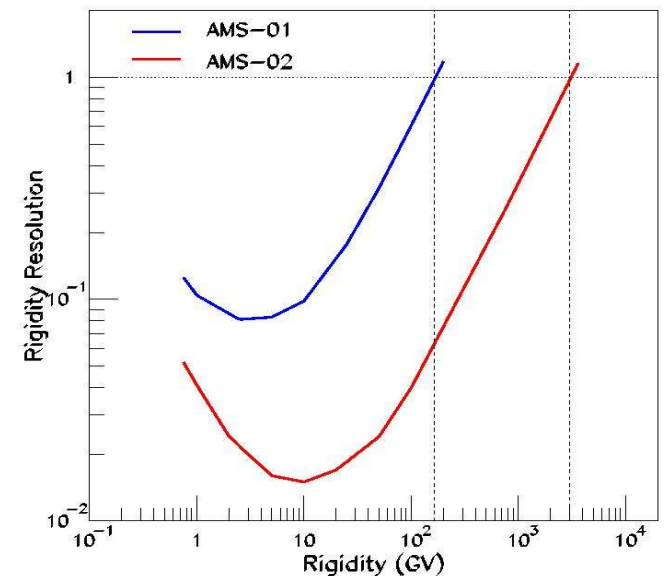


## Tracker Thermal Control System (TTCS)

Liquid CO<sub>2</sub> used to remove heat generated by the front-end electronics in the interior of the magnet.



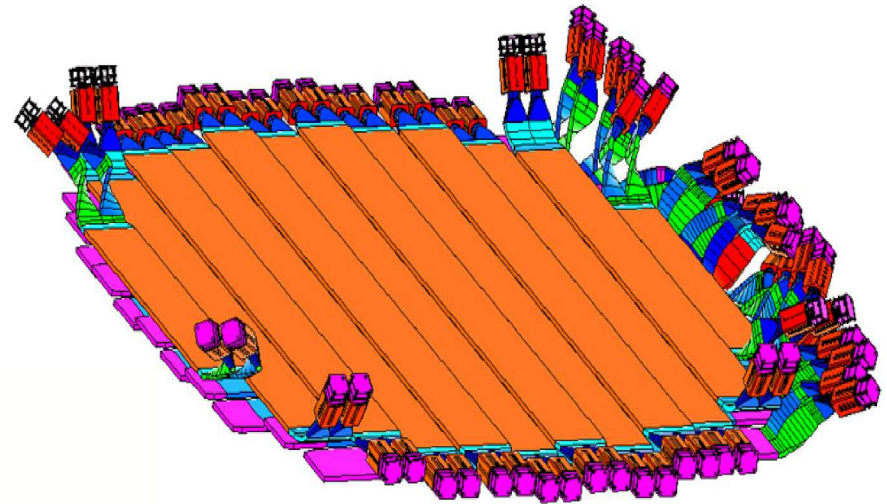
MDR ~2 TV



# AMS-02 Subsystems

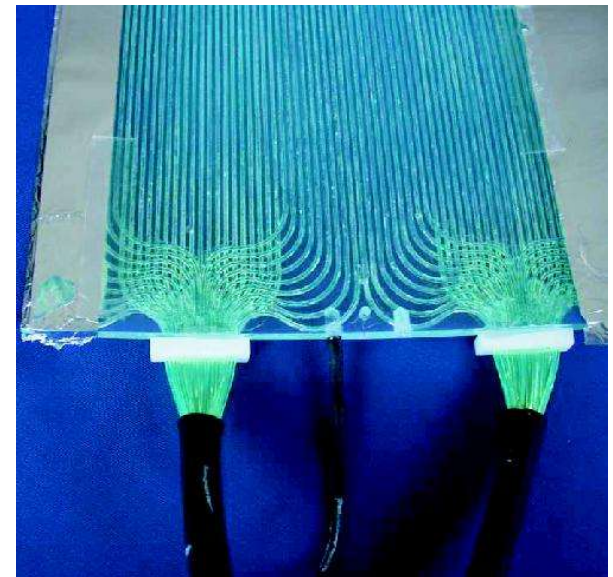
- TOF:

- Curved light guides and fine mesh PMs.
- Time resolution  $\sim 140$  ps.
- Segmentation reduced to 8, 8, 10, and 8 paddles per plan.



- ACC:

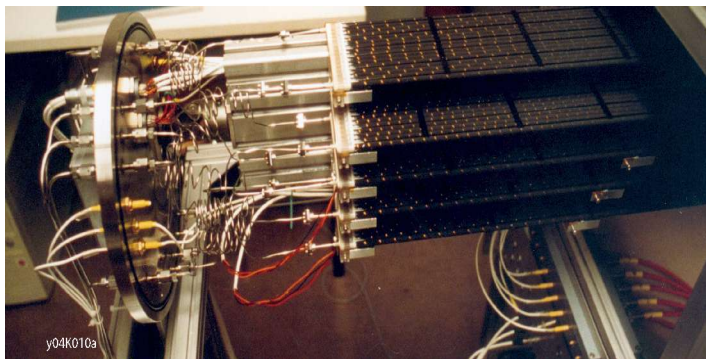
- Sixteen  $22 \times 83$  cm<sup>3</sup> scintillator paddles, thickness reduced from 10 to 8 mm.
- The number of PMs is reduced from 32 to 16
- Light collection with scintillating fibers





# AMS-02 Subsystems

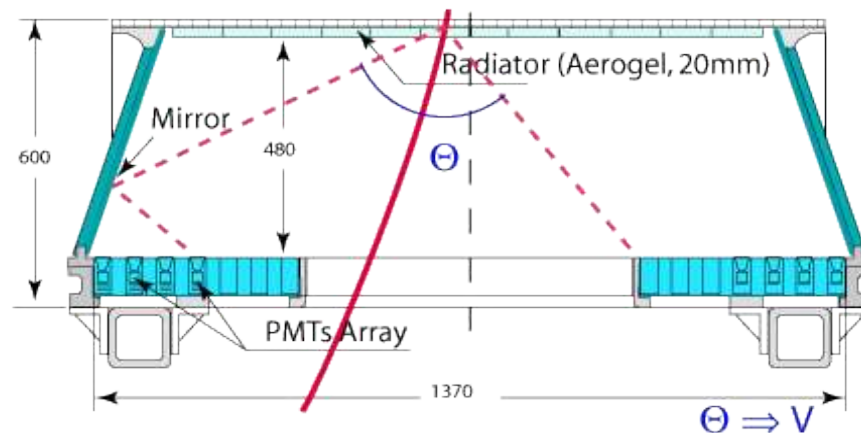
- TRD:
  - 20 layers of 21 mm thick polyethylene/polypropylene fleece radiator alternating with 6 mm dia. gas-filled straw tubes ( 80% Xe, 20% CO<sub>2</sub> @ 1 bar; 50 kg of gas embarked for 3 y operation (p rejection 10<sup>2</sup>-10<sup>3</sup> in range 20 to 250 GeV with 90% e<sup>±</sup> detection efficiency)
  - Straw tubes grouped in 16-tube modules, 328 modules are arranged in a conically-shaped octagonal structure.
  - Layers are oriented orthogonally to provide 3-dimensional track reconstruction.



# AMS-02 Subsystems

- Ring-Imaging Cherenkov (RICH):

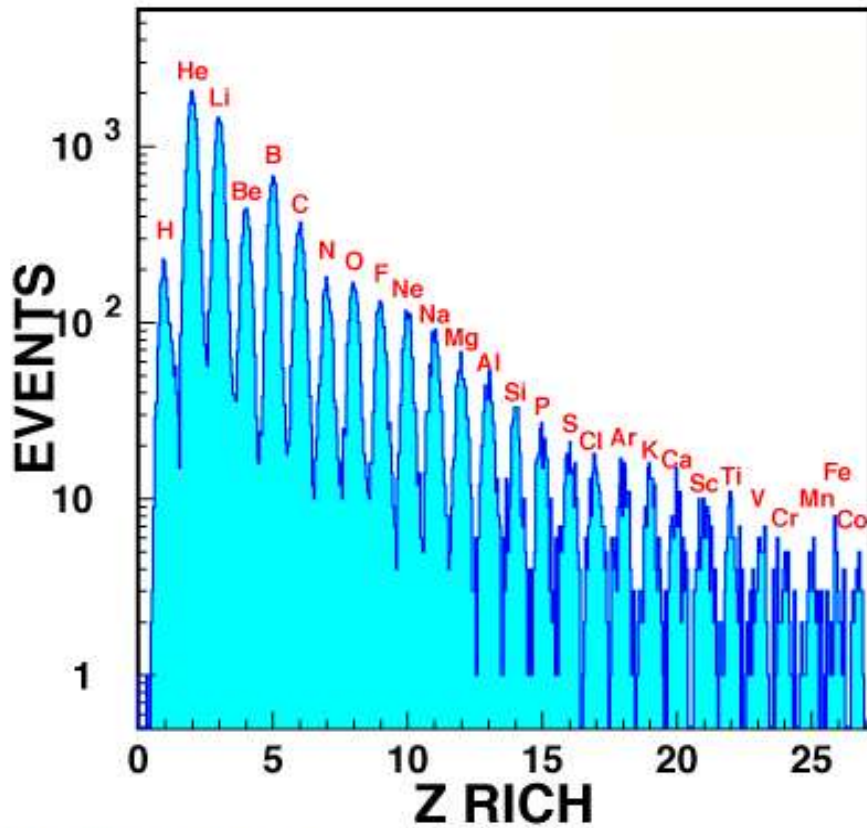
- 3 cm thick radiator consisting of  $1.13 \times 1.13 \times 1 \text{ cm}^3$  silica aerogel tiles ( $n \sim 1.03\text{-}1.05$ ) is mounted on support panel of last TOF plane
- Cherenkov photons detected in an array of 680,  $25 \text{ mm}^2$  multi-node ( $4 \times 4$ ) PMs @ 800 V, located 45 cm from the radiator; a  $64.3 \times 63.8 \text{ cm}^2$  aperture in front of the ECAL limits the geometric factor to  $\sim 0.4 \text{ m}^2\text{sr}$



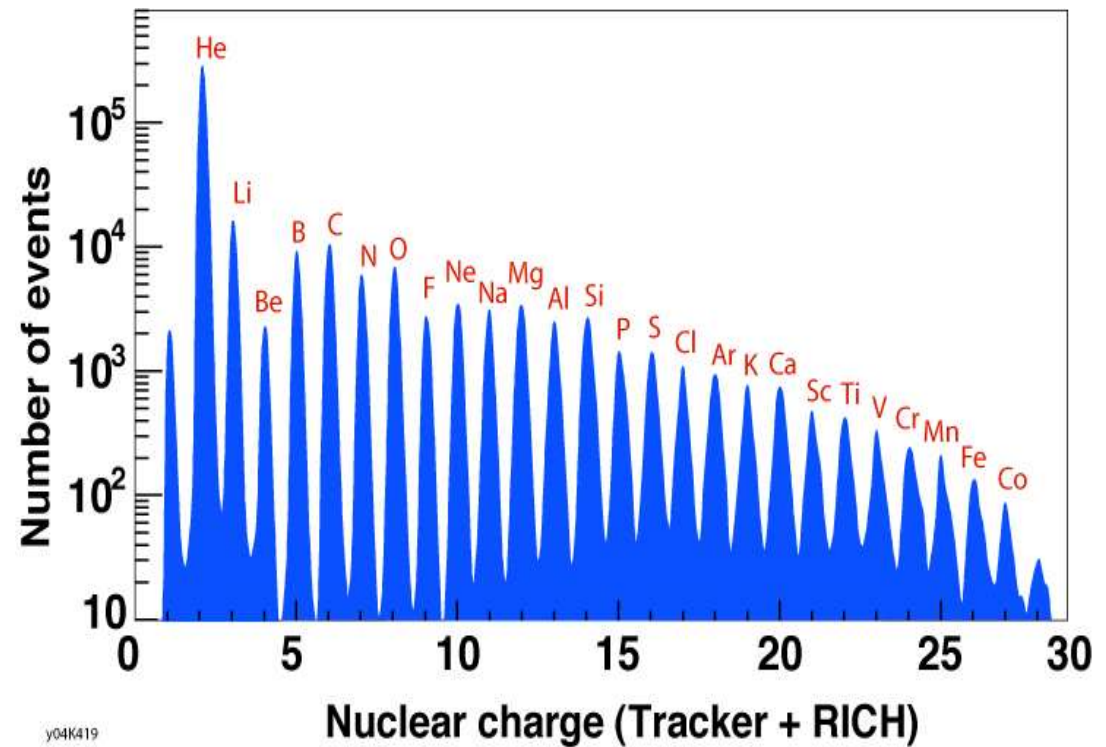
- velocity measurement resolution  $\sim 0.1\%$ , light isotope separation between 2 to 10 GeV/n, charge identification up to  $Z \sim 26$

# Test Beam Results – Particle Charge

with RICH

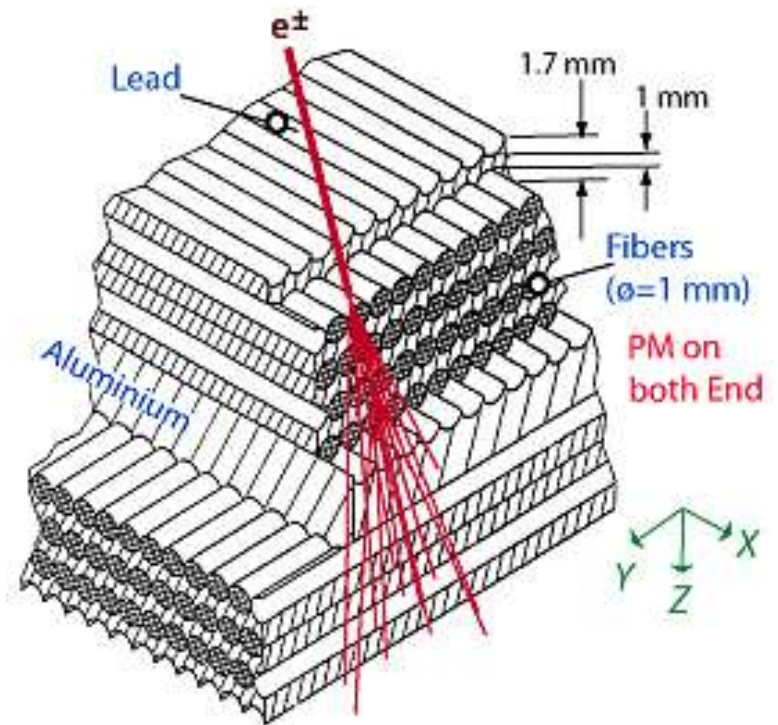


with RICH and Tracker



# AMS-02 Subsystems

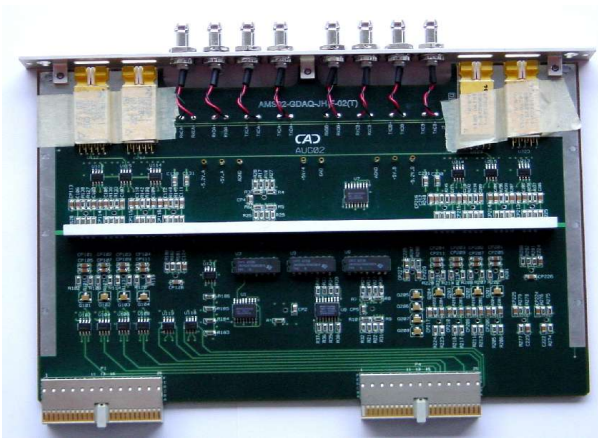
- ECAL, a Pb-scintillating fiber sampling calorimeter:
  - The  $65.8 \text{ cm}^2$  is divided in 9, 1.85 cm thick “superlayers”, each “superlayer” consists of 11, 1 mm thick grooved Pb foils interleaved with layers of 1 mm dia. scintillating fiber.
  - Each “superlayer” is read out by 36 multi-anode PMs @ 650 V with a  $0.9 \times 0.9 \text{ mm}^2$  pixel size.
  - Radiation length of 9.6 mm  
⇒ calorimeter thickness  $17.3 X_0$
  - $E_{\text{TOT}}$  from a few GeV to 1 TeV with an energy resolution of  $11.9\% / E^{1/2}(\text{GeV}) + 2.8\%$



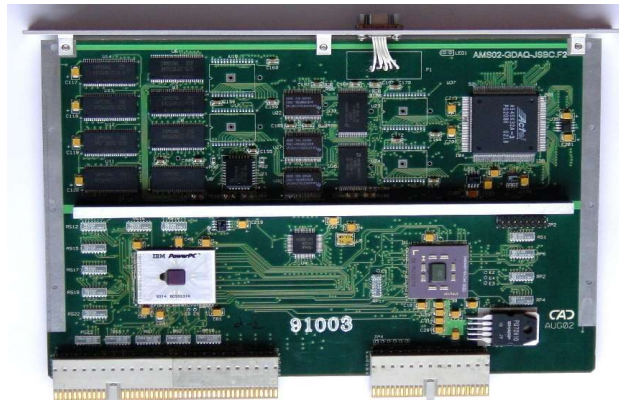


# AMS-02 Subsystems

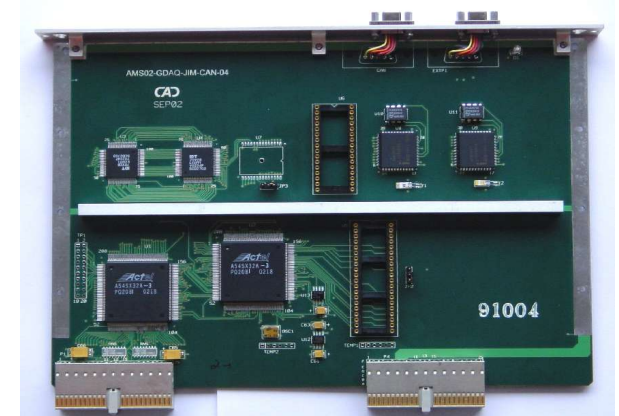
- DAQ:
  - ~227,300 channels (196,608 for the tracker), 16 bits/channel with event rates up to 2 kHz represent a raw data rate of 7.0 Gb/s
  - DAQ electronics must reduce to allocated **2 Mb/s downlink rate.**



JHIF



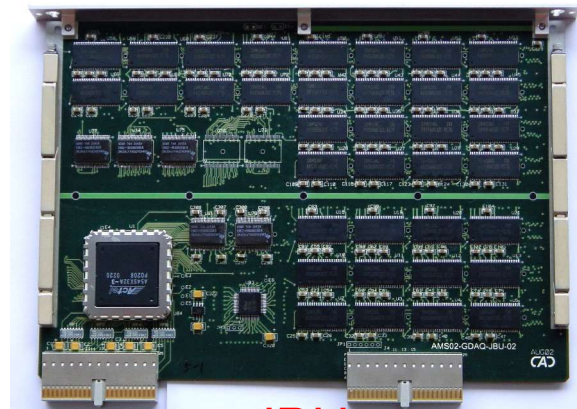
JSBC



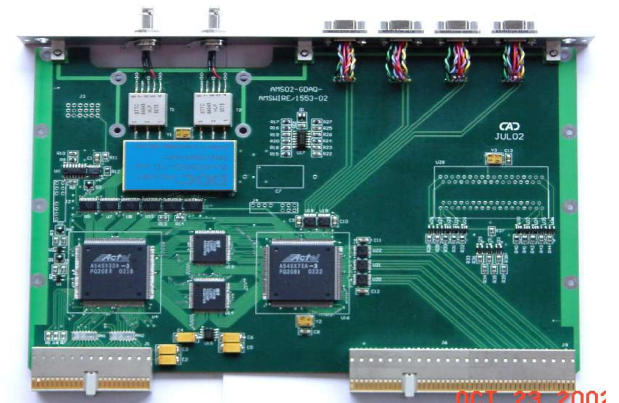
JIM-CAN



JIM-HRDL



JBU



JIM-AMSW&1553



# Activities Prior to Launch

- Complete subsystems
- Thermal vacuum and vibration tests of key detector elements
- EM testing
- Final detector integration (CERN)  
**present estimate exceeds 430 d**
- Test beam (?)
- Delivery to NASA 6 months before launch date

# Launch Dates (*at present*)

- Projected date for resumption of shuttle flights  
**June 2005**
- Projected data for AMS-02 launch  
**second half of 2007**

# AMS-02 on the ISS

